

requirements of paragraph (b) of this AD. After doing those actions, the AFM revision required by paragraph (b) of this AD may be removed from the AFM:

- (1) Install the spoiler support bracket assemblies and relays, and
- (2) Revise the spoiler lockout relay wiring.

#### Alternative Methods of Compliance

(e)(1) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

(2) Alternative methods of compliance, approved previously in accordance with AD 2001-07-10, amendment 39-12176, are approved as alternative methods of compliance with this AD.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

#### Special Flight Permits

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

#### Incorporation by Reference

(g) The actions required by paragraph shall be done in accordance with Boeing Alert Service Bulletin MD90-27A031, dated January 29, 2001, or Boeing Alert Service Bulletin MD90-27A031, Revision 01, dated March 26, 2001. The optional terminating modification specified in paragraph (c) of this AD, if accomplished, shall be done in accordance with Boeing Alert Service Bulletin MD80-27A359, dated January 29, 2001, or Boeing Alert Service Bulletin MD80-27A359, Revision 01, dated March 26, 2001. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

#### Effective Date

(h) This amendment becomes effective on March 21, 2002.

Issued in Renton, Washington, on February 5, 2002.

**Ali Bahrami,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 02-3289 Filed 2-13-02; 8:45 am]

**BILLING CODE 4910-13-U**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. 2000-NM-350-AD; Amendment 39-12512; AD 2001-23-13]**

**RIN 2120-AA64**

#### Airworthiness Directives; Boeing Model 747 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule; correction.

**SUMMARY:** This document corrects information in an existing airworthiness directive (AD) that applies to certain Boeing Model 747 series airplanes. That AD currently requires an inspection of the flap drive transmission of the trailing edge flaps at positions 2 and 7 to determine if a discrepant torque brake is installed; and corrective action, if necessary. That AD also imposes certain restrictions on the installation of affected spare parts. This document corrects and clarifies that the spares requirement in paragraph (b) of the final rule applies to only positions 2 and 7 of the trailing edge flaps, as identified in the Boeing service bulletin. This correction is necessary to ensure that operators are made aware that the spares requirement does not apply to positions 4 and 5 of the trailing edge flaps.

**DATES:** Effective December 31, 2001.

The incorporation by reference of certain publications listed in the regulations was approved previously by the Director of the Federal Register as of December 31, 2001 (66 FR 58918, November 26, 2001).

#### FOR FURTHER INFORMATION CONTACT:

Barbara Mudrovich, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2983; fax (425) 227-1181.

**SUPPLEMENTARY INFORMATION:** On November 15, 2001, the Federal Aviation Administration (FAA) issued AD 2001-23-13, amendment 39-12512 (66 FR 58918, November 26, 2001), which applies to certain Boeing Model 747 series airplanes. That AD requires

an inspection of the flap drive transmission of the trailing edge flaps at positions 2 and 7 to determine if a discrepant torque brake is installed; and corrective action, if necessary. That AD also imposes certain restrictions on the installation of affected spare parts. The actions required by that AD are intended to prevent damage to the flap system, adjacent systems, or structural components; or excessive skew of the trailing edge flap, which could result in flap asymmetry and consequent reduced controllability of the airplane.

#### Need for the Correction

Information obtained recently by the FAA indicates that the spares requirement in paragraph (b) of the final rule needs to be clarified and corrected.

As published, paragraph (b) of the final rule states that "no person shall install on any airplane any transmission or torque brake assembly identified in the "Existing Part Number" column of Paragraph 2.E. of Boeing Service Bulletin 747-27-2374, dated November 18, 1999."

Although paragraph (b) of the final rule did not limit the spares requirement to only positions 2 and 7 of the trailing edge flaps of the flap drive transmission, as indicated in the "Summary" of the final rule, and as clearly described in the "Differences" paragraph in the Notice of Proposed Rulemaking, it was the FAA's intent to do so.

The FAA has determined that a correction to AD 2001-23-13 is necessary to correct and clarify the spares requirement. This correction will specify that the spares requirement in paragraph (b) of this AD is limited to the transmission or torque brake assembly of the trailing edge flaps at positions 2 and 7.

#### Correction of Publication

This document corrects the error in AD 2001-23-13 and correctly adds the AD as an amendment to section 39.13 of the Federal Aviation Regulations (14 CFR 39.13).

The AD is reprinted in its entirety for the convenience of affected operators. The effective date of the AD remains December 31, 2001.

Since this action only clarifies and corrects a current requirement, it has no adverse economic impact and imposes no additional burden on any person. Therefore, the FAA has determined that notice and public procedures are unnecessary.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Correction**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Corrected]**

2. Section 39.13 is amended by correctly adding the following airworthiness directive (AD):

**2001-23-13 Boeing:** Amendment 39-12512. Docket 2000-NM-350-AD.

**Applicability:** Model 747 series airplanes, line numbers 0001 through 1207, certificated in any category; excluding the airplanes having line number 1174 and Model 747SP series airplanes.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent damage to the flap system, adjacent systems, or structural components; or excessive skew of the trailing edge flap; which could result in flap asymmetry and consequent reduced controllability of the airplane, accomplish the following:

**Part Verification/Replacement/Modification**

(a) Within 18 months or 7,500 flight hours after December 31, 2001, whichever occurs later: Inspect the flap drive transmission of the trailing edge flaps at positions 2 and 7 to determine if a discrepant ("Belleville" spring design) torque brake is installed in the transmission, by verifying the transmission part number, per Boeing Service Bulletin 747-27-2374, dated November 18, 1999. Then do the actions specified in paragraphs (a)(1) and (a)(2) of this AD, as applicable.

(1) If the part number of the flap drive transmission shows that no discrepant torque brake is installed, no further action is required by this AD.

(2) If the part number of the flap drive transmission shows that a discrepant torque brake may be installed, within the compliance time required by paragraph (a) of this AD: Inspect the part number of the torque brake to verify whether it is a discrepant torque brake, per the Accomplishment Instructions of the service bulletin.

(i) If the part number of the torque brake shows that it is not a discrepant torque brake, no further action is required by this AD.

(ii) If the part number of the torque brake shows that it is a discrepant torque brake: Within the compliance time required by paragraph (a) of this AD either replace the transmission with a new, improved transmission or rework the existing transmission by replacing the torque brake with a new or reworked torque brake having the part number specified in the service bulletin; per the Accomplishment Instructions of the service bulletin.

**Spares**

(b) As of December 31, 2001, no person shall install on any airplane any transmission or torque brake assembly of the trailing edge flaps at positions 2 or 7, as identified in the "Existing Part Number" column of Paragraph 2.E. of Boeing Service Bulletin 747-27-2374, dated November 18, 1999.

**Alternative Methods of Compliance**

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

**Special Flight Permit**

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

**Incorporation by Reference**

(e) The actions shall be done in accordance with Boeing Service Bulletin 747-27-2374, dated November 18, 1999. This incorporation by reference was approved previously by the Director of the Federal Register as of December 31, 2001 (66 FR 58918, November 26, 2001). Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Effective Date**

(f) The effective date of this amendment remains December 31, 2001.

Issued in Renton, Washington, on February 7, 2002.

**Ali Bahrami,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 02-3588 Filed 2-13-02; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF HEALTH AND HUMAN SERVICES****Food and Drug Administration****21 CFR Part 520****Oral Dosage Form New Animal Drugs; Carprofen**

**AGENCY:** Food and Drug Administration, HHS.

**ACTION:** Final rule.

**SUMMARY:** The Food and Drug Administration (FDA) is amending the animal drug regulations to reflect approval of a supplemental new animal drug application (NADA) filed by Pfizer, Inc. The supplemental NADA provides for a once daily, 2-milligram per pound (mg/lb) dosage of carprofen, by oral chewable tablet, for the relief of pain and inflammation associated with osteoarthritis in dogs.

**DATES:** This rule is effective February 14, 2002.

**FOR FURTHER INFORMATION CONTACT:** Melanie R. Berson, Center for Veterinary Medicine (HFV-110), Food and Drug Administration, 7500 Standish Pl., Rockville, MD 20855, 301-827-7540, e-mail: mberson@cvm.fda.gov.

**SUPPLEMENTARY INFORMATION:** Pfizer, Inc., 235 East 42d St., New York, NY 10017-5755, filed a supplement to approved NADA 141-111 that provides for veterinary prescription use of RIMADYL (carprofen) Chewable Tablets for the relief of pain and inflammation associated with osteoarthritis in dogs. The supplemental NADA provides for a once daily, 2-mg/lb dosage for the oral chewable tablet dosage form. The supplemental application is approved as of November 26, 2001, and the regulations are amended in 21 CFR 520.309 to reflect the approval. The basis of approval is discussed in the freedom of information summary.

In accordance with the freedom of information provisions of 21 CFR part 20 and 514.11(e)(2)(ii), a summary of safety and effectiveness data and information submitted to support approval of this application may be seen in the Dockets Management Branch (HFA-305), Food and Drug Administration, 5630 Fishers Lane, rm. 1061, Rockville, MD 20852, between 9