

Actions	Compliance	Procedures
(3) Install rivets where rivets are missing and replace rivets that are the wrong size and/or type with the correct rivet.	Prior to further flight after the inspection required in paragraph (d)(2) of this AD, unless already accomplished.	In accordance with the Accomplishment Instructions section of Raytheon Mandatory Service Bulletin SB 54-3308, Issued: October, 2000, and the applicable maintenance manual.

Note 1: Although not required by this AD, Raytheon Mandatory Service Bulletin SB 54-3308, Issued: October, 2000, recommends inspecting the airplane in accordance with the Hard Landing Inspection procedure, Chapter 5-50-00, Beech King Air 90 Maintenance Manual, if the airplane should experience a hard landing prior to the repair required by this AD. If serious structural damage occurred, contact Raytheon Technical Support for assistance.

(e) *Can I comply with this AD in any other way?* You may use an alternative method of compliance or adjust the compliance time if:

(1) Your alternative method of compliance provides an equivalent level of safety; and

(2) The Manager, Wichita Aircraft Certification Office (ACO), approves your alternative. Submit your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note 2: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

(f) *Where can I get information about any already-approved alternative methods of compliance?* Contact Steve Potter, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Mid-Continent Airport, Wichita, Kansas 67209; telephone: (316) 946-4124; facsimile: (316) 946-4407.

(g) *What if I need to fly the airplane to another location to comply with this AD?* The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.

(h) *Are any service bulletins incorporated into this AD by reference?* Actions required by this AD must be done in accordance with Raytheon Mandatory Service Bulletin SB 54-3308, Issued: October, 2000. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from Raytheon Aircraft Company, P.O. Box 85, Wichita, Kansas 67201-0085. You can look at copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the

Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

(i) *When does this amendment become effective?* This amendment becomes effective on June 21, 2002.

Issued in Kansas City, Missouri, on April 30, 2002.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-11333 Filed 5-8-02; 8:45 am]

BILLING CODE 4910-13-P

The Federal Aviation Administration (FAA) must receive any comments on this rule on or before July 8, 2002.

ADDRESSES: Submit comments to FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2002-CE-17-AD, 901 Locust, Room 506, Kansas City, Missouri 64106. You may view any comments at this location between 8 a.m. and 4 p.m., Monday through Friday, except Federal holidays. You may also send comments electronically to the following address: 9-ACE-7-Docket@faa.gov. Comments sent electronically must contain "Docket No. 2002-CE-17-AD" in the subject line. If you send comments electronically as attached electronic files, the files must be formatted in Microsoft Word 97 for Windows or ASCII text.

You may get the service information referenced in this AD from Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277; telephone: (316) 517-5800; facsimile: (316) 942-9006. You may view this information at FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2002-CE-17-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Robert Adamson, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: 316-946-4145; facsimile: 316-946-4407.

SUPPLEMENTARY INFORMATION:

Discussion

What Events Have Caused This AD?

The FAA has received evidence of chafing and/or arcing of the electrical wiring leading to the fuel boost pump reservoir. Further investigation revealed confirmed reports of chafing and/or arcing of the fuel boost pump wiring inside the fuel pump reservoir that supplies fuel to each engine.

What Are the Consequences if the Condition Is Not Corrected?

This condition, if not corrected, could result in ignition of explosive vapor within the fuel storage system.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Cessna Aircraft Company (Cessna) Model CESSNA 441 airplanes. This AD requires you to do a one-time inspection of the fuel boost pump wiring inside and outside the boost pump reservoir, and repair or replace the wiring as necessary. This AD is the result of several reports of chafing and/or arcing of the fuel boost pump wiring inside and outside the fuel pump reservoir. The actions specified by this AD are intended to detect and correct chafing and/or arcing boost pump wiring, which could result in arcing within the wing fuel storage system. Such failure could lead to ignition of explosive vapor within the fuel storage system.

DATES: This AD becomes effective on May 31, 2002.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulation as of May 31, 2002.

Is There Service Information That Applies to This Subject?

Cessna has issued Conquest Service Bulletin No. CQB02-1R1, Revision 1, dated April 22, 2002.

The service bulletin includes procedures for:

- Inspecting the 5718106-1 wire harness and fuel boost pump lead wires for chafing or damage; and
- Repairing or replacing the chafed or damaged wiring as necessary.

The FAA's Determination and an Explanation of the Provisions of This AD

What Has FAA Decided?

The FAA has reviewed all available information, including the service information referenced above; and determined that:

- The unsafe condition referenced in this document exists or could develop on other Cessna Model CESSNA 441 airplanes of the same type design;
- The actions specified in the previously-referenced service information (as specified in this AD) should be accomplished on the affected airplanes; and
- AD action should be taken in order to correct this unsafe condition.

What Does This AD Require?

This AD requires you to: (1) Do a one-time inspection of the electrical wiring going to the fuel boost pump reservoir and the boost pump wiring inside the reservoir, and (2) repair or replace the wiring as necessary.

In preparation of this rule, we contacted type clubs and aircraft operators to obtain technical information and information on operational and economic impacts. We did not receive any information through these contacts. If received, we would have included, in the rulemaking docket, a discussion of any information that may have influenced this action.

The FAA is not including a repetitive inspection requirement in this AD. The Administrative Procedure Act does not permit the FAA to "bootstrap" a long-term requirement into an urgent safety of flight action where the rule becomes effective at the same time the public has the opportunity to comment. The short-term action and the long-term action are analyzed separately for justification to bypass prior public notice.

After issuing this AD, the FAA may initiate further AD action (notice of proposed rulemaking followed by a final rule) to require these inspections to be repetitive. Credit will be given in any subsequent action for the initial inspection done under this AD.

Will I Have the Opportunity To Comment Prior to the Issuance of the Rule?

Because the unsafe condition described in this document could result in ignition of explosive vapor within the fuel storage system, we find that notice and opportunity for public prior comment are impracticable. Therefore, good cause exists for making this amendment effective in less than 30 days.

Comments Invited

How Do I Comment on This AD?

Although this action is in the form of a final rule and was not preceded by notice and opportunity for public comment, FAA invites your comments on the rule. You may submit whatever written data, views, or arguments you choose. You need to include the rule's docket number and submit your comments to the address specified under the caption **ADDRESSES**. We will consider all comments received on or before the closing date specified above. We may amend this rule in light of comments received. Factual information that supports your ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether we need to take additional rulemaking action.

Are There Any Specific Portions of the AD I Should Pay Attention to?

We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. You may view all comments we receive before and after the closing date of the rule in the Rules Docket. We will file a report in the Rules Docket that summarizes each FAA contact with the public that concerns the substantive parts of this AD.

How Can I Be Sure FAA Receives My Comment?

If you want us to acknowledge the receipt of your comments, you must include a self-addressed, stamped postcard. On the postcard, write "Comments to Docket No. 2002-CE-17-AD." We will date stamp and mail the postcard back to you.

Compliance Time of This AD

What Is the Compliance Time of This AD?

The compliance time of this proposed AD is within the next 25 hours time-in-service (TIS) or 60 calendar days, whichever occurs first, after the effective date of this AD.

Why Is the Compliance Time of This AD Presented in Both Hours TIS and Calendar Time?

The affected airplanes are used in general aviation operations. Those operators may accumulate 25 hours TIS on the airplane in less than 60 calendar days and many owners have numerous affected airplanes. We have determined that the dual compliance time:

- Gives all owners/operators of the affected airplanes adequate time to schedule and do the actions in this AD; and
- Ensures that the unsafe condition referenced in this AD will be corrected within a reasonable time period without inadvertently grounding any of the affected airplanes.

Regulatory Impact

Does This AD Impact Various Entities?

These regulations will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, FAA has determined that this final rule does not have federalism implications under Executive Order 13132.

Does This AD Involve a Significant Rule or Regulatory Action?

We have determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and is not a significant regulatory action under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket (otherwise, an evaluation is not required). A copy of it, if filed, may be obtained from the Rules Docket.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends § 39.13 by adding a new airworthiness directive (AD) to read as follows:

2002-09-13 Cessna Aircraft Company:

Amendment 39-12746; Docket No. 2002-CE-17-AD.

(a) *What airplanes are affected by this AD?*

This AD applies to the following airplane models and serial numbers that are certificated in any category:

Model	Serial Nos.
CESSNA 441	0001 through 0362; and 698.

(b) *Who must comply with this AD?*

Anyone who wishes to operate any of the airplanes identified in paragraph (a) of this AD must comply with this AD.

(c) *What problem does this AD address?*

The actions specified by this AD are intended to detect and correct chafing and/or arcing boost pump wiring, which could result in arcing within the wing fuel system. Such failure could lead to ignition of explosive vapor within the fuel storage system.

(d) *What must I do to address this problem?* To address this problem, you must accomplish the following actions:

Actions	Compliance	Procedures
(1) Inspect the 5718106-1 wire harness and fuel boost pump lead wires for chafing or damage.	Within the next 25 hours time-in-service (TIS) after May 31, 2002 (the effective date of this AD) or 60 days after May 31, 2002 (the effective date of this AD), whichever occurs first.	In accordance with Cessna Conquest Service Bulletin No.: CQB02-1R1, Revision 1, dated April 22, 2002.
(2) If any wire harness or fuel boost pump lead wires are found chafed or damaged during the inspection required in paragraph (d)(1) of this AD, repair or replace the harness or lead wires.	Before further flight, after the inspection required in paragraph (d)(1) of this AD.	In accordance with Cessna Conquest Service Bulletin No.: CQB02-1R1, Revision 1, dated April 22, 2002.

(e) *Can I comply with this AD in any other way?* You may use an alternative method of compliance or adjust the compliance time if:

(1) Your alternative method of compliance provides an equivalent level of safety; and

(2) The Manager, Wichita ACO, approves your alternative. Submit your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

(f) *Where can I get information about any already-approved alternative methods of compliance?* Contact Robert Adamson, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: 316-946-4145; facsimile: 316-946-4407.

(g) *What if I need to fly the airplane to another location to comply with this AD?* The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.

(h) *Are any service bulletins incorporated into this AD by reference?* Actions required by this AD must be done in accordance with Cessna Conquest Service Bulletin No.

CQB02-1R1, Revision 1, dated April 22, 2002. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277; telephone: (316) 517-5800; facsimile: (316) 942-9006. You may view this information at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) *When does this amendment become effective?* This amendment becomes effective on May 31, 2002.

Issued in Kansas City, Missouri, on May 1, 2002.

Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-11523 Filed 5-8-02; 8:45 am]

BILLING CODE 4910-13-P

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

14 CFR Part 1240

[Notice (02-054)]

RIN 2700-AC47

Inventions and Contributions

AGENCY: National Aeronautics and Space Administration (NASA).

ACTION: Final rule.

SUMMARY: NASA is amending its regulation to provide definitions, to add a new category of initial awards for

release of software, to provide initial awards for the issuance of patents based upon continuation-in-part and divisional patent applications, to increase the amount of certain awards, and to change delegations of authority from the NASA Administrator.

EFFECTIVE DATE: May 9, 2002.

ADDRESSES: Inventions and Contributions Board, Code RI, National Aeronautics and Space Administration, Washington, DC 20546-0001.

FOR FURTHER INFORMATION CONTACT: Mr. Walter D. Hussey Director of Staff, Inventions and Contributions Board, 202-358-2468.

SUPPLEMENTARY INFORMATION: To aid the NASA Inventions and Contributions Board in processing applications for awards, all applications must now be submitted using electronic media.

NASA now provides initial awards for the filing of a non-provisional U.S. patent application, or upon the issuance of a patent for a continuation-in-part or a divisional patent application, for an invention made and reported by an employee of NASA or an employee of a NASA contractor. The amounts of these awards are at least \$1,000 for a sole inventor and at least \$500 for each joint inventor. Also, no additional award is authorized for a continuation of a patent application where an initial award was authorized for the parent application and the parent application will be or has been abandoned. Furthermore, initial awards are not authorized for provisional applications under 35 U.S.C.