

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39****[Docket No. 2001-NE-17-AD; Amendment 39-12769; AD 2002-11-08]****RIN 2120-AA64****Airworthiness Directives; Rolls-Royce plc. RB211 Trent 875, 877, 884, 892, 892B, and 895 Series Turbofan Engines****AGENCY:** Federal Aviation Administration, DOT.**ACTION:** Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to Rolls-Royce plc (RR) RB211 Trent 875, 877, 884, 892, 892B, and 895 series turbofan engines with certain part number (P/N) low pressure compressor (LPC) fan blades installed. This action requires initial and repetitive ultrasonic inspection of the fan blade dovetail roots. This amendment is prompted by the loss of an LPC fan blade during takeoff. The actions specified in this AD are intended to prevent multiple LPC fan blade failures due to cracks, which could result in uncontained engine failure and possible damage to the airplane.

DATES: Effective June 21, 2002. The incorporation by reference of certain publications listed in the rule is approved by the Director of the Federal Register as of June 21, 2002.

Comments for inclusion in the Rules Docket must be received on or before August 5, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2001-NE-17-AD, 12 New England Executive Park, Burlington, MA 01803-5299. Comments may also be sent via the Internet using the following address: *9-ane-adcomment@faa.gov*. Comments sent via the Internet must contain the docket number in the subject line. The service information referenced in this AD may be obtained from Rolls-Royce plc, P.O. Box 31, Derby DE24 6BJ, UK; Telephone 44 (0) 1332 242424; fax 44 (0) 1332 249936. This information may be examined, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800

North Capitol Street, NW, suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Keith Mead, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7744; fax (781) 238-7199.

SUPPLEMENTARY INFORMATION: The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom (UK), recently notified the FAA that an unsafe condition may exist on RR RB211 Trent 875, 877, 884, 892, 892B, and 895 series turbofan engines with certain P/N LPC fan blades installed. The CAA advises there has been an incident involving the loss of an LPC fan blade during takeoff. The release of the blade occurred as the result of the initiation and propagation of a crack in the LPC fan blade root, convex side, located on either side of the shear key slot. A subsequent "around the fleet inspection" revealed a similar condition in four additional LPC fan blades. The effects of dry film lubrication and improved blade root lubrication (Metco 58) were determined to be critical in preventing the initiation of cracking in the root of the LPC fan blade. In addition, blade root configurations, airplane type, and engine ratings were found to affect initial and repetitive inspection requirements.

Manufacturer's Service Information

RR has issued service bulletin (SB) RB.211-72-D344, Revision 4, dated March 15, 2002, that provides procedures to ultrasonic-inspect the blade root on LPC fan blades. The CAA classified this service bulletin as mandatory and issued AD 001-02-2001, dated February 2, 2000, in order to assure the airworthiness of these RR engines in the UK.

Bilateral Airworthiness Agreement

This engine model is manufactured in the UK and is type certificated for operation in the United States under the provisions of § 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the CAA has kept the FAA informed of the situation described above. The FAA has examined the findings of the CAA, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

FAA's Determination of an Unsafe Condition and Required Actions

Since an unsafe condition has been identified that is likely to exist or develop on other RR RB211 Trent 875, 877, 884, 892, 892B, and 895 series turbofan engines of the same type design, this AD is being issued to prevent multiple LPC fan blade failures due to cracks, which could result in uncontained engine failure and possible damage to the airplane. This AD requires ultrasonic inspection of the dovetail roots of LPC fan blades P/N's FK30838, FK30840, FK30842, FW12960, FW12961, FW12962, and FW13175. The actions must be done in accordance with the service bulletin described previously.

Immediate Adoption of This AD

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire.

Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments

submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2001-NE-17-AD." The postcard will be date stamped and returned to the commenter.

Regulatory Analysis

This final rule does not have federalism implications, as defined in Executive Order 13132, because it would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Accordingly, the FAA has not consulted with state authorities prior to publication of this final rule.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2002-11-08 Rolls-Royce plc: Amendment 39-12769. Docket No. 2001-NE-17-AD.

Applicability

This airworthiness directive (AD) is applicable to Rolls-Royce plc. (RR) RB211 Trent 855, 877, 884, 892, 892B, and 895 series turbofan engines with low pressure compressor (LPC) fan blades, part numbers (P/N's) FK30838, FK30840, FK30842, FW12960, FW12961, FW12962, and FW13175, installed. These engines are installed on, but not limited to, Boeing Company 777 series airplanes.

Note 1: This AD applies to each engine identified in the preceding applicability provision, regardless of whether it has been

modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (h) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance

Compliance with this AD is required as indicated, unless already done.

To prevent multiple LPC fan blade failures due to cracks, which could result in uncontained engine failure and possible damage to the airplane, do the following:

(a) Ultrasonic-inspect and disposition the dovetail roots of LPC fan blades, P/N's FK30838, FK30840, FK30842, FW12960, FW12961, FW12962, and FW13175, that are removed from the engine, in accordance with 3.A.(1) through 3.A.(5) or, for blades that are not removed from the engine, in accordance with 3.B.(1) through 3.B.(5) of the Accomplishment Instructions of RR service bulletin (SB) RB.211-72-D344, Revision 4, dated March 15, 2002.

(b) For blades P/N's FK30838, FK30840, and FK30842, that have not been relubricated using either RR SB RB.211-72-D344 or RB.211-72-D347, during any interval exceeding 600 cycles-since-new (CSN) or cycles-since-rework (CSR), inspect in accordance with paragraph (a) of this AD and within the compliance times specified in the following Table 1:

TABLE 1

Engine series	Boeing 777 series (IGW)	Airplane maximum gross weight (times 1000 pounds)	Initial inspection (CSN)	Repetitive inspection (cycles-since-last-inspection (CSLI))
(1) -892	-300	(i) 660 and 632.5 (ii) 580	600 2,000	80 600
(2) -884, -892, -892B, and -895	-200 with IGW	(i) 632.5 and 648 (ii) 656 (iii) 555	1,200 600 2,000	100 80 600
(3) -875	-200	535	2,000	600
(4) -877	-200	545	2,000	600

(c) For blades P/N's FK30838, FK30840, and FK30842, that have been relubricated at intervals not exceeding 600 CSN or CSR using either RR SB RB.211-72-D344 or SB RB.211-72-D347, inspect in accordance with paragraph (a) of this AD and within the compliance times specified in the following Table 2:

TABLE 2

Engine series	Boeing 777 series (IGW)	Airplane maximum gross weight (times 1000 pounds)	Initial inspection (CSN)	Repetitive inspection (cycles-since-last-inspection (CSLI))
(1) -892	-300	(i) 660 and 632.5 (ii) 580	600 2,400	80 600
(2) -884, -892, -892B, and -895	-200 with IGW	(i) 632.5 and 648 (ii) 656 (iii) 555	2,400 600 2,400	100 80 600
(3) -875	-200	535	2,400	600
(4) -877	-200	545	2,400	600

(d) For blades P/N's FW12960, FW12961, FW12962, and FW13175, either new or reworked to that configuration at greater than 600 CSN or since previous rework, or that have not been relubricated during any interval exceeding 600 CSN or CSR using either RR SB RB.211-72-D344 or RB.211-72-D347 requirements, inspect in accordance with paragraph (a) of this AD and within the compliance times specified in the following Table 3:

TABLE 3

Engine series	Boeing 777 series (IGW)	Airplane maximum gross weight (times 1000 pounds)	Initial inspection (CSN)	Repetitive inspection (CSLI)
(1) -892	-300	(i) 660 and 632.5 (ii) 580	600 2,000	100 600
(2) -884, -892, -892B, and -895	-200 with IGW	(i) 632.5 and 648 (ii) 656 (iii) 555	1,200 600 2,000	125 100 600
(3) -875	-200	535	2,000	600
(4) -877	-200	545	2,000	600

(e) For blades P/N's FW12960, FW12961, FW12962, and FW13175, either new or reworked to that configuration at fewer than 600 CSN or since previous rework, and that have been relubricated using either RR SB RB.211-72-D344 or SB RB.211-72-D347 at intervals not exceeding 600 CSN or repetitive lubrication, inspect in accordance with paragraph (a) of this AD and within the compliance times specified in the following Table 4:

TABLE 4

Engine series	Boeing 777 series (IGW)	Airplane maximum gross weight (times 1000 pounds)	Initial inspection (CSN)	Repetitive inspection (CSLI)
(1) -892	-300	(i) 660 and 632.5 (ii) 580	600 2,400	100 1,200
(2) -884, -892, -892B, and -895	-200 with IGW	(i) 632.5 and 648 (ii) 656 (iii) 555	2,400 600 2,400	125 100 1,200
(3) -875	-200	535	2,400	1,200
(4) -877	-200	545	2,400	1,200

(f) When engines containing blades P/N's FK30838, FK30840, FK30842, FW12960, FW12961, FW12962, and FW13175 are moved from one gross weight category to another, the inspection schedule that is applicable to the higher gross weight category must be used.

Optional Terminating Action

(g) Replacement of LPC fan blades P/N's FK30838, FK30840, FK30842, FW12960,

FW12961, FW12962, and FW13175 with a complete set of LPC fan blades that have a P/N that is not listed in this AD constitutes terminating action for the repetitive inspection requirements of paragraphs (a) through (e) of this AD.

Alternative Methods of Compliance

(h) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be

used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(i) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be done.

Documents That Have Been Incorporated By Reference

(j) The inspection must be done in accordance with Rolls-Royce plc. (RR) service bulletin RB.211-72-D344, Revision 4, dated March 15, 2002. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Rolls-Royce plc P.O. Box 31, Derby DE24 6BJ, UK; Telephone 44 (0) 1332 242424; fax 44 (0) 1332 249936. Copies may be inspected, by appointment, at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in CAA airworthiness directive 001-02-2001, dated February 2, 2000.

Effective Date

(k) This amendment becomes effective on June 21, 2002.

Issued in Burlington, Massachusetts, on May 27, 2002.

Mark C. Fulmer,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 02-13885 Filed 6-5-02; 8:45 am]

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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

14 CFR Part 1260

RIN 2700-AC53

NASA Grant and Cooperative Agreement Handbook—Limitations on Incremental Funding and Deobligations on Grants, and Elimination of Delegation of Closeout of Grants and Cooperative Agreements to Office of Naval Research (ONR); Correction

AGENCY: National Aeronautics and Space Administration

ACTION: Final rule; correction.

SUMMARY: NASA published a final rule document in the **Federal Register** of Tuesday, May 7, 2002 (FR DOC. 02-11167) to revise the threshold for incrementally funding grants and to

establish dollar thresholds for incremental funding and funding deobligation actions under grants. This document corrects the RIN number, which was incorrect in that rule.

DATES: Effective on May 7, 2002.

FOR FURTHER INFORMATION CONTACT: Rita Svarcas, NASA Headquarters, Code HC, Washington, DC, (202) 358-0464, e-mail: rsvarcas@hq.nasa.gov.

SUPPLEMENTARY INFORMATION: In the rule document published on page 30544 in the **Federal Register** of Tuesday, May 7, 2002, The RIN number is corrected to read "RIN 2700-AC53."

Scott Thompson,

Acting Assistant Administrator for Procurement.

[FR Doc. 02-14160 Filed 6-5-02; 8:45 am]

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DEPARTMENT OF COMMERCE

Bureau of Industry and Security

15 CFR Parts 732, 734, 738, 740, 742, 748, 770, 772, and 774

[Docket No. 020502105-2105-01]

RIN 0694-AC61

Revisions and Clarifications to Encryption Controls in the Export Administration Regulations—Implementation of Changes in Category 5, Part 2 ("Information Security"), of the Wassenaar Arrangement List of Dual-Use Goods and Other Technologies

AGENCY: Bureau of Industry and Security, Commerce.

ACTION: Interim final rule.

SUMMARY: This rule amends the Export Administration Regulations (EAR) to reflect changes made to the Wassenaar Arrangement List of dual-use items, and to update and clarify other provisions of the EAR pertaining to encryption export controls. Consistent with the Wassenaar changes, Note No. 3 ("Cryptography Note") to Category 5—part II (Information Security) of the Commerce Control List (CCL) is amended to allow mass market treatment for all encryption products, including products with symmetric algorithms employing key lengths greater than 64-bits, that previously were not eligible for mass market treatment. As a result, for the first time, mass market encryption

commodities and software with symmetric key lengths exceeding 64 bits may be exported and reexported to most destinations without a license under Export Control Classification Numbers

(ECCNs) 5A992 and 5D992, following a 30-day review by the Bureau of Industry and Security (BIS) (formerly the Bureau of Export Administration (BXA)). In addition, this rule, for the first time, allows equipment controlled under ECCN 5B002 to be exported and reexported under License Exception ENC. For all other information security items, including encryption source code that would be considered publicly available, this rule updates and clarifies existing notification, review, licensing and post-export reporting requirements. Restrictions on exports and reexports of encryption items to terrorist-supporting states (Cuba, Iran, Iraq, Libya, North Korea, Sudan and Syria), their nationals and other sanctioned persons (individuals and entities) are not changed by this rule.

DATES: This rule is effective June 6, 2002.

FOR FURTHER INFORMATION CONTACT: Norman E. LaCroix, Office of Strategic Trade and Foreign Policy Controls, Bureau of Industry and Security, Telephone: (202) 482-4439.

SUPPLEMENTARY INFORMATION:

Background

On October 19, 2000, the United States updated its encryption export regulations to provide consistent treatment with regulations adopted by the European Union (EU) easing export and reexport restrictions among the 15 EU member states and Australia, Czech Republic, Hungary, Japan, New Zealand, Norway, Poland and Switzerland. Subsequent to the publication of this amendment to the Export Administration Regulations (EAR), the member nations of the Wassenaar Arrangement agreed to remove key length restrictions on encryption hardware and software that is subject to the Cryptography Note (Note No. 3) to Category 5—part II (Information Security) of the Commerce Control List (CCL). This action effectively removed "mass market" encryption products from the list of dual-use items controlled by the Wassenaar Arrangement.

The U.S. encryption export control policy continues to rest on three principles: review of encryption products prior to sale, streamlined post-export reporting, and license review of certain exports of strong encryption to foreign government end-users. Consistent with these principles, this amendment updates the U.S. encryption export control policy in several areas.

For "mass market" encryption hardware and software products, this rule removes Encryption Item ("EI") and