estimates that 24 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 2 work hours per airplane to accomplish the proposed actions, and that the average labor rate is $60 per work hour. Required parts would be supplied by the airplane manufacturer at no cost to the operators. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be $2,880, or $120 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39
Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:


Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent possible overheating of cargo control unit (C–U) components, which could result in smoke and/or fire in the cargo compartment, accomplish the following:

Replacement or Reidentification

(a) Within 6 months after the effective date of this AD, do the applicable actions specified in paragraphs (a)(1) and (a)(2) of this AD per Boeing Alert Service Bulletin MD11–24A189, dated June 22, 2000.

(1) For airplanes identified as Group 1 and Group 2 in the service bulletin: Replace the cargo roller circuit breakers with new circuit breakers.

(2) For airplane identified as Group 2 in the service bulletin: Reidentify the aft circuit breaker panel.

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permit

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.


Vi L. Lipski,
Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 01–4051 Filed 2–16–01; 8:45 am]

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000–NM–185–AD]

RIN 2120–AA64

Airworthiness Directives; McDonnell Douglas Model MD–11 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model MD–11 series airplanes. This proposal would require a one-time general visual inspection of the electrical wiring of the right side of the cockpit to determine if the electrical wiring is chafing against the observer station and to detect damaged wires; and corrective actions, if necessary. This action is necessary to prevent chafing and damage to electrical wires of the cockpit and consequent electrical arcing due to wires that were routed improperly during production of the airplane, which could result in fire and smoke in the airplane. This action is intended to address the identified unsafe condition.

DATES: Comments must be received by April 6, 2001.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 2000–NM–185–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be submitted at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227–1232. Comments may be submitted via fax to (425) 227–
The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1–L51 (2–60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT:

SUPPLEMENTARY INFORMATION:

Comments Invited
Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:
• Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
• For each issue, state what specific change to the proposed AD is being requested.
• Include justification (e.g., reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: “Comments to Docket Number 2000–NM–185–AD.” The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Discussion
As part of its practice of re-examining all aspects of the service experience of a particular aircraft whenever an accident occurs, the FAA has become aware of an accident in which, during an inspection, electrical wires were found chafing in the right side of the cockpit against the observer station. This incident occurred in a McDonnell Douglas Model MD–11 series airplane. Investigation revealed that the wires were routed improperly during production of the airplane. This condition, if not corrected, could result in chafing and damage to electrical wires of the cockpit and consequent electrical arcing, which could result in fire and smoke in the airplane. This incident is considered to be related to an accident that occurred off the coast of Nova Scotia involving a McDonnell Douglas Model MD–11 series airplane. The cause of that accident is still under investigation.

Other Related Rulemaking
The FAA, in conjunction with Boeing and operators of Model MD–11 series airplanes, is continuing to review all aspects of the service history of those airplanes to identify potential unsafe conditions and to take appropriate corrective actions. This proposed airworthiness directive (AD) is one of a series of actions identified during that process. The process is continuing and the FAA may consider additional rulemaking actions as further results of the review become available.

Explanation of Relevant Service Information
The FAA has reviewed and approved Boeing Alert Service Bulletin MD11–24A117, dated May 18, 2000, which describes procedures for a one-time general visual inspection of the electrical wiring of the right side of the cockpit to determine if the electrical wiring is chafing against the observer station and to detect damaged wires; and corrective actions, if necessary. The corrective actions include loosening wire clamps; repositioning wires; tightening wire clamps; repairing damaged insulation; and replacing damaged wires with new wires. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition.

Explanation of Requirements of Proposed Rule
Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously.

Cost Impact
There are approximately 148 Model MD–11 series airplanes of the affected design in the worldwide fleet. The FAA estimates that 43 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 1 work hour per airplane to accomplish the proposed inspection, and that the average labor rate is $60 per work hour. Based on these figures, the cost impact of the inspection proposed by this AD on U.S. operators is estimated to be $2,580, or $60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact
The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal
would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:


Applicability: Model MD–11 series airplanes, as listed in Boeing Alert Service Bulletin MD11–24A117, dated May 18, 2000; certificate in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent chafing and damage to electrical wires of the cockpit and consequent electrical arcing due to wires that were routed improperly during production of the airplane, which could result in fire and smoke in the airplane, accomplish the following:

One-Time General Visual Inspection

(a) Within 6 months after the effective date of this AD, do a one-time general visual inspection of the electrical wiring of the right side of the cockpit to determine if the electrical wiring is chafing against the observer station and to detect damaged wires, per Boeing Alert Service Bulletin MD11–24A117, dated May 18, 2000.

Note 2: For the purposes of this AD, a general visual inspection is defined as “A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or drop-light, and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked.”

Condition 1 (No Chafing)

(b) If all electrical wires are found not to be chafing against the observer station during the inspection required by paragraph (a) of this AD, no further action is required by this AD.

Condition 2 (Chafing and No Wire Damage)

(c) If any electrical wire is found to be chafing against the observer station and if no wire is found damaged during the inspection required by paragraph (a) of this AD, before further flight, loosen the wire clamps, reposition the wires, and tighten the wire clamps, per Boeing Alert Service Bulletin MD11–24A117, dated May 18, 2000.

Condition 3 (Chafing and Wire Damage)

(d) If any electrical wire is found to be chafing against the observer station and if any wire is found damaged during the inspection required by paragraph (a) of this AD, before further flight, do the action specified in paragraph (d)(1) or (d)(2) of this AD, as applicable, AND do the action specified in paragraph (d)(3) of this AD; per Boeing Alert Service Bulletin MD11–24A117, dated May 18, 2000.

(1) For damage within repairable limits: Repair damaged insulation.

(2) For damage outside repairable limits: Replace damaged wires with new wires.

(3) Loosen the wire clamps, reposition the wires, and tighten the wire clamps.

Alternative Methods of Compliance

(e) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permit

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.


Vi L. Lipski, Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 01–4050 Filed 2–16–01; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Airspace Docket No. 00–AEA–16]

Establishment of Class E Airspace; South Albany, NY

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking.

SUMMARY: This notice proposes to establish Class E airspace at South Albany, NY. An Area Navigation (RNAV) GPS approach, has been developed for South Albany Airport, South Bethlehem, NY. Controlled airspace extending upward from 700 feet Above Ground Level (AGL) is needed to contain aircraft executing an instrument approach. The area would be depicted on aeronautical charts for pilot reference.

DATES: Comments must be received on or before March 22, 2001.

ADDRESSES: Send comments on the proposal in triplicate to: Manager, Airspace Branch, AEA–520, Docket No. 00–AEA–16, F.A.A. Eastern Region, 1 Aviation Plaza, Jamaica, NY 11434–4809.

The official docket may be examined in the Office of the Regional Counsel, AEA–7, F.A.A. Eastern Region, 1 Aviation Plaza, Jamaica, NY 11434–4809. An informal docket may also be examined during normal business hours in the Airspace Branch, AEA–520, F.A.A. Eastern Region, 1 Aviation Plaza, Jamaica, NY 11434–4809.

FOR FURTHER INFORMATION CONTACT: Mr. Francis T. Jordan, Jr., airspace Specialist, Airspace Branch, AEA–520, F.A.A. Eastern Region, 1 Aviation Plaza, Jamaica, NY 11434–4809; telephone: (718) 553–4521.

SUPPLEMENTARY INFORMATION: