

or obtained on terms and conditions no more favorable than those offered to the general public.

(ii) *Examiners.* (A) An examiner, or the spouse or minor child of an examiner to whom the prohibition in paragraph (b)(1) of this section applies, may seek or obtain a credit card from a national bank the examiner is not assigned to examine so long as the credit card is obtained on terms and conditions no more favorable than those offered to the general public and the examiner submits to the Chief Counsel or designee a written disqualification from the examination of that bank. Such a recusal would not prevent an examiner from participating in other bank supervision matters outside the scope of an examination, such as licensing or supervisory policy decisions.

(B) For purposes of this section, examiners are assigned to examine a bank if they work:

(1) In a district, and the bank is one they examine or that is assigned to their Assistant Deputy Comptroller or rating official; or

(2) In Large Bank Supervision or Washington, D.C. Headquarters, and the bank is one to which they are regularly or otherwise assigned.

* * * * *

Dated: June 27, 2002.

David D. Aufhauser,
General Counsel, Department of the Treasury.

Approved: July 9, 2002.

Amy L. Comstock,
Director, Office of Government Ethics.

[FR Doc. 02-17918 Filed 7-16-02; 8:45 am]

BILLING CODE 4810-33-P

DEPARTMENT OF THE TREASURY

Office of the Comptroller of the Currency

12 CFR Part 25

[Docket No. 02-09]

RIN 1557-AB95

Prohibition Against Use of Interstate Branches Primarily for Deposit Production

AGENCY: Office of the Comptroller of the Currency, Treasury (OCC).

ACTION: Final rule; correction.

SUMMARY: On June 6, 2002, the OCC, the Board of Governors of the Federal Reserve System, and the Federal Deposit Insurance Corporation (collectively the Agencies) published a final rule in the **Federal Register** that amended each

Agency's regulation governing deposit production offices. This document corrects a typographical error in the OCC's regulation.

EFFECTIVE DATE: The correction made in this document is effective October 1, 2002.

FOR FURTHER INFORMATION CONTACT:

Patrick T. Tierney, Attorney, Legislative and Regulatory Activities Division (202-874-5090).

SUPPLEMENTARY INFORMATION: The comma that appears at the end of paragraph (d)(1) of 12 CFR 25.62 should be a semicolon, and paragraph (d)(2) of § 25.62 should begin on a new line. Therefore, in the final rule FR Doc. 02-14130, published on June 6, 2002 (67 FR 38844), make the following correction:

1. On page 38847, in the third column, in § 25.62, paragraphs (d)(1) and (d)(2) are correctly revised to read as follows:

§ 25.62 Definitions.

* * * * *

(d) * * *

(1) With respect to a State bank, the State that chartered the bank;

(2) With respect to a national bank, the State in which the main office of the bank is located;

* * * * *

Dated: July 8, 2002.

Julie L. Williams,
First Senior Deputy Comptroller and Chief Counsel, Office of the Comptroller of the Currency.

[FR Doc. 02-17757 Filed 7-16-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 23

[Docket No. CE186, Special Condition 23-119-SC]

Special Conditions; S-TEC on the New Piper Aircraft Corporation, PA 34-200T, Seneca V; Protection for High Intensity Radiated Fields (HIRF)

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final special conditions; request for comments.

SUMMARY: These special conditions are issued to S-TEC, One S-TEC Way Municipal Airport, Mineral Wells, Texas 76007, for a Supplemental Type Certificate for New Piper Aircraft Corporation, PA 34-200T, Seneca V airplanes. These airplanes will have

novel and unusual design features when compared to the state of technology envisaged in the applicable airworthiness standards. These novel and unusual design features include the installation of electronic flight instrument system (EFIS) displays Model Magic manufactured by Meggitt Avionics for which the applicable regulations do not contain adequate or appropriate airworthiness standards for the protection of these systems from the effects of high intensity radiated fields (HIRF). These special conditions contain the additional safety standards that the Administrator considers necessary to establish a level of safety equivalent to the airworthiness standards applicable to these airplanes.

DATES: The effective date of these special conditions is July 5, 2002. Comments must be received on or before August 16, 2002.

ADDRESSES: Comments may be mailed in duplicate to: Federal Aviation Administration, Regional Counsel, ACE-7, Attention: Rules Docket Clerk, Docket No. CE186, Room 506, 901 Locust, Kansas City, Missouri 64106. All comments must be marked: Docket No. CE186. Comments may be inspected in the Rules Docket weekdays, except Federal holidays, between 7:30 a.m. and 4 p.m.

FOR FURTHER INFORMATION CONTACT: Ervin Dvorak, Aerospace Engineer, Standards Office (ACE-110), Small Airplane Directorate, Aircraft Certification Service, Federal Aviation Administration, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone (816) 329-4123.

SUPPLEMENTARY INFORMATION: The FAA has determined that notice and opportunity for prior public comment hereon are impracticable because these procedures would significantly delay issuance of the approval design and thus delivery of the affected aircraft. In addition, the substance of these special conditions has been subject to the public comment process in several prior instances with no substantive comments received. The FAA, therefore, finds that good cause exists for making these special conditions effective upon issuance.

Comments Invited

Interested persons are invited to submit such written data, views, or arguments as they may desire. Communications should identify the regulatory docket or notice number and be submitted in duplicate to the address specified above. All communications received on or before the closing date for comments will be considered by the

Administrator. The special conditions may be changed in light of the comments received. All comments received will be available in the Rules Docket for examination by interested persons, both before and after the closing date for comments. A report summarizing each substantive public contact with FAA personnel concerning this rulemaking will be filed in the docket. Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must include a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. CE186." The postcard will be date stamped and returned to the commenter.

Background

On November 13, 2001, S-TEC Corporation, One S-TEC Way, Mineral Wells Airport, Mineral Wells, Texas 76067, made an application to the FAA for a new Supplemental Type Certificate for the New Piper Aircraft Corporation PA 34-200T Seneca V airplanes. The Seneca V is currently approved under Type Certificate No. A7SO. The proposed modification incorporates a novel or unusual design feature, such as digital avionics consisting of an EFIS, that is vulnerable to HIRF external to the airplane.

Type Certification Basis

Under the provisions of 14 CFR part 21, § 21.101, S-TEC must show that the New Piper Aircraft Company PA 34-200T Seneca V aircraft meets the following provisions, or the applicable regulations in effect on the date of application for the change to the PA 34-200T Seneca V: The Certification Basis that is incorporated by reference for airplane model PA 34-200T Seneca V of the Type Certificate Data Sheet No. A7SO: FAR 23 August 1, 1967, through Amendment 23-6, FAR 23.1301, 1309, 1311, and 1321 as amended by Amendment 23-49, and the special conditions adopted by this rulemaking action.

Discussion

If the Administrator finds that the applicable airworthiness standards do not contain adequate or appropriate safety standards because of novel or unusual design features of an airplane, special conditions are prescribed under the provisions of § 21.16.

Special conditions, as appropriate, as defined in § 11.19, are issued in accordance with § 11.38 after public notice and become part of the type certification basis in accordance with § 21.101 (b)(2).

Special conditions are initially applicable to the model for which they are issued. Should the applicant apply for a supplemental type certificate to modify any other model already included on the same type certificate to incorporate the same novel or unusual design feature, the special conditions would also apply to the other model under the provisions of § 21.101.

Novel or Unusual Design Features

S-TEC plans to incorporate certain novel and unusual design features into an airplane for which the airworthiness standards do not contain adequate or appropriate safety standards for protection from the effects of HIRF. These features include EFIS, which are susceptible to the HIRF environment, that were not envisaged by the existing regulations for this type of airplane.

Protection of Systems from High Intensity Radiated Fields (HIRF)

Recent advances in technology have given rise to the application in aircraft designs of advanced electrical and electronic systems that perform functions required for continued safe flight and landing. Due to the use of sensitive solid state advanced components in analog and digital electronics circuits, these advanced systems are readily responsive to the transient effects of induced electrical current and voltage caused by the HIRF. The HIRF can degrade electronic systems performance by damaging components or upsetting system functions.

Furthermore, the HIRF environment has undergone a transformation that was not foreseen when the current requirements were developed. Higher energy levels are radiated from transmitters that are used for radar, radio, and television. Also, the number of transmitters has increased significantly. There is also uncertainty concerning the effectiveness of airframe shielding for HIRF. Furthermore, coupling to cockpit-installed equipment through the cockpit window apertures is undefined.

The combined effect of the technological advances in airplane design and the changing environment has resulted in an increased level of vulnerability of electrical and electronic systems required for the continued safe flight and landing of the airplane. Effective measures against the effects of exposure to HIRF must be provided by the design and installation of these systems. The accepted maximum energy levels in which civilian airplane system installations must be capable of operating safely are based on surveys

and analysis of existing radio frequency emitters. These special conditions require that the airplane be evaluated under these energy levels for the protection of the electronic system and its associated wiring harness. These external threat levels, which are lower than previous required values, are believed to represent the worst case to which an airplane would be exposed in the operating environment.

These special conditions require qualification of systems that perform critical functions, as installed in aircraft, to the defined HIRF environment in paragraph 1 or, as an option to a fixed value using laboratory tests, in paragraph 2, as follows:

(1) The applicant may demonstrate that the operation and operational capability of the installed electrical and electronic systems that perform critical functions are not adversely affected when the aircraft is exposed to the HIRF environment defined below:

Frequency	Field strength (volts per meter) ¹	
	Peak	Average
10 kHz–100 kHz	50	50
100 kHz–500 kHz	50	50
500 kHz–2 MHz	50	50
2 MHz–30 MHz	100	100
30 MHz–70 MHz	50	50
70 MHz–100 MHz	50	50
100 MHz–200 MHz ...	100	100
200 MHz–400 MHz ...	100	100
400 MHz–700 MHz ...	700	50
700 MHz–1 GHz	700	100
1 GHz–2 GHz	2000	200
2 GHz–4 GHz	3000	200
4 GHz–6 GHz	3000	200
6 GHz–8 GHz	1000	200
8 GHz–12 GHz	3000	300
12 GHz–18 GHz	2000	200
18 GHz–40 GHz	600	200

¹ The field strengths are expressed in terms of peak root-mean-square (rms) values.

or,

(2) The applicant may demonstrate by a system test and analysis that the electrical and electronic systems that perform critical functions can withstand a minimum threat of 100 volts per meter, electrical field strength, from 10 kHz to 18 GHz. When using this test to show compliance with the HIRF requirements, no credit is given for signal attenuation due to installation.

A preliminary hazard analysis must be performed by the applicant, for approval by the FAA, to identify either electrical or electronic systems that perform critical functions. The term "critical" means those functions whose failure would contribute to, or cause, a failure condition that would prevent the continued safe flight and landing of the

airplane. The systems identified by the hazard analysis that perform critical functions are candidates for the application of HIRF requirements. A system may perform both critical and non-critical functions. Primary electronic flight display systems, and their associated components, perform critical functions such as attitude, altitude, and airspeed indication. The HIRF requirements apply only to critical functions.

Compliance with HIRF requirements may be demonstrated by tests, analysis, models, similarity with existing systems, or any combination of these. Service experience alone is not acceptable since normal flight operations may not include an exposure to the HIRF environment. Reliance on a system with similar design features for redundancy as a means of protection against the effects of external HIRF is generally insufficient since all elements of a redundant system are likely to be exposed to the fields concurrently.

Applicability

As discussed above, these special conditions are applicable to New Piper Aircraft Corporation PA 34-200T Seneca V airplane. Should S-TEC Corporation, apply at a later date for a supplemental type certificate to modify any other model on the same type certificate to incorporate the same novel or unusual design feature, the special conditions would apply to that model as well under the provisions of § 21.101.

Conclusion

This action affects only certain novel or unusual design features on one model of airplane. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the airplane.

The substance of these special conditions has been subjected to the notice and comment period in several prior instances and has been derived without substantive change from those previously issued. It is unlikely that prior public comment would result in a significant change from the substance contained herein. For this reason, and because a delay would significantly affect the certification of the airplane, which is imminent, the FAA has determined that prior public notice and comment are unnecessary and impracticable, and good cause exists for adopting these special conditions upon issuance. The FAA is requesting comments to allow interested persons to submit views that may not have been submitted in response to the prior

opportunities for comment described above.

List of Subjects in 14 CFR Part 23

Aircraft, Aviation safety, Signs and symbols.

Citation

The authority citation for these special conditions is as follows:

Authority: 49 U.S.C. 106(g), 40113 and 44701; 14 CFR 21.16 and 21.101; and 14 CFR 11.38 and 11.19.

The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for New Piper Aircraft Corporation PA 34-200T Seneca V airplane modified by S-TEC Corporation to add an EFIS.

1. *Protection of Electrical and Electronic Systems from High Intensity Radiated Fields (HIRF).* Each system that performs critical functions must be designed and installed to ensure that the operations, and operational capabilities of these systems to perform critical functions, are not adversely affected when the airplane is exposed to high intensity radiated electromagnetic fields external to the airplane.

2. For the purpose of these special conditions, the following definition applies: *Critical Functions:* Functions whose failure would contribute to, or cause, a failure condition that would prevent the continued safe flight and landing of the airplane.

Issued in Kansas City, Missouri on July 5, 2002.

James E. Jackson,

Acting Manager, Small Airplane Directorate.

[FR Doc. 02-18018 Filed 7-16-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-367-AD; Amendment 39-12821; AD 2002-14-21]

RIN 2120-AA64

Airworthiness Directives; Boeing Model 737-600, -700, and -800 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD),

applicable to certain Boeing Model 737-600, -700, and -800 series airplanes, that currently requires repetitive inspections to detect discrepancies of the quick-disconnect coupling on the fuel hose located at the fan case firewall; corrective action, if necessary; and installation of a clamp shell on the coupling to prevent separation of the coupling halves. This amendment limits the applicability of the existing requirements, clarifies certain existing requirements, and requires removal of the clamp shell installed previously and replacement of the existing quick-disconnect fuel supply hose, coupling, and strut fitting with new, fixed-B-nut-type parts. Such replacement ends the requirement for repetitive inspections. The actions specified by this AD are intended to prevent major fuel leakage due to excessive wear of the quick-disconnect coupling on the fuel hose, fire in the engine nacelle, and consequent loss of thrust from the affected engine, which could result in reduced controllability of the airplane. This action is intended to address the identified unsafe condition.

DATES: Effective August 21, 2002.

The incorporation by reference of Boeing Alert Service Bulletin 737-73A1011, Revision 2, dated July 13, 2000, as listed in the regulations, is approved by the Director of the Federal Register as of August 21, 2002.

The incorporation by reference of Boeing Alert Service Bulletin 737-73A1011, dated November 25, 1998, as listed in the regulations, was approved previously by the Director of the Federal Register as of February 19, 1999 (64 FR 5590, February 4, 1999).

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Douglas Pegors, Aerospace Engineer, Propulsion Branch, ANM-140S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1446; fax (425) 227-1181.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 99-03-08, amendment 39-11022 (64 FR 5590, February 4, 1999), which is applicable