

Idaho and Eastern Oregon and Marketing Order No. 959 regulates onions grown in South Texas. Fresh onion shipments from Idaho-Eastern Oregon are regulated throughout the year, while onion shipments from South Texas had been regulated from March 1 through June 15 each year. On the basis of past shipment data, the Secretary determined that onions imported during the March 10 through June 15 period were in most direct competition with onions grown in South Texas and found that the minimum grade, size, quality, and maturity requirements for onions imported during that period should be the same as those established for South Texas onions under Marketing Order No. 959. The Secretary further determined that onions imported during the June 16 through March 9 period were in most direct competition with onions grown in Idaho-Eastern Oregon and that the minimum grade, size, quality, and maturity requirements for onions imported during that period should be the same as those established for Idaho-Eastern Oregon onions under Marketing Order No. 959.

Based on a recommendation of the South Texas Onion Committee (committee), the agency responsible for local administration of Marketing Order No. 959, the Department is proposing to change the end of the South Texas regulatory period from June 15 to June 4. Because South Texas onions would no longer be regulated after June 4, and Idaho-Eastern Oregon onions are regulated throughout the year, it is proposed that onions imported during the March 10 through June 4 period are in most direct competition with onions produced in South Texas and that the minimum grade, size, quality, and maturity requirements established under the South Texas marketing order should apply to onions imported during the March 10 through June 4 period, instead of the current March 10 through June 15 period. Imports of onions during the June 5 through March 9 period should be required to meet minimum grade, size, quality, and maturity requirements based on those established under the Idaho-Eastern Oregon marketing order.

In accordance with section 8e of the Act, the U.S. Trade Representative has concurred with the issuance of this proposed rule.

Based on the above, the Administrator of the AMS has determined that this action would not have a significant economic impact on a substantial number of small entities.

A 30-day comment period is provided to allow interested persons to respond to this proposal. All written comments

timely received will be considered before a final determination is made on this matter.

List of Subjects in 7 CFR Part 980

Food grades and standards, Imports, Marketing agreements, Onions, Potatoes, Tomatoes.

For the reasons set forth in the preamble, 7 CFR part 980 is proposed to be amended as follows:

PART 980—VEGETABLES; IMPORT REGULATIONS

1. The authority citation for 7 CFR part 980 continues to read as follows:

Authority: 7 U.S.C. 601–674.

§ 980.117 [Amended]

2. In § 980.117, paragraph (a)(2) is amended by removing “June 16” and adding in its place “June 5” and by removing “June 15” and adding in its place “June 4”; paragraph (b)(1) is amended by removing “June 16” and adding in its place “June 5”; and paragraph (b)(2) is amended by removing “June 15” and adding in its place “June 4”.

Dated: January 31, 1996.

Sharon Bomer Lauritsen,
Deputy Director, Fruit and Vegetable Division.
[FR Doc. 96–2751 Filed 2–8–96; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Chapter I

[Summary Notice No. PR–96–1]

Petition for Rulemaking Summary of Petitions Received; Dispositions of Petitions Issued

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of petitions for rulemaking received and of dispositions of prior petitions.

SUMMARY: Pursuant to FAA’s rulemaking provisions governing the application, processing, and disposition of petitions for rulemaking (14 CFR Part 11), this notice contains a summary of certain petitions requesting the initiation of rulemaking procedures for the amendment of specified provisions of the Federal Aviation Regulations and of denials or withdrawals of certain petitions previously received. The purpose of this notice is to improve the public’s awareness of, and participation in, this aspect of FAA’s regulatory activities. Neither publication of this

notice nor the inclusion or omission of information in the summary is intended to affect the legal status of any petition or its final disposition.

DATES: Comments on petitions received must identify the petition docket number involved and must be received April 9, 1996.

ADDRESSES: Send comments on any petition in triplicate to: Federal Aviation Administration, Office of the Chief Counsel, Attn: Rules Docket No. _____, 800 Independence Avenue, SW., Washington, DC 20591.

The petition, any comments received, and a copy of any final disposition are filed in the assigned regulatory docket and are available for examination in the Rules Docket (AGC–200), Room 915G, FAA Headquarters Building (FOB 10A), 800 Independence Ave., SW., Washington, DC 20591; telephone (202) 267–3132. Comments may also be sent electronically to the following internet address: nprmcmts@mail.hq.faa.gov.

FOR FURTHER INFORMATION CONTACT:

Mr. D. Michael Smith, Office of Rulemaking (ARM–1), Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone (202) 267–7470.

This notice is published pursuant to paragraph (b) and (f) of § 11.27 of Part 11 of the Federal Aviation Regulations (14 CFR Part 11).

Issued in Washington, DC on February 2, 1996.

Donald P. Byrne,

Assistant Chief Counsel for Regulations.

Disposition of Petitions

Docket No.: 27375.

Petitioner: Professional Pilots Federation.

Sections of the FAR Affected: 14 CFR 121.383(c).

Description of Rule change Sought: To repeal § 121.383(c), referred to as the Age 60 rule.

Petitioner’s Reason for the Request: The petitioner feels that Federal law and policy, operational regulatory developments since promulgation of the rule, and the results of the Hilton Study warrant the removal.

Denial; December 28, 1995.

[FR Doc. 96–2851 Filed 2–8–96; 8:45 am]

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14 CFR Part 39

[Docket No. 95-NM-122-AD]

Airworthiness Directives; Beech Model BAe 125-800A and Model Hawker 800 Airplanes**AGENCY:** Federal Aviation Administration, DOT.**ACTION:** Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Beech Model BAe 125-800A and Model Hawker 800 airplanes. This proposal would require modification of the airframe structure in the lower area of the fuselage aft of the wing rear spar. For certain airplanes, this proposal would also require a functional test to determine if a particular bolt fouls the flap control system. This proposal is prompted by reports of restricted control of the aileron due to water accumulation that froze in the area around an aileron pulley located in the lower area of the fuselage aft of the wing rear spar. The actions specified by the proposed AD are intended to prevent such water accumulation, which could freeze and result in restricted control of the ailerons; subsequently, this could reduce the pilot's ability to initiate roll control during critical phases of flight.

DATES: Comments must be received by March 22, 1996.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 95-NM-122-AD, 1601 Lind Avenue SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Raytheon Aircraft Co., Manger Service Engineering, Hawker Customer Support Department, P.O. Box 85, Wichita, Kansas 67201-0085. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: William Schroeder, Aerospace Engineer, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington 98055-4056; telephone (206) 227-2148; fax (206) 227-1149.

SUPPLEMENTARY INFORMATION:**Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 95-NM-122-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 95-NM-122-AD, 1601 Lind Avenue SW., Renton, Washington 98055-4056.

Discussion

The Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom, recently notified the FAA that an unsafe condition may exist on certain Beech Model BAe 125-800A and Model Hawker 800 airplanes. The CAA advises that it has received several reports of restricted control of the aileron. Investigation revealed that potable water leaked from the potable water supply system of the galley and lavatory and accumulated in the lower area of the fuselage aft of the wing rear spar. This water then drained to and accumulated in the area around an aileron pulley located in the subject area, and eventually froze. These conditions, if not corrected, could result in restricted control of the ailerons, and, subsequently, reduce the pilot's ability

to initiate roll control during critical phases of flight.

The manufacturer has issued Hawker Service Bulletin SB.53-82-3566G, dated March 1, 1995; Revision 1, dated March 14, 1995; Revision 2, dated May 3, 1995; and Revision 3, December 14, 1995 (for certain airplanes, excluding Model BAe 125-800A airplane having constructor's No. 258186). The manufacturer has also issued Hawker Service Bulletin SB.53-85-3566D, dated March 10, 1995, and Revision 1, dated May 23, 1995 (for Model BAe 125-800A airplane having constructor's number 258186). These service bulletins describe procedures for modification of the airframe structure in the lower area of the fuselage aft of the wing rear spar. The modification entails the following actions:

1. Installing new drain holes in the fairing skin assembly of the main landing gear (MLG);
2. Plugging existing vent holes of the keel stringers;
3. Installing a new water barrier diaphragm between the forward diaphragm assembly of the MLG and the aft diaphragm assembly;
4. Installing a blanking plate on the forward diaphragm assembly;
5. Removing existing drain valves from the fuselage keel skin;
6. Blanking off an existing drain valve hole using a new patch plate;
7. Installing two new drain spouts, a drain hose, and an outlet fairing; and
8. Performing a functional test to determine if a bolt fouls the flap control system (Revision 3 of Service Bulletin SB.53-82-3566G only).

Accomplishment of the modification will prevent any leaked fluids from collecting around the aileron pulley, and will allow any water build-up in the fuselage keel area to drain away. The CAA classified these service bulletins as mandatory in order to assure the continued airworthiness of these airplanes in the United Kingdom.

This airplane model is manufactured in CAA and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the CAA has kept the FAA informed of the situation described above. The FAA has examined the findings of the CAA, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same

type design, the proposed AD would require modification of the airframe structure in the lower area of the fuselage aft of the wing rear spar. For certain airplanes, the proposed AD would also require a functional test to determine if a bolt fouls the flap control system. The actions would be required to be accomplished in accordance with the service bulletins described previously. If any fouls are detected, the repair would be required to be accomplished in accordance with a method approved by the FAA.

The FAA estimates that 163 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 25 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would be supplied by the manufacturer at no cost to the operators. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$244,500, or \$1,500 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Beech Aircraft Corporation. (Formerly DeHavilland; Hawker Siddeley; British Aerospace, plc; Raytheon Corporate Jets, Inc.): Docket 95-NM-122-AD.

Applicability: Model BAe 125-800A (including military variants C-29A and U-125); and Model Hawker 800 airplanes, excluding airplanes having constructor's numbers 258079 and 258213; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (d) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To prevent restricted control of the ailerons, which could reduce the pilot's ability to initiate roll control during critical phases of flight, accomplish the following:

(a) For all airplanes, except Model BAe 125-800A airplane having constructor's number 258186: Within 6 months after the effective date of this AD, modify (including functional test) the airframe structure in the lower area of the fuselage aft of the wing rear spar, in accordance with Hawker Service Bulletin SB.53-82-3566G, Revision 3, December 14, 1995.

(b) For airplanes identified in paragraph (a) of this AD on which Hawker Modification 253566G has been installed prior to the effective date of this AD, in accordance with Hawker Service Bulletin SB.53-82-3566G, dated March 1, 1995, Revision 1, dated March 14, 1995, or Revision 2, dated May 3,

1995: Within 30 days after the effective date of this AD, perform a functional test to determine if a bolt fouls the flap control system, in accordance with paragraph 2.A.(18) of the Accomplishment Instructions of Hawker Service Bulletin SB.53-82-3566G, Revision 3, December 14, 1995. If any foul is detected, prior to further flight, repair in accordance with a method approved by the Manager, Standardization Branch, ANM-113, Transport Airplane Directorate, FAA.

(c) For Model BAe 125-800A airplane having constructor's number 258186: Within 6 months after the effective date of this AD, modify the airframe structure in the lower area of the fuselage aft of the wing rear spar, in accordance with Hawker Service Bulletin SB.53-85-3566D, dated March 10, 1995, or Revision 1, dated May 23, 1995.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on February 5, 1996.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 96-2868 Filed 2-8-96; 8:45 am]

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DEPARTMENT OF THE TREASURY

Office of the Under Secretary for Domestic Finance

17 CFR Parts 400 and 420

RIN 1505-AA53

Government Securities Act Regulations: Large Position Rules

AGENCY: Office of the Under Secretary for Domestic Finance, Treasury.

ACTION: Notice of extension of time for submission of comments.

SUMMARY: This document extends until March 18, 1996, the deadline for the submission of comments on the notice of proposed rulemaking addressing large position recordkeeping and reporting requirements for certain Treasury