

of interest provision in § 701.21(c)(8) of this chapter.

(b) *Permissible payments.* This section does not prohibit:

(1) Payment, by you, of salary to your employees;

(2) Payment, by you, of an incentive or bonus to an employee based on your overall financial performance;

(3) Payment, by you, of an incentive or bonus to an employee, other than a senior management employee or paid official, in connection with an activity authorized by this part, provided that your board of directors establishes written policies and internal controls for the incentive program and monitors compliance with such policies and controls at least annually; and

(4) Payment, by a person other than you, of any compensation or benefit to an employee, other than a senior management employee or paid official, in connection with an activity authorized by this part, provided that your board of directors establishes written policies and internal controls regarding third-party compensation and determines that the employee's involvement does not present a conflict of interest.

(c) *Business associates and family members.* All transactions with business associates or family members not specifically prohibited by paragraph (a) of this section must be conducted at arm's length and in the interest of the credit union.

(d) *Definitions.* For purposes of this part, the following definitions apply.

(1) *Senior management employee* means your chief executive officer (typically, this individual holds the title of President or Treasurer/Manager), any assistant chief executive officers (e.g. Assistant President, Vice President, or Assistant Treasurer/Manager), and the chief financial officer (Comptroller).

(2) *Official* means any member of your board of directors, credit committee or supervisory committee.

(3) *Immediate family member* means a spouse or other family member living in the same household.

[FR Doc. 01-19103 Filed 8-3-01; 8:45 am]

BILLING CODE 7535-01-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-298-AD; Amendment 39-12355; AD 2001-15-20]

RIN 2120-AA64

Airworthiness Directives; Fokker Model F.28 Mark 1000, 2000, 3000, and 4000 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to all Fokker Model F.28 Mark 1000, 2000, 3000, and 4000 series airplanes, that requires a one-time inspection to detect the presence of filler plates of the engine support fittings, and corrective action, if necessary. The actions specified by this AD are intended to detect and correct fatigue and stress corrosion in the U-shaped upper and lower legs of the engine support fittings, which could result in reduced structural integrity of the engine support structure. This action is intended to address the identified unsafe condition.

DATES: Effective September 10, 2001.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of September 10, 2001.

ADDRESSES: The service information referenced in this AD may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Fokker Model F.28 Mark 1000, 2000, 3000, and 4000 series airplanes was published in the **Federal Register** on June 11, 2001 (66 FR 31192). That action proposed to

require a one-time inspection to detect the presence of filler plates of the engine support fittings, and corrective action, if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 22 airplanes of U.S. registry will be affected by this AD, that it will take approximately 2 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$2,640, or \$120 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is

contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2001-15-20 Fokker Services B.V.:

Amendment 39-12355. Docket 2000-NM-298-AD.

Applicability: All Model F.28 Mark 1000, 2000, 3000, and 4000 series airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To detect and correct fatigue and stress corrosion in the U-shaped upper and lower legs of the engine support fittings, which could result in reduced structural integrity of the engine support structure, accomplish the following:

Inspection

(a) Within 12 months after the effective date of this AD: Except as required by paragraph (b) of this AD, perform a general visual inspection to detect the presence of filler plates of the engine support fittings, and accomplish all applicable corrective actions (including removing any filler plates, inspecting the support fitting to detect cracks and other discrepancies by using a nondestructive test method, and repairing

discrepancies); in accordance with Fokker Service Bulletin F28/53-149, dated November 15, 1999.

Note 2: For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or drop-light, and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

(b) If the service bulletin specifies to contact Fokker Services for appropriate action: Prior to further flight repair in accordance with a method approved by the Manager, International Branch, ANM-116, FAA Transport Airplane Directorate; or the Rijksluchtvaartdienst (RLD) (or its delegated agent).

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) Except as provided by paragraph (b) of this AD: The actions shall be done in accordance with Fokker Service Bulletin F28/53-149, dated November 15, 1999. This 2 incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 4: The subject of this AD is addressed in Dutch airworthiness directive 1999-153, dated November 30, 1999.

Effective Date

(f) This amendment becomes effective on September 10, 2001.

Issued in Renton, Washington, on July 26, 2001.

Vi L. Lipski,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 01-19246 Filed 8-3-01; 8:45 am]

BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-412-AD; Amendment 39-12356; AD 2001-15-21]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2 and B4; A310; and A300 B4-600, B4-600R, and F4-600R (Collectively Called A300-600) Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Airbus Model A300 B2 and B4; A310; and A300 B4-600, B4-600R, and F4-600R (collectively called A300-600) series airplanes; that requires modification of certain components related to the fuel level sensors. This action is necessary to prevent the possibility of overheating of the fuel level sensors, which could lead to the risk of explosion in the fuel tank. This action is intended to address the identified unsafe condition.

DATES: Effective September 10, 2001.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of September 10, 2001.

ADDRESSES: The service information referenced in this AD may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1175; fax (425) 227-1149.