

accordance with ASB 72–A0126, Revision 5, dated June 6, 2001, at the earliest of:

- (i) The first PPE after 1,000 CSN, or
- (ii) The first ESV after 3,000 CSN, or
- (iii) From July 29, 2001, through January 27, 2003, before 7,000 CSN, and after January 27, 2003, before 3,500 CSN.

(3) Spools not previously inspected in accordance with the requirements of ASB 72–A0137, Revision 3, dated January 31, 2002, or an earlier revision of ASB 72–0137, or SB 72–0137, inspect stage 3–5 dovetail slot bottoms in accordance with the requirements of ASB 72–A0137, Revision 3, dated January 31, 2002, at the earliest of:

- (i) The first PPE after 1,000 CSN, or
- (ii) The first HPCR exposure after 1,000 CSN, or
- (iii) The next required inspection to ASB 72–A0135, Revision 2, dated June 6, 2001.

Repetitive Inspection

(4) For spools that have already been inspected using one of the ASB's listed in Column A of Table 10; OR a combination of one procedure from Column B AND one from Column C; OR a combination of one procedure from Column D AND one from Column E, inspect the hub and bore in accordance with the requirements of ASB 72–A0135, Revision 2, dated June 6, 2001, the web and hub-to-web transition areas in accordance with ASB 72–A0126, Revision 5, dated June 6, 2001, and the stage 3–5 dovetail slot bottoms in accordance with ASB 72–A0137, Revision 3, dated January 31, 2002, at the earlier of:

- (i) Each PPE with more than 1,000 CSLI and 3,500 CSN, or
- (ii) From July 29, 2001, through January 27, 2003, before the cycle limits of Table 6, and after January 27, 2003, before the cycle limits of Table 5.

(5) If inspection findings equal or exceed the reject limits established by ASB 72–A0135, Revision 2, dated June 6, 2001; ASB 72–A0126, Revision 5, dated June 6, 2001; and ASB 72–A0137, Revision 3, dated January 31, 2002; replace spool before further flight.

(6) After the effective date of this AD, do not install any engine that has an HPCR stage 3–9 spool, P/N's 1669M22G01, 1669M22G03, 1782M22G01 and 1782M22G02, installed where the spool has 10,500 or more CSN.

CF6–80E1 8-inch billet 2-piece spools

(m) For CF6 HPCR stage 3–9 spool, P/N 1782M22G04, do the following:

(1) If the spool has not already been inspected using one of the ASB's or SB's listed in Column A of the following Table 9; OR a combination of one procedure from Column B AND one from Column C; OR a combination of one procedure from Column D AND one from Column E, inspect hub and bore in accordance with ASB 72–A0135, Revision 2, dated June 6, 2001, at the earlier of:

- (i) The first PPE after 1,000 CSN, or
- (ii) The first ESV after 6,000 CSN.

(2) For spools that have not been inspected in accordance with the requirements of ASB 72–A0126, Revision 5, dated June 6, 2001, or an earlier revision of ASB 72–A0126, or SB 72–0126, inspect the web and hub-to-web

transition areas in accordance with ASB 72–A0126, Revision 5, dated June 6, 2001, at the earlier of:

- (i) The first PPE after 1,000 CSN, or
- (ii) The first ESV after 6,000 CSN.

(3) For spools that have not been inspected in accordance with the requirements of ASB 72–A0137, Revision 3, dated January 31, 2002, or an earlier revision of ASB 72–A0137, or SB 72–0137, inspect the stage 3–5 dovetail slot bottoms in accordance with ASB 72–A0137, Revision 3, dated January 31, 2002, at the earlier of:

- (i) The first PPE after 1,000 CSN, or
- (ii) The first ESV after 6,000 CSN.

(4) If inspection findings equal or exceed the reject limits established by ASB 72–A0135, Revision 2, dated June 6, 2001; ASB 72–A0126, Revision 5, dated June 6, 2001; and ASB 72–A0137, Revision 3, dated January 31, 2002; replace spool before further flight.

Reporting Requirements

(n) Within five calendar days of inspection, report the results of inspections that equal or exceed the reject criteria to: Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive park, Burlington, MA 01803–5299; telephone (781) 238–7147. Reporting requirements have been approved by the Office of Management and Budget and assigned OMB control number 2120–0056. Be sure to include the following information:

- (1) Part Number
- (2) Serial Number
- (3) Spool CSN
- (4) Spool CSLI
- (5) Date and location where inspection was done.

Definitions

(o) For the purpose of this AD, the following definitions apply:

(1) A module level exposure is a separation of the fan module from the engine.

(2) An HPC rotor exposure is a HPC top or bottom case removal.

(3) A PPE is a disassembly and removal of the stage 3–9 spool from the HPCR structure, regardless of any blades, locking lugs, bolts or balance weights assembled to the spool.

(4) An ESV is the introduction of an engine into the shop where the separation of a major engine flange will occur after the effective date of this AD.

(5) The following maintenance actions, or any combination, are not considered ESV's for requiring repeat inspections:

(i) Introduction of an engine into a shop solely for removal of the compressor top or bottom case for airfoil maintenance.

(ii) Introduction of an engine into a shop solely for removal or replacement of the Stage 1 Fan Disk.

(iii) Introduction of an engine into a shop solely for replacement of the Turbine Rear Frame.

(iv) Introduction of an engine into a shop solely for replacement of the Accessory and/or Transfer Gearboxes.

(v) Introduction of an engine into a shop solely for replacement of the Fan Forward Case.

Alternative Methods of Compliance

(p) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(q) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be done.

Issued in Burlington, Massachusetts, on June 5, 2002.

Francis A. Favara,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 02–14700 Filed 6–11–02; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001–NM–84–AD]

RIN 2120–AA64

Airworthiness Directives; McDonnell Douglas Model MD–90–30 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all McDonnell Douglas Model MD–90–30 airplanes. This proposal would require one-time inspections to detect discrepancies of electrical wiring installations in various areas of the airplane; and corrective actions, if necessary. This action is necessary to prevent electrical arcing and/or heat-damaged wiring due to improper wire installations or maintenance practices, which could result in fire and smoke in various areas of the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by July 29, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport

Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-84-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2001-NM-84-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT:

Technical Information: George Mabuni, Aerospace Engineer, Systems and Equipment Branch, ANM-130L, FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5341; fax (562) 627-5210.

Other Information: Judy Golder, Airworthiness Directive Technical Editor/Writer; telephone (425) 687-4241, fax (425) 227-1232. Questions or comments may also be sent via the Internet using the following address: judy.golder@faa.gov. Questions or comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the

proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2001-NM-84-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2001-NM-84-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has become aware of several instances of damaged wiring insulation and chafed wiring in various areas on McDonnell Douglas Model MD-90-30 airplanes. Investigation revealed that such damage and chafing may be due to improper wire installations or maintenance practices. This condition, if not corrected, could lead to electrical arcing and/or heat-damaged wiring, which could result in fire and smoke in various areas of the airplane.

Other Relevant Rulemaking

The FAA has previously issued AD 2000-11-01, amendment 39-11749 (65 FR 34321, May 26, 2000), applicable to certain McDonnell Douglas Model DC-9-81 (MD-81), DC-9-82 (MD-82), DC-9-83 (MD-83), DC-9-87 (MD-87), MD-88, and MD-90-30 airplanes. That AD requires determining whether, and at what locations, metallized

polyethyleneterephthalate (MPET) insulation blankets are installed, and replacing MPET insulation blankets with new insulation blankets. The FAA recommends that the actions that would be required by this proposed AD be accomplished after accomplishing the replacement required by AD 2000-11-01. This proposed AD would not affect the current requirements of that AD.

Explanation of Relevant Service Information

The FAA has reviewed and approved the following service bulletins:

- Boeing Service Bulletin MD90-24-066, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the flight compartment and forward drop ceiling area.

- Boeing Service Bulletin MD90-24-067, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the electronic/electrical (E/E) compartment.

- Boeing Service Bulletin MD90-24-068, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the forward passenger compartment from stations Y=260.000 to Y=902.000.

- Boeing Service Bulletin MD90-24-069, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the aft passenger compartment from stations Y=902.000 to Y=1395.000.

- Boeing Service Bulletin MD90-24-070, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the forward and mid cargo compartments from stations Y=218.000 to Y=845.000.

- Boeing Service Bulletin MD90-24-071, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the aft cargo compartment from stations Y=1064.000 to Y=1369.000.

- Boeing Service Bulletin MD90-24-072, including Appendix A, Revision 01, dated February 8, 2001, which describes procedures for a one-time visual inspection of all electrical wiring installations in the forward accessory compartment from stations Y=41.000 to Y=70.000.

All of these service bulletins also describe procedures for corrective actions that may be necessary. These include repairing cracked, split, or torn wiring insulation; re-attaching nylon (caterpillar) grommets; installing smaller-sized clamps; adjusting, replacing, or tightening sta-straps; repositioning certain wiring or clamps; tightening or securing clamps, terminals, or wire bundles; re-torquing screw terminals of the flag lug bus bar; repairing or replacing certain wiring, terminals, splices, or connectors; installing protective sleeving over wiring; and installing a silicone glass cloth over conduit ends.

Accomplishment of the actions specified in the service bulletins is intended to adequately address the identified unsafe condition.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletins described previously, except as discussed below.

Differences Between Service Bulletins and Proposed AD

The service bulletins specify accomplishment of "visual" inspections. The FAA has determined that the inspections described in the service bulletins constitute "detailed inspections." Therefore, this proposed AD would require accomplishment of detailed inspections. Note 3 of this proposed AD defines such inspections.

Also, Appendix A of each service bulletin contains a form to report inspection findings. This proposed AD would NOT require such reports to be submitted to the FAA.

Cost Impact

There are approximately 115 airplanes of the affected design in the worldwide fleet. The FAA estimates that 25 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 49 work hours per airplane to accomplish all of the proposed inspections, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the inspections proposed by this AD on U.S. operators is estimated to be \$73,500, or \$2,940 per airplane.

Warranty remedies may be available from the airplane manufacturer for labor costs associated with this proposed AD. As a result, the costs attributable to the

proposed AD may be less than stated above.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

McDonnell Douglas: Docket 2001–NM–84–AD.

Applicability: All Model MD–90–30 airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

Note 2: The FAA recommends that the actions required by this AD be accomplished after the replacement of the metallized polyethyleneterephthalate (MPET) insulation blankets required by AD 2000–11–01, amendment 39–11749.

To prevent electrical arcing and/or heat-damaged wiring due to improper wire installations or maintenance practices, which could result in fire and smoke in various areas of the airplane, accomplish the following:

One-Time Detailed Inspections

(a) Within 5 years after the effective date of this AD, accomplish the actions specified in paragraphs (a)(1), (a)(2), (a)(3), (a)(4), (a)(5), (a)(6), and (a)(7) of this AD.

(1) Do a one-time detailed inspection of all electrical wiring installations in the flight compartment and forward drop ceiling area, according to the Accomplishment Instructions of Boeing Service Bulletin MD90–24–066, including Appendix A, Revision 01, dated February 8, 2001.

Note 3: For the purposes of this AD, a detailed inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required."

Note 4: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90–24–066, including Appendix A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(1) and (b)(1) of this AD, as applicable.

(2) Do a one-time detailed inspection of all electrical wiring installations in the

electronic/electrical (E/E) compartment according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-067, including Appendix A, Revision 01, dated February 8, 2001.

Note 5: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-067, including Appendix A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(2) and (b)(2) of this AD, as applicable.

(3) Do a one-time detailed inspection of all electrical wiring installations in the forward passenger compartment from stations Y=260.000 to Y=902.000 according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-068, including Appendix A, Revision 01, dated February 8, 2001.

Note 6: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-068, including Appendix A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(3) and (b)(3) of this AD, as applicable.

(4) Do a one-time detailed inspection of all electrical wiring installations in the aft passenger compartment from stations Y=902.000 to Y=1395.000 according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-069, including Appendix A, Revision 01, dated February 8, 2001.

Note 7: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-069, including Appendix A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(4) and (b)(4) of this AD, as applicable.

(5) Do a one-time detailed inspection of all electrical wiring installations in the forward and mid cargo compartments from stations Y=218.000 to Y=845.000 according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-070, including Appendix A, Revision 01, dated February 8, 2001.

Note 8: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-070, including Appendix A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(5) and (b)(5) of this AD, as applicable.

(6) Do a one-time detailed inspection of all electrical wiring installations in the aft cargo compartment from stations Y=1064.000 to Y=1369.000 according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-071, including Appendix A, Revision 01, dated February 8, 2001.

Note 9: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-071, including Appendix

A, dated July 28, 2000, are acceptable for compliance with paragraphs (a)(6) and (b)(6) of this AD, as applicable.

(7) Do a one-time detailed inspection of all electrical wiring installations in the forward accessory compartment from stations Y=41.000 to Y=70.000 according to the Accomplishment Instructions of Boeing Service Bulletin MD90-24-072, including Appendix A, Revision 01, dated February 8, 2001.

Note 10: Inspections and corrective actions done before the effective date of this AD according to the Accomplishment Instructions of McDonnell Douglas Service Bulletin MD90-24-072, including Appendix A, dated July 27, 2000, are acceptable for compliance with paragraphs (a)(7) and (b)(7) of this AD, as applicable.

Corrective Action

(b) If any discrepancy is detected during any inspection required by paragraph (a) of this AD: Before further flight, accomplish the applicable corrective action(s) according to the Accomplishment Instructions of the applicable service bulletins listed in paragraphs (b)(1), (b)(2), (b)(3), (b)(4), (b)(5), (b)(6), and (b)(7) of this AD. Corrective actions that may be necessary include repairing cracked, split, or torn wiring insulation; re-attaching nylon (caterpillar) grommets; installing smaller-sized clamps; adjusting, replacing, or tightening sta-straps; repositioning certain wiring or clamps; tightening or securing clamps, terminals, or wire bundles; re-torquing screw terminals of the flag lug bus bar; repairing or replacing certain wiring, terminals, splices, or connectors; installing protective sleeving over wiring; and installing a silicone glass cloth over conduit ends.

(1) Boeing Service Bulletin MD90-24-066, including Appendix A, Revision 01, dated February 8, 2001.

(2) Boeing Service Bulletin MD90-24-067, including Appendix A, Revision 01, dated February 8, 2001.

(3) Boeing Service Bulletin MD90-24-068, including Appendix A, Revision 01, dated February 8, 2001.

(4) Boeing Service Bulletin MD90-24-069, including Appendix A, Revision 01, dated February 8, 2001.

(5) Boeing Service Bulletin MD90-24-070, including Appendix A, Revision 01, dated February 8, 2001.

(6) Boeing Service Bulletin MD90-24-071, including Appendix A, Revision 01, dated February 8, 2001.

(7) Boeing Service Bulletin MD90-24-072, including Appendix A, Revision 01, dated February 8, 2001.

Note 11: Appendix A of the service bulletins referenced in paragraphs (b)(1), (b)(2), (b)(3), (b)(4), (b)(5), (b)(6), and (b)(7) of this AD contains a form to report inspection findings. This AD does NOT require such reports to be submitted to the FAA.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los

Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 12: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permits

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on June 4, 2002.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-14699 Filed 6-11-02; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Airspace Docket No. 00-AAL-08]

RIN 2120-AA66

Proposed Establishment of Colored Federal Airways; AK

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Proposed rule; withdrawal.

SUMMARY: This action withdraws the rulemaking proposal published in the **Federal Register** on February 13, 2001. In that action, the FAA proposed to establish two Federal airways in Alaska. The FAA has determined that withdrawal of the proposed rule is warranted since the proposed routes failed flight inspection due to weak navigational signals.

DATES: June 12, 2002.

FOR FURTHER INFORMATION CONTACT: Ken McElroy, Airspace and Rules Division, ATA-400, Office of Air Traffic Airspace Management, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267-8783.

SUPPLEMENTARY INFORMATION: On February 13, 2001, a proposed rule was published in the **Federal Register** that would have amended 14 CFR part 71 to establish two Federal airways in Alaska (66 FR 9990). Interested parties were invited to participate in the rulemaking process by submitting written data, views, or arguments regarding the