

methods of compliance? Contact Werner Koch, Aerospace Engineer, FAA, Airplane Certification Office, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150; telephone: (817) 222-5133; facsimile: (817) 222-5960.

(g) *What if I need to fly the airplane to another location to comply with this AD?* The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.

(h) *Are any service bulletins incorporated into this AD by reference?*

Actions required by this AD must be done in accordance with Fairchild Aircraft Service Bulletin 226-32-069 including Overhaul Instructions With Parts Breakdown, Issued: October 24, 2001, and Fairchild Aircraft Service Bulletin 227-32-045 including Overhaul Instructions With Parts Breakdown, Issued: October 24, 2001. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from Fairchild Aircraft, Inc., P.O. Box 790490, San Antonio, Texas 78279-0490. You can look at copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

(i) *When does this amendment become effective?* This amendment becomes effective on June 6, 2002.

Issued in Kansas City, Missouri, on April 8, 2002.

James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 02-8988 Filed 4-16-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002-SW-08-AD; Amendment 39-12711; AD 2002-06-52]

RIN 2120-AA64

Airworthiness Directives; Bell Helicopter Textron Canada Model 407 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This document publishes in the **Federal Register** an amendment adopting Airworthiness Directive (AD) 2002-06-52, which was sent previously to all known U.S. owners and operators of Bell Helicopter Textron Canada (BHTC) Model 407 helicopters by individual letters. This AD requires a one-time replacement of certain bearings and, before further flight, adding a limitation and caution to the rotorcraft flight manual (RFM) and at specified intervals, inspecting, replacing, and lubricating certain oil cooler blower bearings. This AD is prompted by several occurrences of failure of an oil cooler blower bearing. The actions specified by this AD are intended to prevent failure of an oil cooler blower bearing, loss of tail rotor drive, and a subsequent forced landing.

DATES: Effective May 2, 2002, to all persons except those persons to whom it was made immediately effective by Emergency AD 2002-06-52, issued on March 15, 2002, which contained the requirements of this amendment.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 2, 2002.

Comments for inclusion in the Rules Docket must be received on or before June 17, 2002.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2002-SW-08-AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. You may also send comments electronically to the Rules Docket at the following address: 9-asw-adcomments@faa.gov.

The applicable service information may be obtained from Bell Helicopter Textron Canada, 12,800 Rue de l'Avenir, Mirabel, Quebec J7Y1R4, telephone (450) 437-2862 or (800) 363-8023, fax (450) 433-0272. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Paul Madej, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, Fort Worth, Texas 76193-0110, telephone (817) 222-5125, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: On February 10, 2000, the FAA issued Final Rule AD 2000-02-12 (65 FR 8032, February 17, 2000), to require inspecting each oil cooler blower bearing (bearing)

for roughness and replacing any rough bearing before further flight. That AD was prompted by reports of failure of the bearing. Since the issuance of that AD, continued bearing failures and identifications of effects of engine exhaust gas ingestion have been reported. On March 15, 2002, the FAA issued superseding Emergency AD 2002-06-52 for BHTC Model 407 helicopters. That emergency AD requires a one-time replacement of certain bearings within 100 hours time-in-service, and before further flight, adding a limitation and caution to the RFM and at specified intervals, inspecting and, if necessary, replacing certain bearings and lubricating certain bearings. That action was prompted by several occurrences of failure of an oil cooler blower bearing. Particular tailwind conditions during flight can result in engine exhaust gas ingestion by the oil cooler blower and deterioration of the bearing grease. This condition, if not corrected, could result in bearing failure, loss of tail rotor drive, and a subsequent forced landing.

The FAA has reviewed Bell Helicopter Textron Alert Service Bulletin (ASB) Nos. 407-01-44, Revision A, dated October 25, 2001; 407-01-47, dated November 9, 2001; and 407-02-49, dated January 7, 2002. ASB 407-01-44, Revision A, dated October 25, 2001, specifies replacing specific oil cooler blower bearings and clarifies and expands the bearing lubrication procedure and schedule. ASB 407-01-47, dated November 9, 2001, updates the inspection and lubrication procedures and schedule for specified bearings at all oil cooler blower and tail rotor driveshaft locations. ASB 407-02-49, dated January 7, 2002, introduces a new limitation and a new caution for tailwind operations in the RFM and maintenance actions for exceeding the limitations.

Transport Canada, which is the airworthiness authority for Canada, notified the FAA that an unsafe condition may exist on this helicopter model. Transport Canada advises that testing indicates premature failure of an oil cooler blower bearing can occur, under certain conditions, due to ingesting exhaust gases into the aft fairing inlet resulting in elevated temperatures. Also, Transport Canada advises that research indicates that over-greasing the bearing can result in elevated bearing temperatures and failure of a bearing. Transport Canada classified the service bulletins as mandatory and issued AD No. CF-2002-18, dated March 4, 2002, to ensure

the continued airworthiness of these helicopters.

This helicopter model is manufactured in Canada and is type certificated for operation in the United States under the provisions of 14 CFR 21.29 and the applicable bilateral agreement. Pursuant to the applicable bilateral agreement, Transport Canada has kept the FAA informed of the situation described above. The FAA has examined the findings of Transport Canada, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

This unsafe condition is likely to exist or develop on other BHTC Model 407 helicopters of the same type design registered in the United States. Therefore, the FAA issued Emergency AD 2002-06-52 to prevent failure of an oil cooler blower bearing, loss of tail rotor drive, and a subsequent forced landing. The AD requires:

- Before further flight, adding the tailwind limitation and caution contained in Temporary Revision 9 (the temporary revision is attached to ASB 407-02-49, dated January 7, 2002) to the RFM.
- At specified intervals, inspecting the oil cooler blower bearings; and if a bearing is rough, a seal is torn, the expelled grease has turned black, or metal particles are visible in the expelled grease, replacing the affected bearing before further flight.
- At a specified time-in-service, replacing certain bearings.
- At specified intervals, lubricating the bearings.

The actions must be accomplished in accordance with the ASBs described previously. The short compliance time involved is required because the previously described critical unsafe condition can adversely affect the structural integrity and controllability of the helicopter. Therefore, the actions previously described are required at the specified time intervals, and this AD must be issued immediately.

Since it was found that immediate corrective action was required, notice and opportunity for prior public comment thereon were impracticable and contrary to the public interest, and good cause existed to make the AD effective immediately by individual letters issued on March 15, 2002, to all known U.S. owners and operators of BHTC Model 407 helicopters. These conditions still exist, and the AD is hereby published in the **Federal Register** as an amendment to 14 CFR 39.13 to make it effective to all persons.

The FAA estimates that 281 helicopters of U.S. registry will be affected by this AD. It will take approximately 1 work hour for each RFM revision; 2 work hours per helicopter for the initial inspection; 0.5 hour for each repetitive inspection; 0.5 hour to lubricate the oil cooler blower bearing; and 4 work hours per helicopter to replace the oil cooler blower bearing. Required parts will cost approximately \$1,926 per helicopter. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$996,426, assuming 20 repetitive inspections and 20 bearing lubrications on each helicopter and bearing replacement on all the helicopters in the fleet.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their mailed comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 2002-SW-08-AD." The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States,

or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

2002-06-52 Bell Helicopter Textron

Canada: Amendment 39-12711. Docket No. 2002-SW-08-AD. Supersedes AD 2000-02-12, Docket No. 99-SW-79-AD, Amendment 39-11579.

Applicability: Model 407 helicopters, with oil cooler blower bearing, part number (P/N) 406-040-339-ALL, 407-340-339-101 or -103, installed, certificated in any category.

Note 1: This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in

accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent oil cooler blower bearing failure, loss of tail rotor drive, and a subsequent forced landing, accomplish the following:

(a) Before further flight, insert the tailwind limitation and caution, contained in Temporary Revision (TR) 9, dated January 15, 2002, into the Bell Model 407 Rotorcraft Flight Manual (RFM), dated February 9, 1996.

Note 2: TR 9 is attached to Bell Helicopter Textron (BHT) Alert Service Bulletin (ASB) 407-02-49, dated January 7, 2002.

(b) Within 10 hours time-in-service (TIS), inspect the forward and aft oil cooler blower bearings by hand-rotating the driveshaft with the oil cooler driveshaft connected. If a bearing is rough, a seal is torn, the expelled grease has turned black, or metal particles are visible in the expelled grease, replace the affected bearing before further flight.

(c) At intervals not to exceed 25 hours TIS, for oil cooler blower bearings, P/N 406-040-339-ALL and 407-340-339-103:

(1) Inspect the bearings by hand-rotating the driveshaft in accordance with the Accomplishment Instructions, Part I, paragraph 2, of BHT ASB 407-01-47, dated November 9, 2001 (ASB 407-01-47). If a bearing is rough, a seal is torn, the expelled grease has turned black, or metal particles are visible in the expelled grease, replace the affected bearing before further flight.

(2) Lubricate the bearings in accordance with the Accomplishment Instructions, Part II, paragraph 2, of ASB 407-01-47.

(d) For oil cooler blower bearings, P/N 407-340-339-101:

(1) At intervals not to exceed 25 hours TIS, inspect the bearings by hand-rotating the driveshaft in accordance with the Accomplishment Instructions, Part II, paragraph 1, of BHT ASB 407-01-44, Revision A, dated October 25, 2001 (ASB 407-01-44, Revision A). If a bearing is rough, a seal is torn, the expelled grease has turned black, or metal particles are visible in the expelled grease, replace the affected bearing before further flight.

(2) At intervals not to exceed 100 hours TIS, lubricate the bearings in accordance with the Accomplishment Instructions, Part III, paragraphs 1 and 2, of ASB 407-01-44, Revision A.

(e) Within 100 hours TIS, replace the forward and aft oil cooler blower bearings, P/N 406-040-339-ALL and 407-340-339-103, if installed, with airworthy bearings, P/N 407-340-339-101. Continue to inspect and lubricate the bearings in accordance with paragraph (d) of this AD.

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through

an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

(g) Special flight permits will not be issued.

(h) The inspections and lubrication of the oil cooler blower bearings shall be done in accordance with the Accomplishment Instructions, Part I, paragraph 2, of Bell Helicopter Textron Alert Service Bulletin 407-01-47, dated November 9, 2001 and Part II, paragraph 1, of Bell Helicopter Textron Alert Service Bulletin 407-01-44, Revision A, dated October 25, 2001. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Bell Helicopter Textron Canada, 12,800 Rue de l'Avenir, Mirabel, Quebec J7J1R4, telephone (450) 437-2862 or (800) 363-8023, fax (450) 433-0272. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2002-SW-08-AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) This amendment becomes effective on May 2, 2002, to all persons except those persons to whom it was made immediately effective by Emergency AD 2002-06-52, issued March 15, 2002, which contained the requirements of this amendment.

Note 4: The subject of this AD is addressed in Transport Canada AD CF-2002-18, dated March 4, 2002.

Issued in Fort Worth, Texas, on April 4, 2002.

David A. Downey,

Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 02-9173 Filed 4-16-02; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Docket No. FAA-2001-9559; Airspace Docket No. 01-AWP-02]

Revision of VOR Federal Airway 105 and Jet Route 86, AZ; and the Establishment of Jet Routes 614 and 616

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This action revises Federal Airway 105 (V-105) and Jet Route 86 (J-86) in the vicinity of Phoenix, AZ. The FAA is revising V-105 between the

Drake and Phoenix, AZ, Very High Frequency Omnidirectional Radio Range and Tactical Air Navigation Aids (VORTAC) in order to enhance the management of aircraft operations in the Phoenix, AZ, terminal area. Additionally, the FAA is revising J-86 between Winslow, AZ, as part of the National Airspace Redesign effort and to improve system efficiency in the Phoenix, AZ, area. The FAA is also modifying the descriptions for J-58 and J-86, and renaming portions of J-58 and J-86 in the state of Florida. These modifications are also part of the National Airspace Redesign effort to improve system efficiency.

EFFECTIVE DATE: 0901 UTC, August 8, 2002.

FOR FURTHER INFORMATION CONTACT: Ken McElroy, Airspace and Rules Division, ATA-400, Office of Air Traffic Airspace Management, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267-8783.

SUPPLEMENTARY INFORMATION:

Background

V-105

On June 20, 2001, FAA-2001-9559, Airspace Docket No. 01-AWP-02, (66 FR 30654), was published in the **Federal Register**. In that airspace docket the FAA proposed to realign V-105 and J-86 in the Phoenix, AZ, area. The June 20, 2001, Notice of Proposed Rulemaking (NPRM) contained an inadvertent error in the proposed description of V-105. Specifically, the description transposed the magnetic and true radials of V-105. A supplemental NPRM (SNPRM) corrected that error. Interested parties were invited to participate in this rulemaking proceeding by submitting written comments on the proposal to the FAA. No comments were received. With the exception of editorial changes, this amendment is the same as that proposed in the notice.

Currently the navigational signal in the vicinity of the Gulf of Mexico is not sufficient to support the segment of J-58 between the Harvey, LA, VORTAC, and the Sarasota VORTAC. The same problem affects that segment of J-86 between the Leeville VORTAC and the Sarasota, FL, VORTAC. Due to the weak navigational signal coverage on these routes, they no longer pass flight inspection. In this action, the FAA revokes the route over the Gulf, and terminates the routes at the Harvey VORTAC (for J-58) and the Leeville VORTAC (for J-86) respectively.

To replace the revoked segments, over-water advanced navigation routes