

section, except those services covered by flat rate user fees elsewhere in this part, will be calculated at the hourly rate listed in the following table for each

employee required to perform the service. The person for whom the service is provided and the person requesting the service are jointly and

severally liable for payment of these user fees in accordance with §§ 130.50 and 130.51.

	User fee			
	Oct. 1, 2000– Sept. 30, 2001	Oct. 1, 2001– Sept. 30, 2002	Oct. 1, 2002– Sept. 30, 2003	Beginning Oct. 1, 2003
Hourly rate:				
Per hour	\$76.00	\$80.00	\$84.00	\$84.00
Per quarter hour	19.00	20.00	21.00	21.00
Per service minimum fee	23.00	24.00	24.00	25.00

- (1) Providing services to live animals for import or entry at airports, ocean ports, and rail ports.
- (2) Conducting inspections, including laboratory and facility inspections, required to obtain permits, either to import animal products, aquaculture products, organisms or vectors, or to maintain compliance with import permits.
- (3) Obtaining samples required to be tested, either to obtain import permits or to ensure compliance with import permits.
- (4) Providing services for imported birds or ratites that are not subject to quarantine.

- (5) Supervising the opening of in-bond shipments.
- (6) Providing services for in-bond or in-transit animals to exit the United States.
- (7) Inspecting an export isolation facility and the animals in it.
- (8) Supervising animal or bird rest periods prior to export.
- (9) Supervising loading and unloading of animals or birds for export shipment.
- (10) Inspecting means of conveyance used to export animals or birds.
- (11) Conducting inspections under part 156 of this chapter.

- (12) Inspecting and approving an artificial insemination center or a semen collection center or the animals in it.
 - (13) Providing other import-or export-related veterinary services for which there is no flat rate user fee specified elsewhere in this part.
- (b) *When do I pay an additional amount for employee(s) working overtime?* You must pay an additional amount if you need an APHIS employee to work on a Sunday, on a holiday, or at any time outside the normal tour of duty of that employee. Instead of paying the hourly rate user fee, you pay the rate listed in the following table for each employee needed to get the work done.

Overtime rates (outside the employee's normal tour of duty)	Premium rate user fee			
	Oct. 1, 2000– Sept. 30, 2001	Oct. 1, 2001– Sept. 30, 2002	Oct. 1, 2002– Sept. 30, 2003	Beginning Oct. 1, 2003
Premium hourly rate Monday through Saturday and holidays:				
Per hour	\$88.00	\$92.00	\$96.00	\$100.00
Per quarter hour	22.00	23.00	24.00	25.00
Premium hourly rate for Sundays:				
Per hour	104.00	104.00	108.00	112.00
Per quarter hour	26.00	26.00	27.00	28.00

(Approved by the Office of Management and Budget under control numbers 0579-0055 and 0579-0094)

14. Section 130.50 is amended as follows:
- a. By revising the paragraph (b)(3)(ii).
 - b. In paragraph (c)(2), by removing the reference to “§ 130.21” and adding in its place a reference to “§ 130.30”.
 - c. In paragraph (c)(5), by removing the reference to “§ 130.9” and adding in its place a reference to “§ 130.30”.

§ 130.50 Payment of user fees.

* * * * *

- (b) * * *
- (3) * * *

(ii) *What amount do I pay if I receive an hourly rate user fee service?* Instead of paying the normal hourly rate user fee under § 130.30(a), you pay the premium rate listed in § 130.30(b) for

each employee needed to get the work done.

* * * * *

Done in Washington, DC, this 22nd day of August 2000.

Bobby R. Acord,
Acting Administrator, Animal and Plant Health Inspection Service.
 [FR Doc. 00-21898 Filed 8-25-00; 8:45 am]
BILLING CODE 3410-34-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-SW-66-AD; Amendment 39-11882; AD 2000-17-08]

RIN 2120-AA64

Airworthiness Directives; Eurocopter Deutschland GMBH Model BO-105A, BO-105C, BO-105 C-2, BO-105 CB-2, BO-105 CB-4, BO-105S, BO-105 CS-2, BO-105 CBS-2, BO-105 CBS-4, and BO-105LS A-1 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD) that applies to Eurocopter Deutschland

GMBH (ECD) Model BO-105A, BO-105C, BO-105 C-2, BO-105 CB-2, BO-105 CB-4, BO-105S, BO-105 CS-2, BO-105 CBS-2, BO-105 CBS-4, and BO-105LS A-1 helicopters. That AD requires creating a component log card or equivalent record and determining the calendar age and number of flights on each tension-torsion (TT) strap. That AD also requires inspecting and removing, as necessary, certain unairworthy TT straps. This amendment establishes a life limit for certain main rotor TT straps. This amendment is prompted by a need to establish a life limit for certain TT straps because of an accident in which a main rotor blade (blade) separated from an ECD Model MBB-BK 117 helicopter due to fatigue failure of a TT strap. The same part-numbered TT strap is used on the ECD Model BO-105 helicopters. The actions specified by this AD are intended to prevent fatigue failure of the TT strap, loss of a blade, and subsequent loss of control of the helicopter.

EFFECTIVE DATE: October 2, 2000.

FOR FURTHER INFORMATION CONTACT: Charles Harrison, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, Fort Worth, Texas 76193-0110, telephone (817) 222-5128, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 99-19-22, which applies to ECD Model BO-105A, BO-105C, BO-105 C-2, BO-105 CB-2, BO-105 CB-4, BO-105S, BO-105 CS-2, BO-105 CBS-2, BO-105 CBS-4, and BO-105LS A-1 helicopters, was published in the **Federal Register** on April 24, 2000 (65 FR 21673). That action proposed to require establishing a life limit for the TT straps of 120 months or 40,000 flights, whichever occurs first.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that 200 helicopters of U.S. registry will be affected by this AD, that it will take approximately 16 work hours per helicopter to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$10,400 per helicopter. Based on these figures, the total cost impact of this AD on U.S. operators is estimated to be \$2,272,200.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-11357 (64 FR 54770, October 8, 1999), and by adding a new airworthiness directive (AD), Amendment 39-11882, to read as follows:

2000-17-08 Eurocopter Deutschland

GMBH: Amendment 39-11882. Docket No. 99-SW-66-AD. Supersedes AD 99-19-22, Amendment 39-11357, Docket No. 99-SW-52-AD.

Applicability: Model BO-105A, BO-105C, BO-105 C-2, BO-105 CB-2, BO-105 CB-4, BO-105S, BO-105 CS-2, BO-105 CBS-2, BO-105 CBS-4, and BO-105LS A-1 helicopters, with part number (P/N) 2604067 (Bendix) or J17322-1 (Lord) rotor tension-torsion (TT) strap, installed, certificated in any category.

Note 1: This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent fatigue failure of a TT strap, loss of a main rotor blade (blade), and subsequent loss of control of the helicopter, accomplish the following:

- (a) Before further flight,
 - (1) Create a component log card or equivalent record for each TT strap.
 - (2) Review the history of the helicopter and each TT strap. Determine the age since initial installation on any helicopter (age) and the number of flights on each TT strap. Enter both the age and the number of flights for each TT strap on the component log card or equivalent record. When the number of flights is unknown, multiply the number of hours time-in-service (TIS) by 5 to determine the number of flights. If a TT strap has been previously used at any time on Model BO-105LS A-3 "SUPER LIFTER", BO-105 CB-5, BO-105 CBS-5, BO-105 DBS-5, or any MBB-BK 117 series helicopter, multiply the number of flights accumulated on those other models by a factor of 1.6 and then add that result to the number of flights accumulated on the helicopters affected by this AD.
 - (3) Remove any TT strap from service if the total hours TIS or number of flights and age cannot be determined.

(b) On or before January 1, 2001, remove any TT strap that has been in service 120 months since initial installation on any helicopter or accumulated 40,000 flights (a flight is a takeoff and a landing), on any helicopter. Replace the TT strap with an airworthy TT strap.

(c) This AD revises the Airworthiness Limitations Section of the maintenance manual by establishing a life limit for the TT strap, P/N 2604067 and J17322-1, of 120 months or 40,000 flights, whichever occurs first.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR

21.197 and 21.199) to operate the helicopter to a location where the requirements of this AD can be accomplished.

(f) This amendment becomes effective on October 2, 2000.

Note 3: The subject of this AD is addressed in the Luftfahrt Bundesamt (Federal Republic of Germany) AD 1999-300/3, dated August 31, 1999.

Issued in Fort Worth, Texas, on August 21, 2000.

Eric Bries,

*Acting Manager, Rotorcraft Directorate,
Aircraft Certification Service.*

[FR Doc. 00-21871 Filed 8-25-00; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-SW-33-AD; Amendment 39-11881; AD 2000-17-07]

RIN 2120-AA64

Airworthiness Directives; Eurocopter France Model EC120B Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for Eurocopter France Model EC120B helicopters. This action requires adjusting the clearance of the cabin sliding door if necessary. This amendment is prompted by an in-flight loss of a cabin sliding door, which had been locked in the fully opened position. The actions specified in this AD are intended to prevent in-flight loss of a cabin sliding door, impact with the horizontal stabilizer, main rotor, or fenestron tail rotor, and subsequent loss of control of the helicopter.

DATES: Effective September 12, 2000.

Comments for inclusion in the Rules Docket must be received on or before October 27, 2000.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Office of the Regional Counsel, Southwest Region, Attention: Rules Docket No. 2000-SW-33-AD, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. You may also send comments electronically to the Rules Docket at the following address: 9-asw-adcomments@faa.gov. Comments may be inspected at the Office of the Regional Counsel between 9 a.m. and 3 p.m., Monday through Friday, except Federal holidays.

FOR FURTHER INFORMATION CONTACT:

Richard Monschke, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, Fort Worth, Texas 76193-0110, telephone (817) 222-5116, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: The Direction Generale De L'Aviation Civile (DGAC), the airworthiness authority for France, notified the FAA that an unsafe condition may exist on Eurocopter France Model EC120B helicopters. The DGAC advises that the cabin sliding door must be adjusted, if necessary, to prevent in-flight loss of the cabin sliding door.

Eurocopter France has issued Service Telex No. 05-005, dated June 30, 2000, which specifies adjusting any cabin sliding door if a roller is not completely inside its rail with a minimum clearance of 3 mm. Eurocopter France received a report of an in-flight loss of the cabin sliding door. An investigation shows that the loss of the door was due to the forward upper roller being out of its guide rail. The door edge thus exposed to the slipstream caused the forward lower roller train to be driven out of the guide rail due to the aerodynamic loads. The door aft hinges failed, and the door departed from the aircraft. The DGAC classified this service telex as mandatory and issued AD T2000-285-005(A), dated June 30, 2000, to ensure the continued airworthiness of these helicopters in France.

This helicopter model is manufactured in France and is type certificated for operation in the United States under the provisions of 14 CFR 21.29 and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

We have identified an unsafe condition that is likely to exist or develop on other Eurocopter France Model EC120B helicopters of the same type design registered in the United States. This AD is being issued to prevent in-flight loss of a cabin sliding door, impact with the horizontal stabilizer, main rotor, or fenestron tail rotor, and subsequent loss of control of the helicopter. This AD requires adjusting the clearance of any cabin sliding door to a minimum of 3 mm from the aft end of the rail. The short compliance time involved is required because the previously described

critical unsafe condition can adversely affect the structural integrity and controllability of the helicopter. Therefore, adjusting the clearance of the cabin sliding door to a minimum of 3 mm from the aft end of the rail is required before further flight with the door in the open position and this AD must be issued immediately.

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

The FAA estimates that 24 helicopters will be affected by this AD, that it will take approximately 0.25 work hours to adjust the cabin sliding door, and that the average labor rate is \$60 per work hour. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$360.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their mailed comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made:

"Comments to Docket No. 2000-SW-