

(ii) A State agency shall make adjustments to an account to correct an auditable, out-of-balance settlement condition that occurs during the redemption process as a result of a system error. A system error is defined as an error resulting from a malfunction at any point in the redemption process: from the system host computer, to the switch, to the third party processors, to a store's host computer or point of sale (POS) device. These adjustments may occur after the availability date and may result in either a debit or credit to the household.

(A) *Client-initiated adjustments.* The State agency must act on all requests for adjustments made by client households within 90 calendar days of the error transaction. The State agency has 10 business days from the date the household notifies it of the error to investigate and reach a decision on an adjustment and move funds into the client account. This timeframe also applies if the State agency or entity other than the household discovers a system error that requires a credit adjustment to the household. Business days are defined as calendar days other than Saturdays, Sundays, and Federal holidays.

(B) *Retailer-initiated adjustments.* The State agency must act upon all adjustments to debit a household's account no later than 10 business days from the date the error occurred, by placing a hold on the adjustment balance in the household's account. If there are insufficient benefits to cover the entire adjustment, a hold shall be placed on any remaining balance that exists, with the difference being subject to availability only in the next future month. The household shall be given, at a minimum, adequate notice in accordance with § 273.13 of this chapter. The notice must be sent at the time the initial hold is attempted on the household's current month's remaining balance, clearly state the full adjustment amount, and advise the household that any amount still owing is subject to collection from the household's next future month's benefits.

(1) The household shall have 90 days from the date of the notice to request a fair hearing.

(2) Should the household dispute the adjustment and request a hearing within 10 days of the notice, a provisional credit must be made to the household's account by releasing the hold on the adjustment balance within 48 hours of the request by the household, pending resolution of the fair hearing. If no request for a hearing is made within 10 days of the notice, the hold is released on the adjustment balance, and this

amount is credited to the retailer's account. If there are insufficient funds available in the current month to cover the full adjustment amount, the hold may be maintained and settled at one time after the next month's benefits become available.

(iii) The appropriate management controls and procedures for accessing benefit accounts after the posting shall be instituted to ensure that no unauthorized adjustments are made in accordance with paragraph (f)(7)(iii) of this section.

* * * * *

(10) * * *

(viii) Disclosure information regarding adjustments and a household's rights to notice, fair hearings, and provisional credits. The disclosure must also state where to call to dispute an adjustment and request a fair hearing. State agencies that have already implemented EBT shall have one year in which to grandfather adjustment disclosure into their training materials.

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Dated: June 28, 2000.

Samuel Chambers, Jr.,
Administrator, Food and Nutrition Service.
[FR Doc. 00-16944 Filed 7-3-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 98-SW-74-AD; Amendment 39-11807; AD 2000-13-08]

RIN 2120-AA64

Airworthiness Directives; Eurocopter Deutschland GmbH Model EC 135 Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD) that applies to Eurocopter Deutschland GmbH (ECD) Model EC 135 helicopters. That AD requires conducting a tail rotor drive shaft vibration survey (survey), installing a Fenestron Shaft Retrofit Kit, inspecting each tail rotor drive shaft bearing (bearing) attaching lock plate for bent-open tabs and broken or missing slippage marks, and visually inspecting each bearing support for cracks. This AD requires conducting the survey and installing the Fenestron Shaft Retrofit Kit. This AD also requires installing double bearing supports and struts,

revising the required compliance time for the repetitive inspections of the bearing attach hardware and supports, and removing the requirement to contact the FAA if a lock plate tab is bent open or if slippage marks are broken or missing. This amendment is prompted by continued reports of misaligned or cracked bearing supports and loose bearing attachment bolts. The actions specified by this AD are intended to prevent loss of drive to the tail rotor and subsequent loss of control of the helicopter.

DATES: Effective August 9, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 9, 2000.

ADDRESSES: The service information referenced in this AD may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Uday Garadi, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Rotorcraft Standards Staff, Fort Worth, Texas 76193-0110, telephone (817) 222-5123, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 98-15-25, Amendment 39-10866 (63 FR 59206, November 3, 1998), that applies to ECD Model EC 135 helicopters, was published in the **Federal Register** on April 6, 2000 (65 FR 18010). That action proposed to require the following:

- Conducting a vibration survey and installing the Fenestron Shaft Retrofit Kit L535M3002 882;
- Installing double bearing supports and struts;
- Inspecting and, if necessary, replacing bearing attach hardware; and
- Increasing the repetitive inspection interval for the bearing supports and attach hardware to 50 hours time-in-service (TIS).

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. The FAA has determined that air safety and the

public interest require the adoption of the rule as proposed.

The FAA estimates that 16 helicopters of U.S. registry will be affected by this AD. The 50-hour inspection will take approximately 2 work hours to complete. The average labor rate is \$60 per work hour. ECD has stated in its ASB's that the baseline vibration measurements and initial installation of all new parts will be provided at no charge to the owner/operator. Assuming the helicopters are operated 900 hours TIS per year, the total cost impact of this AD on U.S. operators is estimated to be \$34,560.

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules

Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-10866, and by adding a new airworthiness directive (AD), Amendment 39-11807, to read as follows:

2000-13-08 Eurocopter Deutschland

GmbH: Amendment 39-11807. Docket No. 98-SW-74-AD. Supersedes AD 98-15-25, Amendment 39-10866, Docket No. 98-SW-35-AD.

Applicability: Model EC 135 helicopters, certificated in any category.

Note 1: This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance

of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent loss of drive to the tail rotor and subsequent loss of control of the helicopter, accomplish the following:

(a) Before further flight, conduct a tail rotor drive shaft vibration survey and install a Fenestron Shaft Retrofit Kit L535M3002 882 in accordance with Eurocopter Deutschland GmbH Alert Service Bulletin (ASB) EC 135-53A-005, Revision 3, dated September 2, 1998.

(b) Before further flight, install double bearing supports for the tail rotor driveshaft and tail boom struts in accordance with ASB EC 135-53A-004, dated August 14, 1998.

Note 2: ASB EC 135-53A-002, Revision 2, dated September 2, 1998, pertains to the subject of this AD.

(c) Before further flight and thereafter at intervals not to exceed 50 hours time-in-service, perform the following:

(1) Clean each tail rotor drive shaft bearing support. Using a 6-power or higher magnifying glass and a bright light, visually inspect the attach lugs of the bearing supports B and C (shown in Figure 1) for cracks, particularly in the area extending from the bend radius to the attaching screws and rivets connecting the bearing supports to the tail boom. Before further flight, replace each cracked bearing support with an airworthy bearing support.

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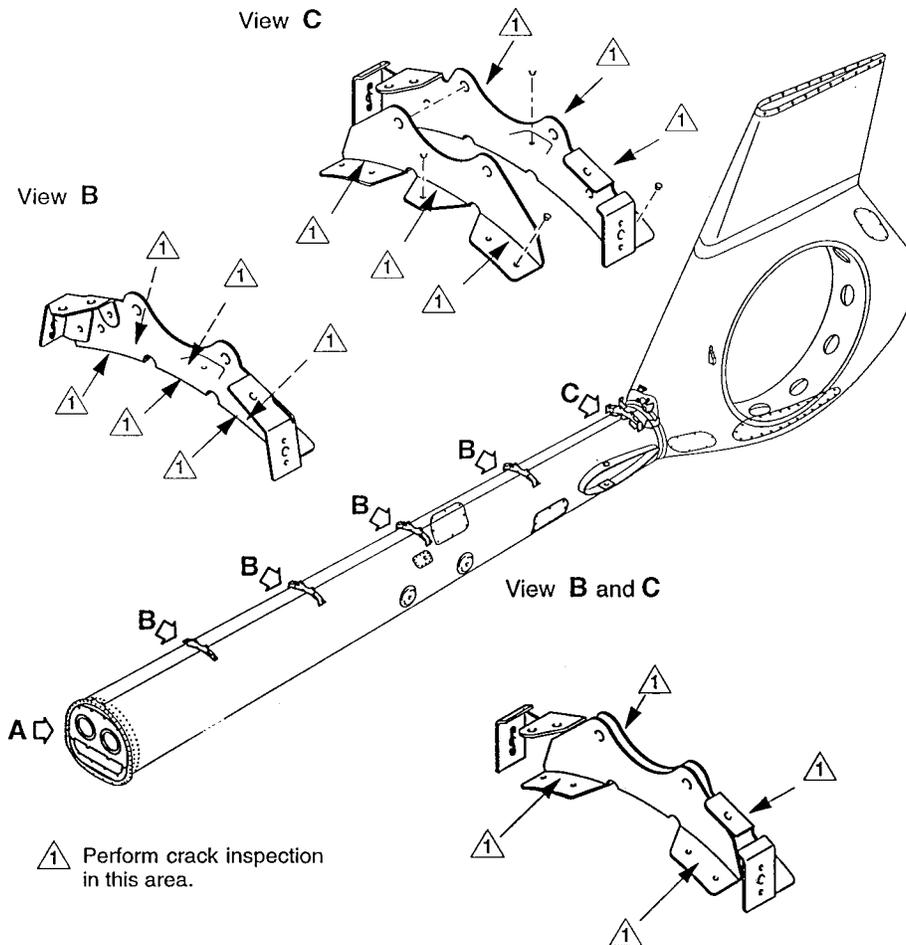


Figure 1

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(2) Inspect each bearing attach hardware lock plate for bent-open tabs and slippage marks for attach hardware looseness or rotation. Before further flight, replace any loose bearing attach hardware (including lock plates found bent or open due to bolt rotation) with airworthy hardware.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the helicopter to a location where the requirements of this AD can be accomplished.

(f) The modifications shall be done in accordance with Eurocopter Deutschland GmbH Alert Service Bulletin EC 135-53A-005, Revision 3, dated September 2, 1998, and Eurocopter Deutschland GmbH Alert Service Bulletin EC 135-53A-004, dated

August 14, 1998. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527. Copies may be inspected at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(g) This amendment becomes effective on August 9, 2000.

Note 4: The subject of this AD is addressed in Luftfahrt-Bundesamt (Federal Republic of Germany) AD's 1998-033/7 and 1998-389, both dated September 14, 1998.

Issued in Fort Worth, Texas, on June 26, 2000.

Eric Bries,

Acting Manager, Rotorcraft Directorate, Aircraft Certification Service.

[FR Doc. 00-16781 Filed 7-3-00; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 71

[Airspace Docket No. 00-AAL-1]

Revision of Class E Airspace; Barrow, AK

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This action revises Class E airspace at Barrow, AK. The revision of instrument approach procedures to runway (RWY) 6 and RWY 24 at Wiley Post—Will Rogers Memorial Airport, Barrow, AK, made this action necessary. This rule provides adequate controlled airspace for aircraft flying IFR procedures at Barrow, AK.

EFFECTIVE DATE: 0901 UTC, October 5, 2000.

FOR FURTHER INFORMATION CONTACT: Bob Durand, Operations Branch, Federal Aviation Administration, 222 West 7th Avenue, Box 14, Anchorage, AK 99513-