

(including balance sheets and profit and loss statements). Based on this information, SBA will assess your continued need for disbursements under this program. Upon making such assessment, SBA will notify you of the status of future disbursements.

#### § 123.512 What is the interest rate on a Military Reservist EIDL?

The interest rate on a Military Reservist EIDL will be 4 percent per annum or less. SBA will publish the interest rate quarterly in the **Federal Register**.

Dated: June 30, 2000.

**Aida Alvarez,**  
Administrator.

[FR Doc. 00-17560 Filed 7-12-00; 8:45 am]

BILLING CODE 8025-01-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Chapter 1

[Docket No. FAA-2000-7623]

#### Review of Existing Regulations

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Review of regulations; request for comments.

**SUMMARY:** This notice invites you, as a member of the public, to tell us, the FAA, which regulations now in effect you believe we should amend, eliminate, or simplify. We are publishing this notice in response to Presidential Executive Order No. 12866, directing certain Federal agencies to periodically review their regulations. We need to ensure that they are consistent with statutory authority and are in the public interest. Your comments will assist us in conducting this review and in determining what actions we should take, if any.

**DATES:** Comments should be submitted on or before October 11, 2000.

**ADDRESSES:** Comments should be mailed or delivered in duplicate to: U.S. Department of Transportation Dockets, Docket No. [FAA-2000-7623], 400 Seventh Street, SW., Room Plaza 401, Washington, DC 20590. Comments may be filed and examined in Room Plaza 401 between 10 a.m. and 5 p.m. weekdays, except Federal holidays. Comments also may be sent electronically to the Dockets Management System (DMS) at the following Internet address: <http://dms.dot.gov>. Commenters who wish to file comments electronically should

follow the instructions on the DMS web site.

**FOR FURTHER INFORMATION CONTACT:** Gerri Robinson, ARM-24, Office of Rulemaking, Federal Aviation Administration, 800 Independence Ave. SW., Washington, DC 20591; telephone (202) 267-9678, facsimile (202) 267-5075.

**SUPPLEMENTARY INFORMATION:** In recent years, the FAA conducted several regulatory reviews. In his 1992 State of the Union address, then-President Bush called for a 90-day moratorium on and review of Federal regulations. We responded by asking for public comments on our regulatory program as part of that overall government review (57 FR 4744, Feb. 7, 1992). Based on comments we received, we revised our regulatory agenda.

In 1994, we did another public review (59 FR 1362, Jan. 10, 1994) responding to recommendations from the National Commission to Ensure a Strong Competitive Airline Industry. We were also responding to Vice President Gore's National Performance Review and acting on Department of Transportation (DOT) and FAA regulatory initiatives. We initiated that review of our regulations to reduce any unjustified burdens and as a result of that review we also revised our regulatory agenda and our priorities. At the same time, we announced a Regulatory Review Program to seek public input every three years (60 FR 44142, Aug. 24, 1995). After each review, we published a disposition of the comments.

The most recent review in the 3-year review cycle was announced in the **Federal Register** on May 15, 1997 (62 FR 26894, May 15, 1997). As a result of the Review of Existing Rules, the FAA identified several issues that it determined would be addressed in future rulemaking projects and concluded the review with a general disposition of comments on October 22, 1998 (63 FR 56539, Oct. 22, 1998).

#### Three-Year Regulatory Review Program; Request for Comments

As part of this ongoing Regulatory Review Program, you may submit a total of three regulations, in priority order, that you believe should be amended, revised, or eliminated. Our agency's goal is to identify regulations which impose unjustified regulatory burdens or are no longer necessary. We also want to identify regulations that need to be clarified or simplified, or overlap, duplicate, or conflict with other regulations. Also, please identify any regulations that have a significant economic burden on a substantial

number of small entities that you consider no longer justified.

To focus on areas of greatest interest, and to effectively manage FAA resources, we ask that you limit your comments to the issues you consider most urgent, and list them in priority order. We will review the issues addressed by all the commenters in light of our current regulatory agenda (64 FR 64682, November 22, 1999). We will consider your comments and adjust our regulatory priorities consistent with our statutory responsibilities. When we are done reviewing all comments, we will publish a summary and an explanation of how we will act on them, telling you how we will adjust our priorities.

Finally, please give us any specific suggestions where the regulations could be redone to be performance-based rather than prescriptive and submit your suggested language.

Issued in Washington DC, on July 7, 2000.

**Thomas E. McSweeney,**  
Associate Administrator for Regulation and Certification.

[FR Doc. 00-17790 Filed 7-12-00; 8:45 am]

BILLING CODE 4910-13-M

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 99-NM-243-AD]

RIN 2120-AA64

#### Airworthiness Directives; McDonnell Douglas Model MD-11 and MD-11F Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes the superseding of an existing airworthiness directive (AD), applicable to certain McDonnell Douglas Model MD-11 and MD-11F series airplanes, that currently requires opening the circuit breaker of the pneumatic sense line heater tape, installing an inoperative ring, and coiling and stowing the electrical wire to the circuit breaker of the pneumatic sense line heater tape. That AD also provides for an optional inspection, which, if accomplished, constitutes terminating action for deactivation of the pneumatic sense line heater tape. This proposal is prompted by the FAA's determination that the one-time optional terminating inspection in the existing AD does not adequately detect chafing, electrical arcing, or inadequate

clearance of the subject area. The actions specified by the proposed AD are intended to detect and correct such inadequate clearance, which could result in a hole in the fuel feed pipe caused by electrical arcing, and consequent fuel leakage and possible ignition of the fuel vapors. This action would require repetitive inspections of the subject area and corrective actions, if necessary, and would provide for an optional terminating modification(s) for the repetitive inspection requirements.

**DATES:** Comments must be received by August 28, 2000.

**ADDRESSES:** Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-243-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays. Comments may be submitted via fax to (425) 227-1232. Comments may also be sent via the Internet using the following address: 9-anm-nprmcomment@faa.gov. Comments sent via fax or the Internet must contain "Docket No. 99-NM-243-AD" in the subject line and need not be submitted in triplicate. Comments sent via the Internet as attached electronic files must be formatted in Microsoft Word 97 for Windows or ASCII text.

The service information referenced in the proposed rule may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Technical Publications Business Administration, Dept. C1-L51 (2-60). This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the the FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

**FOR FURTHER INFORMATION CONTACT:** Stephen Kolb, Senior Aerospace Engineer, Propulsion Branch, ANM-140L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712; telephone (562) 627-5244; fax (562) 627-5210.

**SUPPLEMENTARY INFORMATION:**

**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall

identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Submit comments using the following format:

- Organize comments issue-by-issue. For example, discuss a request to change the compliance time and a request to change the service bulletin reference as two separate issues.
- For each issue, state what specific change to the proposed AD is being requested.
- Include justification (*e.g.*, reasons or data) for each request.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 99-NM-243-AD." The postcard will be date stamped and returned to the commenter.

**Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-243-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

**Discussion**

On April 16, 1998, the FAA issued AD 98-08-11, amendment 39-10491 (63 FR 20066, April 23, 1998), applicable to certain McDonnell Douglas Model MD-11 and MD-11F series airplanes, to require opening the circuit breaker of the pneumatic sense line heater tape, installing an inoperative ring, and coiling and stowing the electrical wire to the circuit breaker of the pneumatic sense line heater tape. That AD also provides for an optional inspection, which, if accomplished, constitutes terminating action for deactivation of the pneumatic sense line heater tape. That action was prompted by a report indicating that, while an airplane was

on the ground, fuel was found leaking from the fuel feed pipe of the number 2 engine due to inadequate clearance between the fuel feed pipe and the pneumatic sense line heater tape. The requirements of that AD are intended to detect and correct such inadequate clearance, which could result in a hole in the fuel feed pipe caused by electrical arcing, and consequent fuel leakage and possible ignition of the fuel vapors.

**Actions Since Issuance of Previous Rule**

Since the issuance of AD 98-08-11, the FAA has determined that the optional one-time inspection provided by that AD does not ensure adequate clearance between the heater tape of the pneumatic sense lines and fuel feed pipe of the number 2 engine, which could result in a hole in the fuel feed pipe caused by electrical arcing, and consequent fuel leakage and possible ignition of the fuel vapors. Because the pneumatic sense lines can move and cause the heater tape to contact the fuel feed pipe of the number 2 engine, the FAA finds that repetitive detailed visual inspections of the subject area are necessary in order to address the identified unsafe condition of this AD.

**Explanation of Relevant Service Information**

The FAA has reviewed and approved McDonnell Douglas Alert Service Bulletin MD11-36A030, Revision 03, dated December 14, 1999. The alert service bulletin describes procedures for opening the circuit breaker of the pneumatic sense line heater tape, installing an inoperative ring, and coiling and stowing the electrical wire to the circuit breaker of the pneumatic sense line heater tape. Accomplishment of the above actions deactivates the pneumatic sense line heater tape. The alert service bulletin also describes procedures for repetitive detailed visual inspections to detect chafing, electrical arcing, or inadequate clearance of the heater tape of the pneumatic sense lines and fuel feed pipe of the number 2 engine; and corrective actions, if necessary. The corrective actions involve repositioning the pneumatic sense lines, rewinding the insulation on the pneumatic sense lines, and repairing or replacing damaged parts with new parts. Accomplishment of the repetitive inspections eliminates the need for deactivation of the pneumatic sense line heater tape.

The FAA also has reviewed and approved the following optional service bulletins. Accomplishment of the applicable actions specified in these service bulletins eliminates the need for

the repetitive inspections described above.

- McDonnell Douglas Service Bulletin MD11-36-018 R01, Revision 1, dated July 18, 1995, describes, for certain airplanes, procedures for modification of the high stage pilot valve located in the aft accessory compartment (including purging the sense lines and revising wiring of the high stage pilot valve).

- McDonnell Douglas Service Bulletin MD11-36-026, dated September 30, 1996, describes, for certain airplanes, procedures for disconnecting and splicing together the heater tape wires of the pneumatic sense lines for the high stage and fan air valves from the terminal strips in the lower vertical stabilizer.

- McDonnell Douglas Service Bulletin MD11-36-025 R01, Revision 01, dated July 31, 1997, describes, for certain airplanes, modification and reidentification of the pilot pressure regulator valve located in the aft accessory compartment (including purging the sense lines and revising the wiring of the pilot pressure regulator valve).

- McDonnell Douglas Service Bulletin MD11-36-028, dated December 7, 1998, describes, for certain airplanes, procedures for disconnecting the heater tape wires from their respective terminal strips and splicing the wire ends together.

#### **Explanation of Requirements of Proposed Rule**

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would supersede AD 98-08-11 to require accomplishment of the actions specified in McDonnell Douglas Alert Service Bulletin MD11-36A030, Revision 03, dated December 14, 1999, described previously. The proposed AD also would provide for an optional terminating modification for the repetitive inspection requirements. The proposed AD also would require that operators report results of inspection findings to the FAA.

The FAA is not proposing to mandate the modification specified in paragraph (d)(1), (d)(2), (d)(3), or (d)(4) of this AD for several reasons:

1. Accessing the pneumatic sense lines and fuel feed pipe of the number 2 engine for inspection is easily accomplished.
2. The chafing, electrical arcing, or inadequate clearance of the subject area is easily detectable.
3. The repetitive detailed visual inspections will minimize the

probability of a hole in the fuel feed pipe being caused by electrical arcing, which may result in fuel leakage and possible ignition of the fuel vapor.

#### **Differences Between the Proposed AD and Relevant Service Information**

Operators should note that, although McDonnell Douglas Alert Service Bulletin MD11-26A030 recommends accomplishing the repetitive detailed visual inspections at intervals not to exceed 5,000 flight hours, the FAA has determined that an interval of 5,000 flight hours or 18 months, whichever occurs later, would address the identified unsafe condition in a timely manner. In developing an appropriate compliance time for this AD, the FAA considered not only the manufacturer's recommendation, but the degree of urgency associated with addressing the subject unsafe condition, the high utilization of some operator's affected fleet, and the time necessary to perform the inspection (one hour). In light of all of these factors, the FAA finds an interval of 5,000 flight hours or 18 months, whichever occurs later, for the repetitive detailed visual inspections to be warranted, in that it represents an appropriate interval of time allowable for affected airplanes to continue to operate without compromising safety.

In addition to the procedures described above, McDonnell Douglas Service Bulletin MD11-36-018 R01, Revision 1, describes procedures for modification of the high stage pilot valve of the left and right wings, and McDonnell Douglas Service Bulletin MD11-36-025 R01, Revision 01, describes procedures for modification and reidentification of the pilot pressure regulator valve of the left and right wings. Accomplishment of these modifications is not necessary to comply with certain optional actions provided by this AD. These particular modifications do not address the identified unsafe condition of this AD.

#### **Cost Impact**

There are approximately 174 Model MD-11 and MD-11F series airplanes of the affected design in the worldwide fleet. The FAA estimates that 67 airplanes of U.S. registry would be affected by this proposed AD.

The modification that is currently required by AD 98-08-11, and retained in this proposed AD, takes approximately 1 work hour per airplane to accomplish, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the currently required actions on U.S. operators is estimated to be \$4,020, or \$60 per airplane.

The new inspection that is proposed in this AD action would take approximately 1 work hour per airplane to accomplish, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the new inspection proposed by this AD on U.S. operators is estimated to be \$4,020, or \$60 per airplane, per inspection cycle.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the current or proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Should an operator elect to accomplish the optional terminating action that would be provided by paragraph (d)(1) of this proposed AD, it would take approximately 4 work hours to accomplish it, at an average labor rate of \$60 per work hour. The cost of required parts would be approximately \$4,500 per airplane. Based on these figures, the cost impact of this optional terminating action would be \$4,740 per airplane.

Should an operator elect to accomplish the optional terminating action that would be provided by paragraph (d)(2) of this proposed AD, it would take approximately 1 work hour to accomplish it, at an average labor rate of \$60 per work hour. The cost of required parts would be approximately \$50 per airplane. Based on these figures, the cost impact of this optional terminating action would be \$110 per airplane.

Should an operator elect to accomplish the optional terminating action that would be provided by paragraph (d)(3) of this proposed AD, it would take approximately 2 work hours to accomplish it, at an average labor rate of \$60 per work hour. The cost of required parts would be approximately \$2,500 per airplane. Based on these figures, the cost impact of this optional terminating action would be \$2,620 per airplane.

Should an operator elect to accomplish the optional terminating action that would be provided by paragraph (d)(4) of this proposed AD, it would take approximately 4 work hours to accomplish it, at an average labor rate of \$60 per work hour. The cost of required parts would be approximately \$50 per airplane. Based on these figures, the cost impact of this optional terminating action would be \$290 per airplane.

#### **Regulatory Impact**

The regulations proposed herein would not have a substantial direct effect on the States, on the relationship

between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this proposal would not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39-10491 (63 FR 20066, April 23, 1998), and by adding a new airworthiness directive (AD), to read as follows:

**McDonnell Douglas:** Docket 99-NM-243-AD. Supersedes AD 98-08-11, Amendment 39-10491.

**Applicability:** Model MD-11 and MD-11F series airplanes, having manufacturer's fuselage numbers 0447 through 0552 inclusive, and 0554 through 0620 inclusive; certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For

airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To detect and correct inadequate clearance between the fuel feed pipe of the number 2 engine and the pneumatic sense line heater tape, which could result in a hole in the fuel feed pipe caused by electrical arcing, and consequent fuel leakage and possible ignition of the fuel vapors, accomplish the following:

#### Restatement of Requirements of AD 98-08-11

##### Modification

(a) Within 7 days after April 28, 1998 (the effective date of AD 98-08-11, amendment 39-10491), open the circuit breaker of the pneumatic sense line heater tape, install an inoperative ring, and coil and stow the electrical wire to the circuit breaker of the pneumatic sense line heater tape, in accordance with Phase 1 of the Accomplishment Instructions of McDonnell Douglas Alert Service Bulletin MD11-36A030, dated April 2, 1998; Revision 01, dated September 28, 1998; Revision 02, dated July 27, 1999; or Revision 03, dated December 14, 1999. Accomplishment of these actions deactivates the pneumatic sense line heater tape.

**Note 2:** The pneumatic sense line heater tape of the number 2 engine has been deactivated. This deactivation may cause a nuisance shutdown of the bleed air system of the number 2 engine at top of descent.

#### New Requirements of This AD

##### Repetitive Inspections

(b) Except as provided in paragraph (d) of this AD, within 6 months after the effective date of this AD, perform a detailed visual inspection to detect chafing, electrical arcing, or inadequate clearance of the pneumatic sense lines and fuel feed pipe of the number 2 engine, in accordance with Phase 2 of the Accomplishment Instructions of McDonnell Douglas Alert Service Bulletin MD11-36A030, Revision 03, dated December 14, 1999. Repeat the inspection thereafter at intervals not to exceed 5,000 flight hours or 18 months, whichever occurs later. Accomplishment of the detailed visual inspection constitutes terminating action for the deactivation requirements of paragraph (a) of this AD.

**Note 3:** For the purposes of this AD, a detailed visual inspection is defined as: "An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc. may be used. Surface cleaning and elaborate access procedures may be required."

**Note 4:** Detailed visual inspections accomplished before the effective date of this AD in accordance with McDonnell Douglas Alert Service Bulletin MD11-36A030, dated April 2, 1998, Revision 01, dated September 28, 1998, or Revision 02, dated July 27, 1999; are considered acceptable for compliance with the requirements of paragraph (b) of this AD.

#### Corrective Actions

(c) If any discrepancy (*i.e.*, as identified in Conditions 1, 2, 3, 4, and 5 of the Accomplishment Instructions of McDonnell Douglas Alert Service Bulletin MD11-36A030, Revision 03, dated December 14, 1999) is detected during any inspection required by paragraph (b) of this AD, before further flight, perform the applicable corrective actions in accordance with Conditions 1, 2, 3, 4, or 5 of the Accomplishment Instructions of McDonnell Douglas Alert Service Bulletin MD11-36A030, Revision 03, dated December 14, 1999, except as indicated in paragraphs (c)(1) and (c)(2) of this AD.

(1) Accomplishment of the modification of the high stage pilot valve of the left and right wings in accordance with McDonnell Douglas Service Bulletin MD11-36-018 R01, Revision 1, dated July 18, 1995, is NOT necessary to comply with the applicable corrective action in Condition 5 of the Accomplishment Instructions of the service bulletin.

(2) Accomplishment of the modification and reidentification of the pilot pressure regulator valve of the left and right wings in accordance with McDonnell Douglas Service Bulletin MD11-36-025 R01, Revision 01, dated July 31, 1997, is NOT necessary to comply with the applicable corrective action in Condition 5 of the Accomplishment Instructions of the service bulletin.

#### Optional Actions

(d) Accomplishment of the action(s) specified in paragraphs (d)(1), (d)(2), (d)(3), and (d)(4) of this AD, as applicable, constitutes terminating action for the repetitive inspection requirements of paragraph (b) of this AD.

(1) For airplanes having manufacturer's fuselage numbers 0447 through 0552 inclusive, and 0554

through 0573 inclusive: Before or in conjunction with the actions specified in paragraph (d)(2) of this AD, modify the high stage pilot valve located in the aft accessory compartment (including purging the sense lines and revising wiring of the high stage pilot valve), in accordance with McDonnell Douglas Service Bulletin MD11-36-018 R01, Revision 1, dated July 18, 1995.

**Note 5:** In addition to the procedures for modification of the high stage pilot valve located in the aft accessory compartment, McDonnell Douglas Service Bulletin MD11-36-018 R01, Revision 1, dated July 18, 1995, also describes procedures for modification of the high stage pilot valve of the left and right wings. Accomplishment of modification of the high stage pilot valve of the left and right wings is NOT necessary to comply with the optional action provided by paragraph (d)(1) of this AD.

**Note 6:** Modification of the high stage pilot valve of the aft accessory compartment accomplished before the effective date of this AD in accordance with McDonnell Douglas Service Bulletin MD11-36-018, dated March 28, 1995, is considered acceptable for compliance with the actions specified in paragraph (d)(1) of this AD.

(2) For airplanes having manufacturer's fuselage numbers 0447 through 0552 inclusive, and 0554 through 0608 inclusive: Disconnect and splice together the heater tape wires of the pneumatic sense lines for the high stage and fan air valves from the terminals strips in the lower vertical stabilizer, in accordance with McDonnell Douglas Service Bulletin MD11-36-026, dated September 30, 1996.

(3) For airplanes having manufacturer's fuselage numbers 0447 through 0552 inclusive, and 0554 through 0608 inclusive: Before or in conjunction with the actions specified in paragraph (d)(4) of this AD, modify and reidentify the pilot pressure regulator valve located in the aft accessory compartment (including purging the sense lines and revising the wiring of the pilot pressure regulator valve), in accordance with McDonnell Douglas Service Bulletin MD11-36-025 R01, Revision 01, dated July 31, 1997.

**Note 7:** In addition to the procedures for modification and reidentification of the pilot pressure regulator valve located in the aft accessory compartment, McDonnell Douglas Service Bulletin MD11-36-025 R01, Revision 01, dated July 31, 1997, also describes procedures for modification and reidentification of the pilot pressure regulator valve of the left and right wings. Accomplishment of the modification and reidentification of the pilot pressure regulator valve of the left and right wings is *not* necessary to comply with the optional action provided by paragraph (d)(3) of this AD.

**Note 8:** Modification and reidentification of the pilot pressure regulator valve of the aft accessory compartment accomplished before the effective date of this AD in accordance with McDonnell Douglas Service Bulletin MD11-36-025, dated February 14, 1997; is considered acceptable for compliance with the actions specified in paragraph (d)(3) of this AD.

(4) For airplanes having manufacturer's fuselage numbers 0447 through 0464 inclusive, 0466 through 0552 inclusive, and 0554 through 0620 inclusive: Disconnect the heater tape wires from their respective terminal strips and splice the wire ends together, in accordance with McDonnell Douglas Service Bulletin MD11-36-028, dated December 7, 1998.

#### Reporting

(e) Within 10 days after accomplishing any inspection required by paragraph (b) of this AD, submit a report of the inspection results (only negative findings) to the Manager, Los Angeles Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate, 3960 Paramount Boulevard, Lakewood, California 90712-4137; fax (562) 627-5210. Information collection requirements contained in this regulation have been approved by the Office of Management and Budget (OMB) under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 *et seq.*) and have been assigned OMB Control Number 2120-0056.

#### Alternative Methods of Compliance

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles ACO. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

**Note 9:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

#### Special Flight Permits

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on July 7, 2000.

**John J. Hickey,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

[FR Doc. 00-17758 Filed 7-12-00; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF HEALTH AND HUMAN SERVICES

### Food and Drug Administration

#### 21 CFR Parts 20, 58, 170, 171, 174, and 179

[Docket No. 99N-5556]

#### Food Additives: Food Contact Substance Notification System

**AGENCY:** Food and Drug Administration, HHS.

**ACTION:** Proposed rule.

**SUMMARY:** The Food and Drug Administration (FDA) is proposing to implement the premarket notification process for food contact substances (FCS's) established by the Food and Drug Administration Modernization Act (FDAMA) of 1997. Once implemented, the notification process will be the primary method for authorizing new uses of food additives that are FCS's. FDA is proposing regulations that identify the circumstances under which a food additive petition (FAP) will be required to authorize the use of an FCS; specify the information required in a notification for an FCS; describe the administration of the notification process; and establish the procedure by which the agency may deem a notification to no longer be effective. Additionally, FDA is announcing elsewhere in this issue of the **Federal Register** the availability of an administrative guidance document relating to the preparation of premarket notifications (PMN's).

**DATES:** Submit written comments by September 26, 2000, except that comments regarding information collection provisions should be submitted by August 14, 2000.

**ADDRESSES:** Submit written comments to the Dockets Management Branch (HFA-305), Food and Drug Administration, 5630 Fishers Lane, rm. 1061, Rockville, MD 20852. Submit written comments on the information collection requirements to the Office of Information and Regulatory Affairs, Office of Management and Budget (OMB), New Executive Office Bldg., 725 17th St. NW., rm. 10235, Washington, DC 20503, ATTN: Desk Officer for FDA.