

available to supply these products to the Federal Government. The effect of a waiver would be to allow an otherwise qualified Nonmanufacturer to supply other than the product of a domestic small business manufacturer or processor on a Federal contract set aside for small businesses or awarded through the SBA 8(a) Program. The purpose of this document is to solicit comments and potential source information from interested parties.

DATES: Comments and sources must be submitted on or before December 12, 2000.

ADDRESSES: Edith Butler, Program Analyst, U.S. Small Business Administration, 409 3rd Street SW., Washington, DC 20416, Tel: (202) 619-0422.

FOR FURTHER INFORMATION CONTACT: Edith Butler, Program Analyst, (202) 619-0422, FAX (202) 205-6845.

SUPPLEMENTARY INFORMATION: Public Law 100-656, enacted on November 15, 1988, incorporated into the Small Business Act the previously existing regulation that recipients of Federal contracts set-aside for small businesses or the SBA 8(a) Program procurement must provide the product of a small business manufacturer or processor, if the recipient is other than the actual manufacturer or processor. This requirement is commonly referred to as the Nonmanufacturer Rule. The SBA regulations imposing this requirement are found a 13 CFR 121.406(b). Section 303(h) of the law provides for waiver of this requirement by SBA for any "class of products" for which there are no small business manufacturers or processors in the Federal market. To be considered available to participate in the Federal market on these classes of products, a small business manufacturer must have submitted a proposal for a contract solicitation or received a contract from the Federal Government within the last 24 months. The SBA defines "class of products" based on two coding systems. The first is the Office of Management and Budget North American Industry Classification System (NAICS). The second is the Product and Service Code (PSC) established by the Federal Procurement Data System.

The Small Business Administration is currently processing a request for a waiver of the Nonmanufacturer Rule for Surge Arresters (SIC 3643, NAICS 335931), Current and Voltage Transformers (SIC 3612, NAICS 335311), Disconnect Switches (SIC 3613, NAICS 335313), Sutorransformers (SIC 3612, NAICS 335311), Power Transformers (multiple winding type)

(SIC 3612, NAICS 335311), Insulator Assemblies for transmission lines (porcelain and polymer type) (SIC 3264/3644, NAICS 327113/335932), and Stacking Post Insulators (SIC 3264, NAICS 3327113), and invites the public to comment or provide information on potential small business manufacturers for these products.

In an effort to identify potential small business manufacturers, the SBA has searched Procurement Marketing & Access Network (PRO-Net) and the SBA will publish a notice in the Commerce Business Daily. The public is invited to comment or provide source information to SBA on the proposed waiver of the Nonmanufacturer Rule for these classes of products.

Dated: November 21, 2000.

Luz A. Hopewell,
Associate Administrator for Government Contracting.

[FR Doc. 00-30779 Filed 12-5-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-260-AD]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model DC-9-81, -9-82, -9-83, and -9-87 Series Airplanes; Model MD-88 Airplanes; and Model MD-90-30 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model DC-9-81, -9-82, -9-83, and -9-87 series airplanes; Model MD-88 airplanes; and Model MD-90-30 series airplanes.

This proposal would require repetitive inspections of the number 1 and 2 electric motors of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage; and corrective actions, if necessary. This action is necessary to prevent various failures of electric motors of the auxiliary hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure.

This action is intended to address the identified unsafe condition.

DATES: Comments must be received by January 22, 2001.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-260-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 2000-NM-260-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1-L51 (2-60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: Albert Lam, Aerospace Engineer, Systems and Equipment Branch, ANM-130L, the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712; telephone (526) 627-5346; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this action may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.

- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this action must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 2000-NM-260-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 2000-NM-260-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received reports that, during ground operations or when powered in flight by the air driven generator, the electric motors of the auxiliary hydraulic pump and associated motor feeder cables failed on certain McDonnell Douglas Model MD-80, DC-10, MD-10, MD-11, and MD-90-30 series airplanes. These failures consisted of seized or difficult to turn rotor on the pump assembly, burnt and shorted motor feeder cables, and/or uncontained internal electric arcing failures with the electric motor. Investigation revealed that these failures may be caused by hydraulic fluid contamination to the electric motor portion of the pump, failed rotor bearing, and/or degradation of the stator's encapsulate material. These conditions, if not corrected, could result in a fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure.

Other Relevant Rulemaking

This proposed AD affects McDonnell Douglas Model DC-9-81, -9-82, -9-83,

and -9-87 series airplanes (*i.e.*, MD-80 series airplanes); Model MD-88 airplanes; and Model MD-90-30 series airplanes. The FAA is planning to issue a separate rulemaking action for McDonnell Douglas Model DC-10 series airplanes, Model MD-10 series airplanes, and Model MD-11 series airplanes to address the identified unsafe condition.

Explanation of Relevant Service Information

The FAA has reviewed and approved McDonnell Douglas Alert Service Bulletin MD80-29A067, dated October 21, 1999 (for Model DC-9-81, -9-82, -9-83, and -9-87 series airplanes, and Model MD-88 airplanes); and McDonnell Douglas Alert Service Bulletin MD90-29A018, dated October 21, 1999 (for Model MD-90-30 series airplanes). These service bulletins describe procedures for repetitive inspections of the number 1 and 2 electric motors of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage; and corrective actions, if necessary. The corrective actions involve replacing the auxiliary hydraulic pump with a serviceable pump, troubleshooting, and repairing the wiring.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletins described previously.

Interim Action

This is considered to be interim action until final action is identified, at which time the FAA may consider further rulemaking.

Cost Impact

There are approximately 1,292 Model DC-9-81, -9-82, -9-83, and -9-87 series airplanes; Model MD-88 airplanes; and Model MD-90-30 series airplanes of the affected design in the worldwide fleet. The FAA estimates that 697 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 1 work hour per airplane to accomplish the proposed inspection, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$41,820 or \$60 per airplane, per inspection cycle.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this proposed AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

McDonnell Douglas: Docket 2000–NM–260–AD.

Applicability: Model DC–9–81, –9–82, –9–83, and –9–87 series airplanes, and Model MD–88 airplanes, as listed in McDonnell Douglas Alert Service Bulletin MD80–29A067, dated October 21, 1999; and Model MD–90–30 series airplanes, as listed in McDonnell Douglas Alert Service Bulletin MD90–29A018, dated October 21, 1999; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent various failures of electric motors of the auxiliary hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure, accomplish the following:

Inspection

(a) Do a detailed inspection of the number 1 and 2 electric motors of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage, per McDonnell Douglas Alert Service Bulletin MD80–29A067, dated October 21, 1999 (for Model DC–9–81, –9–82, –9–83, and –9–87 series airplanes, and Model MD–88 airplanes); or McDonnell Douglas Alert Service Bulletin MD90–29A018, dated October 21, 1999 (for Model MD–90–30 series airplanes); as applicable; at the applicable time specified in paragraph (a)(1) or (a)(2) of this AD.

(1) For airplanes that have accumulated 3,000 total flight hours or more as of the effective date of this AD: Inspect within 12 months after the effective date of this AD.

(2) For airplanes that have accumulated less than 3,000 total flight hours as of the effective date of this AD: Inspect within 12 months after accumulating 3,000 total flight hours.

Condition 1, No Failures: Repetitive Inspections

(b) If no failures are detected during the inspection required by paragraph (a) of this AD, repeat the inspection required by paragraph (a) of this AD every 5,000 flight hours.

Condition 2, Failure of Any Pump Motor: Replacement and Repetitive Inspections

(c) If any pump motor fails during any inspection required by paragraph (a) of this

AD, before further flight, replace the failed auxiliary hydraulic pump with a serviceable pump, per McDonnell Douglas Alert Service Bulletin MD80–29A067, dated October 21, 1999 (for Model DC–9–81, –9–82, –9–83, and –9–87 series airplanes, and Model MD–88 airplanes); or McDonnell Douglas Alert Service Bulletin MD90–29A018, dated October 21, 1999 (for Model MD–90–30 series airplanes); as applicable. Repeat the inspection required by paragraph (a) of this AD every 5,000 flight hours.

Condition 3, Failure of Any Wiring: Repair and Repetitive Inspection

(d) If any wiring fails during any inspection required by paragraph (a) of this AD, before further flight, troubleshoot and repair the failed wiring, per McDonnell Douglas Alert Service Bulletin MD80–29A067, dated October 21, 1999 (for Model DC–9–81, –9–82, –9–83, and –9–87 series airplanes, and Model MD–88 airplanes); or McDonnell Douglas Alert Service Bulletin MD90–29A018, dated October 21, 1999 (for Model MD–90–30 series airplanes); as applicable. Repeat the inspection required by paragraph (a) of this AD every 5,000 flight hours.

Alternative Methods of Compliance

(e) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

Special Flight Permit

(f) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on November 30, 2000.

Donald L. Riggan,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00–31067 Filed 12–5–00; 8:45 am]

BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000–NE–12–AD]

RIN 2120–AA64

Airworthiness Directives; Turbomeca S.A. Arrius Models 2B, 2B1, 2F Turboshaft Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to Turbomeca S.A. Arrius Models 2B, 2B1, and 2F turboshaft engines. This proposal would require the replacement of the right injector half manifold, left injector half manifold, and privilege injector pipe with the engine installed on the helicopter. This proposal is prompted by reports from the Direction Generale de L'Aviation Civile (DGAC), which is the airworthiness authority for France, of partially or totally blocked fuel injection manifolds, which were found during inspections at a repair workshop. The actions specified by the proposed AD are intended to prevent engine flameout during rapid deceleration, or the inability to maintain the 2.5 minutes one engine inoperative (OEI) rating. The actions are also intended to prevent injector air path cracks, due to blockage of the fuel injection manifolds.

DATES: Comments must be received by February 5, 2001.

ADDRESSES: Submit comments to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 2000–NE–12–AD, 12 New England Executive Park, Burlington, MA 01803–5299. Comments may also be sent via the Internet using the following address: “9-ane-adcomment@faa.gov”. Comments sent via the Internet must contain the docket number in the subject line. Comments may be inspected at this location between 8 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Turbomeca S.A., 40220 Tarnos, France; telephone: (33) 05 59 64 40 00; fax: (33) 05 59 64 60 80. This information may be examined at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.