

and regulatory requirements of the Federal government. At the same time, the Administrative Committee is constantly engaged in efforts to improve the quality of our online publications, including investments in new technology applications that will enhance e-government services to the public. In addition, GPO recently took new steps to significantly increase server capacity to meet the growing demand for online access to **Federal Register** publications. For members of the public who prefer to read the printed editions, GPO continues to provide free access to **Federal Register** publications at Federal depository libraries located throughout the nation under funding provided by Congress.

The changes to subscription prices in this proposal amount to a 9.6 percent increase for the paper edition of the **Federal Register** and a 9.2 percent increase for the printed CFR. The subscription price of the Weekly Compilation of Presidential Documents would increase by 12 percent. The price changes are reflected in proposed amendments to 1 CFR part 11. The following rates would be effective after the issuance of the final rule. The annual subscription rate for the daily **Federal Register** paper edition would increase from \$638 to \$699. For a combined **Federal Register**, **Federal Register** Index and LSA (List of CFR Sections Affected) subscription, the rate would increase from \$697 to \$764. The price of a single copy of the daily **Federal Register** paper edition would increase from \$9 to \$10. The annual subscription rate for the microfiche edition of the **Federal Register**, which includes the **Federal Register** Index and LSA, would increase from \$253 to \$264. The annual subscription price for the **Federal Register** Index increases from \$28 to \$30. The annual subscription price for the monthly LSA would increase from \$31 to \$35. The annual subscription rate for a full set of the CFR paper edition would increase from \$1094 to \$1195. The annual subscription rate for the microfiche edition of the CFR would increase from \$290 to \$298. The annual subscription rates for the Weekly Compilation of Presidential Documents would increase from \$92 to \$103 for delivery by non-priority mail and from \$151 to \$169 for delivery by first-class mail.

Regulatory Analysis

Executive Order 12866

The proposed rule has been drafted in accordance with Executive Order 12866, section 1(b), "Principles of Regulation." The Administrative Committee has

determined that this proposed rulemaking is not a significant regulatory action, as defined under section 3(f) of Executive Order 12866.

Regulatory Flexibility Act

The Regulatory Flexibility Act, 5 U.S.C. 601 *et seq.*, does not apply to rate increases necessary to recover the costs to the Government for printing and distributing **Federal Register** publications. This action would not have a significant impact on small entities since it would not impose any substantive requirements, and any increased costs could be avoided by accessing **Federal Register** publications through the free GPO Access service on the Internet or at a Federal depository library.

Federalism

This proposed rule has no Federalism implications under Executive Order 13132. It would not impose compliance costs on State or local governments or preempt State law.

List of Subjects in 1 CFR Part 11

Code of Federal Regulations, **Federal Register**, Government publications, Weekly Compilation of Presidential Documents.

For the reasons discussed in the preamble, the Administrative Committee of the Federal Register proposes to amend part 11 of chapter I of title 1 of the Code of Federal Regulations as set forth below:

PART 11—SUBSCRIPTIONS

1. The authority citation for part 11 continues to read as follows:

Authority: 44 U.S.C. 1506; sec. 6, E.O. 10530, 19 FR 2709, 3 CFR, 1954–1958 Comp., p. 189.

2. In § 11.2, revise paragraph (a) to read as follows:

§ 11.2 Federal Register.

(a) The subscription price for the paper edition of the daily **Federal Register**, including postage, is \$699 per year. A combined subscription to the daily **Federal Register**, the monthly **Federal Register** Index, and the monthly LSA (List of CFR Sections Affected), including postage, is \$764 per year for the paper edition, or \$264 per year for the microfiche edition. Six-month subscriptions to the paper and microfiche editions are also available at one-half the annual rate. Limited quantities of current or recent issues may be purchased for \$10 per copy for the paper edition, or \$2 per copy for the microfiche edition.

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3. In § 11.3, revise paragraph (a) to read as follows:

§ 11.3 Code of Federal Regulations.

(a) The subscription price for a complete set of the Code of Federal Regulations, including postage, is \$1195 per year for the bound, paper edition, or \$298 per year for the microfiche edition. The Government Printing Office sells individual volumes of the paper edition of the Code of Federal Regulations at prices determined by the Superintendent of Documents under the general direction of the Administrative Committee. The price of a single volume of the microfiche edition is \$2 per copy.

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4. In § 11.6, revise paragraph (a) to read as follows:

§ 11.6 Weekly Compilation of Presidential Documents.

(a) The subscription price for the paper edition of the Weekly Compilation of Presidential Documents is \$103 per year for delivery by non-priority mail, or \$169 per year for delivery by first-class mail. The price of an individual copy is \$4.

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5. Revise § 11.7 to read as follows:

§ 11.7 Federal Register Index.

The annual subscription price for the monthly **Federal Register** Index, purchased separately, in paper form, is \$30.

6. Revise § 11.8 to read as follows:

§ 11.8 LSA (List of CFR Sections Affected).

The annual subscription price for the monthly LSA (List of CFR Sections Affected), purchased separately, in paper form, is \$35.

By Order of the Committee.

Raymond A. Mosley,

Secretary, Administrative Committee of the Federal Register.

[FR Doc. 01–14217 Filed 6–5–01; 8:45 am]

BILLING CODE 1505–02–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99–NE–39–AD]

Airworthiness Directives; CFE Company Model CFE738–1–1B Turbofan Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: The Federal Aviation Administration (FAA) proposes to revise an existing airworthiness directive (AD), applicable to CFE Company Model CFE738-1-1B turbofan engines. That AD currently requires a one-time visual inspection of Stage 2 high pressure turbine (HPT) aft cooling plates for nicks, dents, raised metal, and scratches, and if necessary, repair of the cooling plates or replacement with serviceable parts. This proposal is prompted by an updated alert service bulletin (ASB) that reduces the number of stage 2 HPT aft cooling plates affected by this AD and identifies the applicable engines by engine serial numbers (SN's). In the event that the affected gas generator modules containing the affected parts have been transferred to another engine, the SN's of those gas generator modules have been added to this AD. The actions specified by the proposed AD are intended to prevent stage 2 HPT aft cooling plate failure, which could result in an uncontained engine failure and damage to the airplane.

DATES: Comments must be received by August 6, 2001.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 99-NE-39-AD, 12 New England Executive Park, Burlington, MA 01803-5299. Comments may also be sent via the Internet using the following address: "9-ane-adcomment@faa.gov." Comments sent via the Internet must contain the docket number in the subject line. Comments may be inspected at this location between 8:00 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from CFE Company, Data Distribution, MS 64-03/2101-201, P.O. Box 29003, Phoenix, AZ 85038-9003; telephone (602) 365-2493, fax (602) 365-5577. This information may be examined at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

FOR FURTHER INFORMATION CONTACT: Keith Mead, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone: (781) 238-7744, fax: (781) 238-7199.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the

proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 99-NE-39-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRM's

Any person may obtain a copy of this NPRM by submitting a request to the FAA, New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 99-NE-39-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

Discussion

On December 29, 1999, the FAA issued AD 99-27-16, Amendment 39-11497 (65 FR 691, January 6, 2000), to require a one-time visual inspection of 72 Stage 2 HPT aft cooling plates for nicks, dents, raised metal, and scratches. If any nicks, dents, raised metal, or scratches were found, that AD also required repair of the cooling plates or replacement with serviceable parts. That action was prompted by reports of Stage 2 HPT aft cooling plates that were dented during the assembly of the cooling plate to the Stage 2 disk due to raised metal on the stage 2 HPT disk post aft mating surface. The raised metal condition on the stage 2 HPT disk post was caused by an improperly installed locating pin in the assembly tool. That condition, if not corrected, could result in HPT aft cooling plate failure, which could result in an uncontained engine failure and damage to the airplane. Since AD 99-27-16 was issued, CFE

Company has isolated the observed assembly damage to a known replacement of the locating pin in the assembly tooling. It has been determined that only ten of the original 72 HPT cooling plate/stage 2 disk assemblies were assembled before tool correction occurred. Records show these ten HPT cooling plate/stage 2 disk assemblies were subsequently assembled in ten gas generator modules, S/N's 800421-800430. These gas generator modules, along with the engines on which these modules were installed, are listed in this AD.

Manufacturer's Service Information

The FAA has reviewed and approved the technical contents of CFE Company Alert Service Bulletin (ASB) CFE738-A72-8031, Revision 2, dated October 17, 2000, that describes the dimensional inspection procedures for indentation depth on stage 2 HPT aft cooling plates, inspection of the stage 2 HPT rotor disk for raised metal, and the acceptance and repair criteria of the stage 2 HPT aft cooling plate and HPT rotor disk.

Differences Between This AD and the Manufacturer's Service Information

Although ASB CFE738-A72-8031, Revision 2, dated October 17, 2000, identifies ten engines by SN, the FAA is also listing the ten gas generator module SN's in which the affected HPT cooling plate/stage 2 HPT disk assemblies are installed. This listing is made to assure that the gas generator modules containing the affected parts are identified and inspected regardless of which engine they are installed on. Gas generator modules may not necessarily stay installed in one and only one engine.

FAA's Determination of an Unsafe Condition and Proposed Actions

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require, on gas generator modules identified by SN, a one-time visual inspection of stage 2 HPT aft cooling plates for nicks, dents, raised metal, and scratches, and if present, replacement with serviceable parts, or, dimensional inspection of indentation depth, repair if indentation is within acceptable limits, and if not, replacement with serviceable parts. This AD would also require inspection of the stage 2 HPT rotor disk post aft surface which mates with the stage 2 HPT aft cooling plate, for raised metal, and, removal of the raised metal, if present, or replacement with a serviceable part. The inspections would be required at the next shop visit

after the effective date of this AD where the HPT assembly is sufficiently disassembled to afford access to the stage 2 HPT aft cooling plate, but not later than 4,500 part cycles-since-new (CSN) in accordance with the ASB described previously.

Economic Impact

There are approximately ten engines of the affected design in the worldwide fleet. The FAA estimates that nine engines installed on airplanes of US registry would be affected by this proposed AD, that it would take approximately four work hours per engine to accomplish the proposed inspection if the inspection did not take place during scheduled maintenance, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$1,536 per engine. Based on these figures, the total cost impact of the proposed AD on US operators is estimated to be \$15,484.

Regulatory Impact

This proposed rule does not have federalism implications, as defined in Executive Order 13132, because it would not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Accordingly, the FAA has not consulted with state authorities prior to publication of this proposed rule.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39–11497 (65 FR 691, January 6, 2000), and by adding a new airworthiness directive, to read as follows:

CFE Company: Docket No. 99–NE–39–AD. Revises AD 99–27–16, Amendment 39–11497.

Applicability

This airworthiness directive (AD) is applicable to CFE Model CFE738–1–1B turbofan engines, part number (P/N) 3050000–5, with gas generator modules P/N 6091T09G01, serial numbers (SN's) 800421, 800422, 800423, 800424, 800425, 800426, 800427, 800428, 800429, and 800430 installed. These modules are currently installed in engine SN's 105323, 105324, 105325, 105326, 105328, 105329, 105331, 105332, 105333, and 105392. These engines are installed on, but not limited to Dassault-Breguet Falcon 2000 series airplanes.

Note 1: This AD applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance

Compliance with this AD is required as indicated, unless already done.

To prevent stage 2 high pressure turbine (HPT) aft cooling plate failure, which could result in an uncontained engine failure and damage to the airplane, accomplish the following:

Inspections and Follow-On Actions

(a) At the next engine shop visit after the effective date of this AD where the HPT assembly is sufficiently disassembled to afford access to the Stage 2 HPT aft cooling plate, but not later than 4,500 part cycles-since-new (CSN), accomplish the following:

(1) Inspect the stage 2 HPT aft cooling plate for nicks, dents, and scratches on surface D in accordance with the requirements of CFE Alert Service Bulletin (ASB) No. CFE738–A72–8031, Revision 2, dated October 17, 2000, paragraph 2.B.(1).

(2) Repair those stage 2 HPT aft cooling plates with indentation 0.003 inch deep or less in accordance with ASB No. CFE738–

A72–8031, Revision 2, dated October 17, 2000, paragraph 2.B.(1).

(3) Remove from service prior to further flight those stage 2 HPT aft cooling plates that have nicks, dents, and/or scratches that exceed the acceptance limits in accordance with ASB No. CFE738–A72–8031, Revision 2, dated October 17, 2000, paragraph 2.B.(1), and replace with serviceable parts.

(4) Inspect the stage 2 HPT rotor disk post aft mating surface for raised metal, and remove raised metal if present in accordance with ASB No. CFE738–A72–8031, Revision 2, dated October 17, 2000, paragraph 2.B.(2).

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators shall submit their request through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the inspection requirements of this AD can be accomplished.

Issued in Burlington, Massachusetts, on May 25, 2001.

Jay J. Pardee,

Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 01–14146 Filed 6–5–01; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 96–NM–143–AD]

RIN 2120–AA64

Airworthiness Directives; Gulfstream Aerospace Corporation Model G–159 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all Gulfstream Aerospace Corporation Model G–159 airplanes. This proposal would require repetitive non-destructive testing inspections to detect corrosion of the skin of certain structural assemblies, and corrective action, if necessary. This