

and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

2000-24-01 Bombardier, Inc. (Formerly Canadair): Amendment 39-12008. Docket 2000-NM-368-AD.

Applicability: The following airplanes, certificated in any category:

Model	Serial Nos.
CL-600-1A11 (CL-600).	1004 though 1085 inclusive.
CL-600-2A12 (CL-601).	3001 through 3066 inclusive.
CL-600-2B16 (CL-601-3A and CL-601-3R).	5001 through 5194 inclusive.
CL-600-2B16 (CL-604).	5301 through 5392 inclusive.

Note 1: This AD applies to each airplane identified in the preceding applicability

provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent water, ice, or slush accumulation on the aileron quadrants and/or control cable pulleys in the wheel bay of the main landing gear (MLG) during ground roll, which could subsequently freeze during the climb to cruise altitude and cause stiffness in the aileron controls and reduced controllability of the airplane, accomplish the following:

Installation of Protection Shields

(a) Within 45 days after the effective date of this AD, install protection shields in the wheel bay of the MLG, per the following applicable Bombardier service bulletin:

Model	Bombardier service bulletin	Service bulletin date
CL-600-1A11 (CL-600)	600-0684	July 15, 1998
CL-600-2A12 (CL-601), and CL-600-2B16 (CL-601-3A and CL-601-3R).	601-0507	June 30, 1998
CL-600-2B16 (CL-604)	604-32-007	June 30, 1998

Alternative Methods of Compliance

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, New York Aircraft Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, New York ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the New York ACO.

Special Flight Permits

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(d) The actions shall be done in accordance with the following service bulletins, as applicable:

Bombardier service bulletin	Service bulletin date
600-0684	July 15, 1998.

Bombardier service bulletin	Service bulletin date
601-0507	June 30, 1998.
604-32-007	June 30, 1998.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Bombardier, Inc., Canadair, Aerospace Group, P.O. Box 6087, Station Centre-ville, Montreal, Quebec H3C 3G9, Canada. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, New York Aircraft Certification Office, 10 Fifth Street, Third Floor, Valley Stream, New York; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in Canadian airworthiness directive CF-2000-30, dated September 12, 2000.

Effective Date

(e) This amendment becomes effective on December 19, 2000.

Issued in Renton, Washington, on November 17, 2000.

Donald L. Riggins,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-30020 Filed 12-1-00; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-CE-42-AD; Amendment 39-11965; AD 2000-22-18]

RIN 2120-AA64

Airworthiness Directives; Raytheon Aircraft Company Beech Model 58 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Raytheon Aircraft Company (Raytheon) Beech Model 58

airplanes. This AD requires you to inspect the rudder bellcrank interconnect tube for damage; replace or refinish the interconnect tube, if necessary; and modify the floorboard. Four reports of damage to the interconnect tube prompted this action. The actions specified by this AD are intended to correct the wrong use of screws and consequent wear in the pilot/copilot pedal interconnect tube, which could result in loss of rudder control.

DATES: This AD becomes effective on December 29, 2000.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of December 29, 2000.

ADDRESSES: You may get the service information referenced in this AD from Raytheon Aircraft Company, P.O. Box 85, Wichita, Kansas 67201-0085; telephone: (800) 429-5372 or (316) 676-3140; on the Internet at <<http://www.raytheon.com/rac/servinfo/27-3013.pdf>>. This file is in Adobe Portable Document Format. The Acrobat Reader is available at <http://www.adobe.com/>. You may examine this information at the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2000-CE-42-AD, 901 Locust, Room 506, Kansas City, Missouri 64106; or at the Office of the

Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Paul C. DeVore, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Mid-Continent Airport, Wichita, Kansas 67209; telephone: (316) 946-4142; facsimile: (316) 946-4407.

SUPPLEMENTARY INFORMATION:

Discussion

What events have caused this AD? The FAA has received four reports of grooves cut in the pilot/copilot rudder interconnect tube. The grooves were discovered during routine inspections.

What are the consequences if the condition is not corrected? This condition could result in jamming or restricting rudder control.

Has FAA taken any action to this point? We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Raytheon Beech Model 58 airplanes. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on September 7, 2000 (65 FR 54184). The NPRM proposed to require you to inspect the rudder bellcrank interconnect tube for damage; if necessary, replace or refinish the rudder bellcrank interconnect tube; and plug the floorboard screw hole.

Was the public invited to comment? Interested persons were afforded an opportunity to participate in the making of this amendment. No comments were received on the proposed rule or the FAA's determination of the cost to the public.

The FAA's Determination

What is FAA's Final Determination on this Issue? After careful review of all available information related to the subject presented above, we have determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. We determined that these minor corrections:

- will not change the meaning of the AD; and
- will not add any additional burden upon the public than was already proposed.

Cost Impact

How many airplanes does this AD impact? We estimate that this AD affects 491 airplanes in the U.S. registry.

What is the cost impact of this AD on owners/operators of the affected airplanes? We estimate the following costs to accomplish the modification of the floorboard and inspection of the rudder bellcrank interconnect tube:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. airplane operators
4 workhours × \$60 per hour = \$240.	Parts are provided at no charge under warranty.	\$240 per airplane	\$240 × 491 = \$117,840.

We estimate the following costs to accomplish the modification or replacement of the bellcrank interconnect tube:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. airplane operators
1 workhour × \$60 per hour = \$60	Parts are provided at no charge under warranty.	\$60 per airplane	\$60 × 491 = \$29,460.

Note: The manufacturer will allow warranty credit for labor and parts to the extent noted in the service bulletin.

Regulatory Impact

Does this AD impact various entities? The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

Does this AD involve a significant rule or regulatory action? For the reasons discussed above, I certify that this action (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the

Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by Reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. FAA amends § 39.13 by adding a new AD to read as follows:

2000-22-18 Raytheon Aircraft Company: Amendment 39-11965; Docket No. 2000-CE-42-AD.

(a) *What airplanes are affected by this AD?* This AD affects Beech Model 58 airplanes; serial numbers TH-1389, and TH-1396 through TH-1885, that are certificated in any category.

(b) *Who must comply with this AD?* Anyone who wishes to operate any of the above airplanes must comply with this AD.

(c) *What problem does this AD address?* The actions specified by this AD are intended to correct the wrong use of screws and consequent wear in the pilot/copilot pedal interconnect tube, which could result in loss of rudder control.

(d) *What actions must I accomplish to address this problem?* To address this problem, you must do the following actions:

Actions	Compliance times	Procedures
(1) Inspect the rudder bellcrank interconnect tube for damage and ensure the floorboard panel screws are 3/4 inch or less in length. Screws that are longer than 3/4 inch in length can damage parts installed immediately below the floorboards.	Inspect within the next 6 calendar months after December 29, 2000 (the effective date of this AD).	Do this inspection in accordance with the ACCOMPLISHMENT INSTRUCTIONS paragraph of Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000, and the Baron Model 58 Shop Manual.
(2) If you find no damage to the rudder bellcrank interconnect tube, discard any self-tapping coarse thread screw installed in the flanges that is longer than 3/4 inch.	Before further flight after the inspection.	Do these actions in accordance with the ACCOMPLISHMENT INSTRUCTIONS paragraph of Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000, and the Baron 58 Shop Manual.
(3) If you find damage to the rudder bellcrank interconnect tube, and the damage has not worn into the aluminum interconnect tube, refinish the interconnect tube and discard any self-tapping coarse thread screw installed in the flanges that is longer than 3/4 inch.	Before further flight after the inspection.	Do these actions in accordance with the ACCOMPLISHMENT INSTRUCTIONS paragraph of Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000, and the Baron 58 Shop Manual.
(4) If you find damage to the rudder bellcrank interconnect tube, and the damage has worn into the aluminum interconnect tube, you must replace the interconnect tube and discard any self-tapping coarse thread screw installed in the flanges that is longer than 3/4 inch.	Before further flight after the inspection.	Do these actions in accordance with the ACCOMPLISHMENT INSTRUCTIONS paragraph of Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000, and the Baron 58 Shop Manual.
(5) Plug the floorboard screw hole	Before further flight after the inspection.	Do these actions in accordance with the ACCOMPLISHMENT INSTRUCTIONS paragraph of Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000, and the Baron 58 Shop Manual.

(e) *Can I comply with this AD in any other way?* You may use an alternative method of compliance or adjust the compliance time if:

- (1) Your alternative method of compliance provides an equivalent level of safety; and
- (2) The Manager, Wichita Aircraft Certification Office (ACO), approves your alternative. Send your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. You should include in the request an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

(f) *Where can I get information about any already-approved alternative methods of compliance?* Contact Paul C. DeVore, Aerospace Engineer, FAA, Wichita Aircraft

Certification Office, 1801 Airport Road, Mid-Continent Airport, Wichita, Kansas 67209; telephone: (316) 946-4142; facsimile: (316) 946-4407.

(g) *What if I need to fly the airplane to another location to comply with this AD?* The FAA can issue a special flight permit under sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can do the requirements of this AD.

(h) *Are any service bulletins incorporated into this AD by reference?* Actions required by this AD must be done in accordance with Raytheon Mandatory Service Bulletin SB 27-3013, Issued: June 2000. The Director of the Federal Register approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51. You can get copies from Raytheon Aircraft Company, P.O. Box 85, Wichita, Kansas 67201-0085; telephone: (800) 429-5372 or (316) 676-3140; or on the Internet at <<http://www.raytheon.com/rac/servinfo/27-3013.pdf>>. This file is in Adobe Portable Document Format. The Acrobat Reader is available at <<http://www.adobe.com/>>. You can look at copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800

North Capitol Street, NW, suite 700, Washington, DC.

(i) *When does this amendment become effective?* This amendment becomes effective on December 29, 2000.

Issued in Kansas City, Missouri, on October 30, 2000.

Marvin R. Nuss,
Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-28438 Filed 12-1-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-28-AD; Amendment 39-12016; AD 2000-24-09]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model MD-11 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.