

(iv) If there is a strike or other labor dispute involving a work stoppage of workers in progress, but such strike or other labor dispute is not certified under paragraph (e)(22)(i) of this section, or the Service has not otherwise been informed by the Secretary that such a strike or labor dispute is in progress, the Commissioner shall not deny entry to an applicant for E status.

\* \* \* \* \*

3. Section 214.6 is amended by revising paragraph (b) to read as follows:

**§ 214.6 Canadian and Mexican citizens seeking temporary entry to engage in business activities at a professional level.**

\* \* \* \* \*

(b) *Definitions.* As used in this section, the terms:

*Business activities at a professional level* means those undertakings which require that, for successful completion, the individual has a least a baccalaureate degree or appropriate credentials demonstrating status as a professional in a profession set forth in Appendix 1603.D.1 of the NAFTA.

*Business person*, as defined in the NAFTA, means a citizen of Canada or Mexico who is engaged in the trade of goods, the provision of services, or the conduct of investment activities.

*Engage in business activities at a professional level* means the performance of prearranged business activities for a United States entity, including an individual. It does not authorize the establishment of a business or practice in the United States in which the professional will be, in substance, self-employed. A professional will be deemed to be self-employed if he or she will be rendering services to a corporation or entity of which the professional is the sole or controlling shareholder or owner.

*Temporary entry*, as defined in the NAFTA, means entry without the intent to establish permanent residence. The alien must satisfy the inspecting immigration officer that the proposed stay is temporary. A temporary period has a reasonable, finite end that does not equate to permanent residence. In order to establish that the alien's entry will be temporary, the alien must demonstrate to the satisfaction of the inspecting immigration officer that his or her work assignment in the United States will end at a predictable time and that he or she will depart upon completion of the assignment.

\* \* \* \* \*

Dated: August 13, 1997.

**Doris Meissner,**  
*Commissioner, Immigration and Naturalization Service.*

[FR Doc. 98-601 Filed 1-8-98; 8:45 am]

BILLING CODE 4410-10-M

---

**NUCLEAR REGULATORY COMMISSION**

**10 CFR Part 50**

RIN 3150-AF73

**Codes and Standards; IEEE National Consensus Standard, Withdrawal; Correction**

**AGENCY:** Nuclear Regulatory Commission.

**ACTION:** Direct final rule; correction.

**SUMMARY:** This document corrects a notice appearing in the **Federal Register** on December 23, 1997 (62 FR 66977). This action is necessary to correct an erroneous **Federal Register** citation.

**FOR FURTHER INFORMATION CONTACT:** Michael T. Lesar, Acting Chief, Rules and Directives Branch, Division of Administrative Services, Office of Administration, Washington, D.C. 20555-0001, telephone (301) 415-7163.

**SUPPLEMENTARY INFORMATION:** On page 66977, in the first column, in the last paragraph, in the second line, **Federal Register** citation "(62 FR 53933)" is corrected to read "(62 FR 53932)".

Dated at Rockville, Maryland, this 6th day of January 1998.

For the Nuclear Regulatory Commission.

**Michael T. Lesar,**  
*Acting Chief Rules and Directives Branch, Division of Administrative Services, Office of Administration.*

[FR Doc. 98-533 Filed 1-8-98; 8:45 am]

BILLING CODE 7590-01-P

---

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. 97-ANE-16; Amendment 39-10270; AD 98-01-06]

RIN 2120-AA64

**Airworthiness Directives; Precision Airmotive Corporation Carburetors**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment supersedes an existing airworthiness directive (AD),

applicable to Precision Airmotive Corporation carburetors, that currently requires the inspection of those carburetors equipped with a two-piece venturi at each annual inspection to determine if the primary venturi is loose or missing, and requires the replacement of a two-piece venturi with a one-piece venturi within 48 months after the effective date of the existing AD. This amendment eliminates the requirement to install a one-piece venturi, and allows the installation of a one-piece venturi on affected carburetors as an optional terminating action; or, requires repetitive inspections of a two-piece venturi on affected carburetors. This AD also adds an additional carburetor model, and requires the installation of a new fuel nozzle on certain carburetors when a one-piece venturi is installed. This amendment is prompted by service difficulty reports describing engines that fail to attain rated power, run rough, or experience power loss after installation of a one-piece venturi in accordance with the existing AD, and by incidents of forced landings of aircraft powered by engines modified to comply with the existing AD. The actions specified by this AD are intended to prevent disruption of fuel flow to the engine resulting in failure to attain rated power, power loss in flight, and forced landings.

**DATES:** Effective February 13, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of February 13, 1998.

**ADDRESSES:** The service information referenced in this AD may be obtained from Precision Airmotive Corporation, 3220 100th Street SW., Building E, Everett, WA 98204; telephone (206) 353-8181, fax (206) 348-3545. This information may be examined at the Federal Aviation Administration (FAA), New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, Burlington, MA 01803-5299; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Richard Simonson, Aerospace Engineer, Seattle Aircraft Certification Office, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW, Renton, WA 98055-4056; telephone (425) 227-2597, fax (425) 227-1181.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding airworthiness directive (AD) 93-18-03, Amendment 39-8688

(58 FR 50843, September 29, 1993), which is applicable to Precision Airmotive Corporation (formerly Facet Aerospace Products Corporation and Marvel-Schebler Corporation) Model MA-3A, MA-3PA, MA-3SPA, and MA-4SPA carburetors equipped with two-piece venturis, was published in the **Federal Register** on August 1, 1997 (62 FR 41321). That action proposed to require repetitive inspections of two-piece venturis, and to allow installation of one-piece venturis as an optional terminating action for those repetitive inspections provided certain conditions are met.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

One commenter (the manufacturer) states that the AD should not allow indefinite repetitive inspections, with no end date for accomplishment of the terminating action (i.e. installation of a one-piece venturi), since the terminating action is necessary and there has been no sound technical basis established for abandonment of the required date for compliance. The FAA does not concur. The FAA has determined that the terminating action should be optional, based upon the engines reported running rough with the one-piece venturi. Continued repetitive inspections of the two-piece venturi or replacement with a one-piece venturi will provide an acceptable level of safety.

The commenter also states that by allowing the reinstallation of two-piece venturis, non-serviceable parts might be used, particularly since the manufacture of these parts was discontinued in the late 1980s. If reinstallation of a two-piece venturi must be allowed, the commenter believes that this option should be viewed as an Alternative Method of Compliance (AMOC) and not as an option within the AD itself. The FAA does not concur. The FAA has determined that the risk of installing non-serviceable two-piece venturis is no greater than installing any other non-serviceable parts. Obviously, if there are no serviceable two-piece venturis available, the operator must continue operation of the one-piece venturi and install a new fuel nozzle in accordance with paragraphs (c), (d), or (e), as applicable, of this AD. Since the options of installing a two-piece venturi and maintaining continuing inspections, or installing a new fuel nozzle on carburetors with one-piece venturis when engines run rough or do not obtain rated power both result in equivalent levels of safety, the FAA has

determined that both options should be presented in the AD instead of relegating the installation of a two-piece venturi to an AMOC.

Two commenters concur with the AD as proposed.

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that 30,000 carburetors installed on aircraft of U.S. registry will be affected by this AD, that it will take approximately 2 work hours per engine to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$75 per carburetor. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$5,850,000.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-8688 (58 FR 50843, September 29, 1993) and by adding a new airworthiness directive, Amendment 39-10270, to read as follows:

**98-01-06 Precision Airmotive Corporation:** Amendment 39-10270. Supersedes airworthiness directive (AD) 93-18-03, Amendment 39-8688.

**Applicability:** Precision Airmotive Corporation (formerly Facet Aerospace Products Corporation and Marvel-Schebler Corporation) Model MA-3, MA-3A, MA-3PA, MA-3SPA, MA-4SPA carburetors installed on but not limited to Textron Lycoming O-235, O-290, and O-320 series engines, and Teledyne Continental Motors A-65, A-75, C-75, C-85, C-90, C-115, C-125, C-145, O-200, and O-300 series engines. These engines are installed on, but not limited to, normally aspirated reciprocating engine powered aircraft manufactured by Cessna, Piper, Raytheon, and Mooney.

**Note 1:** This AD applies to each carburetor identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For carburetors that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (f) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent disruption of fuel flow to the engine resulting in failure to attain rated power, power loss in flight, and forced landings, accomplish the following, accomplish the following:

(a) For Precision Airmotive Corporation Model MA-3A, A-3PA, MA-3SPA, and MA4-SPA carburetors:

(1) If not previously accomplished, prior to further flight, inspect the carburetor to determine if a two-piece venturi is installed. Carburetors that have the letter "V" stamped or etched on the lower portion of the data plate, or that have a black, yellow, or blue data plate showing the Precision Airmotive Corporation name and logo, or that have a black Facet Aerospace Products data plate with a serial number beginning with 750, are already equipped with a one-piece venturi and no further action is necessary provided the engine does not subsequently run rough or experience power loss.

(2) If a two-piece venturi is installed, inspect the carburetor at each annual, 100-hour, or progressive inspection, to determine if the primary venturi is loose or missing. If either of these conditions is found, prior to

further flight, repair the carburetor by installing a serviceable two-piece venturi or by installing a one-piece venturi in accordance with Precision Airmotive Service Bulletin (SB) No. MSA-2, Revision 1, dated November 11, 1991, Revision 2, dated December 28, 1993, or Revision 3, dated October 10, 1995. Installing a one-piece venturi constitutes terminating action for the repetitive inspection requirements of this paragraph.

(3) If a one-piece venturi is already installed, or installed in accordance with sub-paragraph (2) of this paragraph, and the engine subsequently runs rough or experiences power loss, accomplish either of the following:

(i) Modify the carburetor in accordance with paragraphs (c), (d) or (e) of this AD, as applicable; or

(ii) Install a carburetor containing a two-piece venturi and resume the repetitive inspections required by paragraph (a)(2) of this AD.

(b) For Precision Airmotive Corporation Model MA-3 series carburetors: at the next annual, 100-hour, or progressive inspection, whichever occurs first, after the effective date of this AD, inspect the carburetor to determine if the primary venturi is loose or missing. If either of these conditions are

found, prior to further flight, repair the carburetor by installing a serviceable two-piece venturi, or replace the entire carburetor with a serviceable carburetor. Repeat this inspection at each annual, 100-hour, or progressive inspection.

(c) For Precision Airmotive Corporation Model MA-3SPA series carburetors with part numbers (P/N) 10-4894 or 10-4115-1, installed on Teledyne Continental Model O-200A series engines modified on or after the effective date of this AD by installing a one-piece venturi, install a new fuel nozzle in accordance with Precision Airmotive SB MSA-7, dated September 30, 1994, at the time of installation of the one-piece venturi.

(d) For Precision Airmotive Corporation Model MA-3SPA series carburetors with P/Ns 10-4895, 10-4439, or 10-3237, installed on Teledyne Continental Model O-300 or C-145 series engines modified on or after the effective date of this AD by installing a one-piece venturi, install a new fuel nozzle in accordance with Precision Airmotive SB No. MSA-8, dated July 10, 1995, at the time of installation of the one-piece venturi.

(e) For Precision Airmotive Corporation Model MA-3SPA series carburetors with P/Ns 10-4240, 10-4252, 10-4252-1, or 10-4457, installed on Teledyne Continental Model C-75, C-85, or C-90 series engines

modified on or after the effective date of this AD by installing a one-piece venturi, install a new fuel nozzle in accordance with Precision Airmotive SB No. MSA-9, dated October 10, 1995, at the time of installation of the one-piece venturi.

(f) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle Aircraft Certification Office.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the Seattle Aircraft Certification Office.

(g) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the aircraft to a location where the inspection requirements of this AD can be accomplished.

(h) The actions required by this AD shall be done in accordance with the following Precision Airmotive Corporation SBs:

Document No.	Pages	Revision	Date
MSA-2 ..... Total Pages: 3.	1-3	1	November 11, 1991.
MSA-2 ..... Total Pages: 3.	1-3	2	December 28, 1993.
MSA-2 ..... Total Pages: 4.	1-4	3	October 10, 1995.
MSA-7 ..... Total Pages: 3.	1-3	Original	September 30, 1994.
MSA-8 ..... Total Pages: 3.	1-3	Original	July 10, 1995.
MSA-9 ..... Total Pages: 3.	1-3	Original	October 10, 1995.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Precision Airmotive Corporation, 3220 100th Street SW., Building E, Everett, WA 98204; telephone (206) 353-8181, fax (206) 348-3545. Copies may be inspected at the FAA, New England Region, Office of the Assistant Chief Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) This amendment becomes effective on February 13, 1998.

Issued in Burlington, Massachusetts, on December 23, 1997.

**Jay J. Pardee,**

*Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 98-70 Filed 1-8-98; 8:45 am]

**BILLING CODE 4910-13-U**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. 96-CE-66-AD; Amendment 39-10273; AD 98-01-10]

**RIN 2120-AA64**

**Airworthiness Directives; Empresa Brasileira de Aeronautica S.A., Models EMB-110P1 and EMB-110P2 Airplanes**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment supersedes Airworthiness Directive (AD) 87-03-10, which currently requires repetitively inspecting the fillet area of both the left and right main landing gear (MLG) wheel axle/piston tube support junction area for cracks on Empresa Brasileira de Aeronautica S.A. (EMBRAER) Models

EMB-110P1 and EMB-110P2 airplanes, and requires replacing any MLG wheel axle/piston tube assembly where a crack is found. AD 87-03-10 also provided the option of reworking this area when no cracks were found as terminating action for the repetitive inspections. The Federal Aviation Administration's policy on aging commuter-class aircraft is to eliminate or, in certain instances, reduce the number of certain repetitive short-interval inspections when improved parts or modifications are available. This AD requires the following on EMBRAER Models EMB 110-P1 and EMB 110-P2 airplanes that do not have an "R" stamped on both the left and right MLG wheel axle/piston tube assembly end-piece: inspecting (one-time) the fillet area of each MLG wheel axle/piston tube support junction area to assure that the area is free of cracks, replacing any MLG wheel axle/piston tube assembly if a crack is found, and reworking this area on both the left