

responsibilities among the various levels of government. Therefore, in accordance with section 6 of E.O. 13132, the Immigration and Naturalization Service has determined that this rule does not have sufficient federalism implications to warrant the preparation of a federalism summary impact statement.

Executive Order 12988 Civil Justice Reform

This rule meets the applicable standards set forth in section 3(a) and 3(b)(2) of E.O. 12988.

Paperwork Reduction Act

This final rule does not impose any new reporting or recordkeeping requirements. The information collection requirements contained in this rule have previously been approved for use by the OMB under provisions of the Paperwork Reduction Act. The OMB control numbers for these collections are contained in 8 CFR 299.5, Display of control numbers.

List of Subjects in 8 CFR Part 245

Aliens, Immigration, Reporting and recordkeeping requirements.

Accordingly, the interim rule amending 8 CFR part 245 which was published at 62 FR 28314 on May 23, 1997, is adopted as a final rule with the following changes:

PART 245—ADJUSTMENT OF STATUS TO THAT OF PERSON ADMITTED FOR PERMANENT RESIDENCE

1. The authority citation for part 245 continues to read as follows:

Authority: 8 U.S.C. 1101, 1103, 1182, 1255; sec. 202, Pub. L. 105-100, 111 Stat. 2160, 2193; sec. 902, Pub. L. 105-277, 112 Stat. 2681; 8 CFR part 2.

2. Section 245.12 is revised to read as follows:

§ 245.12 What are the procedures for certain Polish and Hungarian parolees who are adjusting status to that of permanent resident under the Illegal Immigration Reform and Immigrant Responsibility Act of 1996?

(a) *How do I apply for adjustment of status under this section?* (1) Each person applying for adjustment of status, under section 646(b) of Public Law 104-208, must file a completed Form I-485, Application to Register Permanent Residence or Adjust Status, with the correct filing fee, with the Service director having jurisdiction over the applicant's place of residence.

(2) The application must include Form G-325A, Biographic Information and the results of the medical

examination made according to § 232.1 of this chapter and § 245.5.

(3) The application must include evidence to show the applicant was a national of Poland or Hungary who, after being denied refugee status, was inspected and granted parole into the United States between November 1, 1989, and December 31, 1991.

(4) The applicant must have been physically present in the United States for at least 1 year before filing a Form I-485.

(5) After receiving the Form I-485, the adjudicating Service office will notify each applicant who is 14 years old or older of the time and location for the required fingerprinting.

(b) *How is my application for adjustment of status affected if I leave the United States while my application is still pending?* The departure from the United States by an applicant for adjustment of status must be considered an abandonment of the application, as provided in § 245.2(a)(4)(ii), unless the applicant was previously granted advance parole for such absence, and was reinspected on returning to the United States.

(c) *Which grounds for inadmissibility do not apply or can be waived?* The provisions of section 212(a) (4), (5), and (7)(A) of the Act will not apply to adjustment of status under § 245.12. In addition, the director may waive any other ground of inadmissibility except section 212(a)(2)(C) or 212(a)(3)(A), (B), (C), or (E) of the Act, for humanitarian purposes, to ensure family unity, or when it is otherwise in the public interest.

(d) *If my application for adjustment of status is approved under § 245.12, what date will be recorded as my admission to permanent residence?* On approval of the application for adjustment of status, the date of the applicant's admission to permanent resident status will be the date of the applicant's inspection and parole, as described in paragraph (a) of this section.

Dated: March 28, 2000.

Doris Meissner,

Commissioner, Immigration and Naturalization Service.

[FR Doc. 00-9320 Filed 4-13-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-07-AD; Amendment 39-11685; AD 2000-07-29]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300, A310, and A300-600 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to certain Airbus Model A300, A310, and A300-600 series airplanes, that currently requires a one-time operational test of the fire shut-off valves (FSOV) to determine if the FSOV's are functioning correctly, and replacement of failed parts with new or serviceable parts. This amendment requires repetitive performance of the operational test. This amendment also limits the applicability to airplanes installed with certain FSOV's. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to detect and correct failure of the FSOV's to close, which could result in failure of the engine fire shut-off system, and consequent inability to extinguish an engine fire.

DATES: Effective May 19, 2000.

The incorporation by reference of certain publications, as listed in the regulations, was approved previously by the Director of the Federal Register as of September 4, 1998 (63 FR 40811, July 31, 1998).

ADDRESSES: The service information referenced in this AD may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 98-16-09, amendment 39-10685 (63 FR 40811, July 31, 1998), which is applicable to certain Airbus Model A300, A310, and A300-600 series airplanes, was published in the Federal Register on February 10, 2000 (65 FR 6563). The action proposed to continue to require a one-time operational test of the fire shut-off valves (FSOV) to determine if the FSOV's are functioning correctly, and replacement of failed parts with new or serviceable parts. The action also proposed to require repetitive performance of the operational test. In addition, the action also proposed to limit the applicability to airplanes installed with certain FSOV's.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

One commenter states no objection to the proposed rule. Another commenter states that it has no comment to the proposed rule.

Conclusion

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

There are approximately 103 airplanes of U.S. registry that will be affected by this AD.

The operational test that is currently required by AD 98-16-09, and retained in this AD, takes approximately 1 work hour per airplane to accomplish, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the currently required test on U.S. operators is estimated to be \$60 per airplane, per test cycle.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is

determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39-10685 (63 FR 40811, July 31, 1998), and by adding a new airworthiness directive (AD), amendment 39-11685, to read as follows:

2000-07-29 Airbus Industrie: Amendment 39-11685. Docket 99-NM-07-AD. Supersedes AD 98-16-09, Amendment 39-10685.

Applicability: Model A300, A310, and A300-600 series airplanes; on which any fire shut-off valve (FSOV) having part number (P/N) B38LC50XX (where XX is 05, 06, 07, 08, 09, or 10) is installed; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of

the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To detect and correct failure of the FSOV's to close, which could result in failure of the engine fire shut-off system, and consequent inability to extinguish an engine fire, accomplish the following:

Repetitive Operational Tests

(a) Within 600 flight hours after the effective date of this AD, perform an operational test of the 4 FSOV's on the airplane, in accordance with Airbus All Operator Telex (AOT) 29-22, dated November 24, 1997. If any FSOV fails the test, prior to further flight, replace the FSOV with a new or serviceable FSOV, in accordance with the AOT. Repeat the operational test thereafter at intervals not to exceed 600 flight hours.

Spares

(b) As of the effective date of this AD, no person shall install an FSOV, part number (P/N) B38LC50XX (where XX is 05, 06, 07, 08, 09, or 10), on any airplane, unless a successful operational test has been performed in accordance with the requirements of this AD.

Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Incorporation by Reference

(e) The actions shall be done in accordance with Airbus All Operator Telex

(AOT) 29-22, dated November 24, 1997.

This incorporation by reference was approved previously by the Director of the Federal as of September 4, 1998 (63 FR 40811, July 31, 1998). Copies may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. Copies may be

inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Note 3: The subject of this AD is addressed in French airworthiness directive 98-356-259(B), dated September 9, 1998.

(f) This amendment becomes effective on May 19, 2000.

Issued in Renton, Washington, on April 7, 2000.

Donald L. Rigglin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 00-9246 Filed 4-13-00; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2000-NM-82-AD; Amendment 39-11680; AD 2000-07-25]

RIN 2120-AA64

Airworthiness Directives; Gulfstream Model G-IV Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to certain Gulfstream Model G-IV series airplanes. This action requires modification of the power feeder cable assemblies of the left and right engine alternators. This amendment is prompted by an incident of an in-flight engine fire on a Model G-IV series airplane due to chafing of the power feeder cable assembly of an engine alternator. The actions specified in this AD are intended to prevent interference and chafing between the alternator power feeder cables and adjacent structure, which could result in an electrical short circuit and consequent fire ignition source in the engine compartment.

DATES: Effective May 1, 2000.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 1, 2000.

Comments for inclusion in the Rules Docket must be received on or before June 13, 2000.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114,

Attention: Rules Docket No. 2000-NM-82-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

The service information referenced in this AD may be obtained from Gulfstream Aerospace Corporation, P.O. Box 2206, M/S D-10, Savannah, Georgia 31402-9980. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia 30349; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Neil Berryman, Aerospace Engineer; ACE-118A, FAA, Small Airplane Directorate, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia 30349; telephone (770) 703-6098; fax (770) 703-6097.

SUPPLEMENTARY INFORMATION: The FAA has received a report indicating that an in-flight engine fire occurred on a Gulfstream Model G-IV series airplane. Inspection revealed that the fire was contained within the engine nacelle. Further investigation indicated the ignition source of the fire to be an alternator power feeder cable chafing against the Approach Idle Solenoid fuel line. Such chafing, if not corrected, could result in an electrical short circuit that could spark and ignite a fire in the engine compartment.

Explanation of Relevant Service Information

The FAA has reviewed and approved Gulfstream IV Customer Bulletin No. 112, dated February 15, 2000, and Gulfstream IV Aircraft Service Change No. 410, dated February 11, 2000. The customer bulletin references the aircraft service change, which describes procedures for modification of the power feeder cable assemblies of the left and right engine alternators. Modification procedures include replacing the cables, rerouting the cables to ensure adequate clearance between the cables and adjacent structure, and ensuring that the cables are properly connected to the terminals and that torque values are within specified limits. Procedures also include installing additional brackets and clamps to secure the cables and eliminate slack in the cables.

Explanation of the Requirements of the Rule

Since an unsafe condition has been identified that is likely to exist or

develop on other Gulfstream Model G-IV series airplanes of the same type design, this AD is being issued to prevent interference and chafing between the alternator feeder cables and adjacent structure, which could result in an electrical short circuit and consequent fire ignition source in the engine compartment. This AD requires modifying the power feeder cable assemblies of the left and right engine alternators. The actions are required to be accomplished in accordance with the service information described previously.

Determination of Rule's Effective Date

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments