

between W2132 (or W0132) and power feeder wire bundle W0142. These actions would be required to be accomplished in accordance with the service letter described previously.

Operators should note that replacement of the steel clamp with the nylon clamp and the installation of additional clamps to prevent contact between W2132 (or W0132) and power feeder wire bundle W0142 are specified as "recommended actions" in the referenced Boeing service letter. However, this proposed rule would mandate accomplishment of those actions. The FAA finds that those actions must be accomplished in order to provide an adequate level of safety for the affected fleet. The FAA has determined that, in cases where certain known unsafe conditions exist, and where actions to detect and correct that unsafe condition can be readily accomplished, those actions must be required.

The proposed AD also would require that operators submit a report of any damage found during the inspection that would be required by this proposed AD. The information obtained from these reports will enable the FAA to determine how widespread any damage is in the fleet.

There are approximately 620 Model 737-300, -400, and -500 series airplanes of the affected design in the worldwide fleet. The FAA estimates that 195 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 1 work hour per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$25 per airplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$16,575, or \$85 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1)

is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### **PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

#### **§ 39.13 [Amended]**

2. Section 39.13 is amended by adding the following new airworthiness directive:

Boeing: Docket 95-NM-117-AD.

*Applicability:* Model 737-300, -400, and -500 series airplanes, as listed in Boeing Service Letter 737-SL-24-106, dated March 10, 1995; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (c) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

*Compliance:* Required as indicated, unless accomplished previously.

To prevent short circuiting of a wire bundle located in the electrical/electronics

(E/E) equipment bay, which could result in smoke and fire, accomplish the following:

(a) Within 12 months after the effective date of this AD, accomplish the requirements of paragraphs (a)(1), (a)(2), and (a)(3), of this AD in accordance with Boeing Service Letter 737-SL-24-106, dated March 10, 1995.

(1) Perform a visual inspection for damage of the wire bundle and clamps in the E/E compartment. If any damage is detected, prior to further flight, repair in accordance with the service letter.

(2) Reclamp wire bundle W2132 (or W0132) by removing the steel cushioned clamp and installing nylon clamp on the aft side of the existing nut and bolt hole at body station (BS) 360, water line (WL) 203, left buttock line (LBL) 57, in accordance with the service letter.

(3) Install additional clamps to wire bundles W2132 (or W0132) and power feeder wire bundle W0142, in accordance with the service letter.

(b) Within 10 days after detecting any damage to the wire bundle or clamp as a result of the inspection required by paragraph (a) of this AD, submit a report of the damage findings to the FAA, Transport Airplane Directorate, Seattle Manufacturing Inspection District Office (MIDO), Attention: George Carter, 1601 Lind Avenue SW., Renton, Washington 98055-4056; telephone (206) 237-6229; fax (206) 965-0264.

Information collection requirements contained in this regulation have been approved by the Office of Management and Budget (OMB) under the provisions of the Paperwork Reduction Act of 1980 (44 U.S.C. 3501 *et seq.*) and have been assigned OMB Control Number 2120-0056.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on October 6, 1995.

Gary L. Killion,

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 95-25451 Filed 10-12-95; 8:45 am]

**BILLING CODE 4910-13-U**

**14 CFR Part 39****[Docket No. 95-CE-13-AD]****Airworthiness Directives; Fairchild Aircraft SA226 and SA227 Series Airplanes****AGENCY:** Federal Aviation Administration, DOT.**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes to revise Airworthiness Directive (AD) 95-17-09, which requires relocating the left-hand (LH) and right-hand (RH) essential bus current limiters (225 amp) to the battery bus (main bus tie) on certain Fairchild Aircraft SA226 and SA227 series airplanes. The FAA has determined that the applicability of the current AD should be changed to reflect a different serial number range and model designation of certain SA227 series airplanes. The proposed action would retain the essential bus current limiter relocations required by AD 95-17-09, and would revise the Applicability section of that AD. The actions specified by the proposed AD are intended to prevent failure of the LH and RH essential bus when engine failure results in a blown generator current limiter, which could result in loss of airplane electrical power.

**DATES:** Comments must be received on or before December 11, 1995.

**ADDRESSES:** Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Assistant Chief Counsel, Attention: Rules Docket No. 95-CE-13-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from Field Support Engineering, Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490; telephone (210) 824-9421; facsimile (210) 820-8609. This information also may be examined at the Rules Docket at the address above.

**FOR FURTHER INFORMATION CONTACT:** Mr. George R. Hash, Aerospace Engineer, FAA, Airplane Certification Office, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150; telephone (817) 222-5134; facsimile (817) 222-5959.

**SUPPLEMENTARY INFORMATION:****Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as

they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 95-CE-13-AD." The postcard will be date stamped and returned to the commenter.

**Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Assistant Chief Counsel, Attention: Rules Docket No. 95-CE-13-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

**Discussion**

On August 10, 1995, the FAA issued AD 95-17-09, Amendment 39-9339 (60 FR 43361, August 21, 1995), to require relocating the left-hand (LH) and right-hand (RH) essential bus current limiters (225 amp) to the battery bus (main bus tie) on certain Fairchild Aircraft SA226 and SA227 series airplanes. AD 95-17-09 specifies accomplishment of the relocations in accordance with Fairchild Aircraft Engineering Kit Drawing 27K82376, "Current Limiter Rebusung Kit," as referenced in Fairchild Service Bulletin (SB) 226-24-034, SB 227-24-015, and SB CC7-24-002, all Issued: September 29, 1994.

A safety recommendation detailing potential electrical failure problems on Fairchild SA226 and SA227 series airplanes prompted AD 95-17-09. Flight simulation revealed that electrical power loss could occur on the affected airplanes because of failure of the LH essential bus. Switching delays between the left and right side electrical systems result in left generator motor action,

which could then cause the left side current limiter to open. This would result in failure of the left essential bus, which will result in loss of alternating current (AC) power to the primary attitude indicator and the lighting for the standby attitude indicator.

The FAA has determined that AD 95-17-09 contains the following errors:

—Incorrect serial numbers for the Fairchild Aircraft Models SA227-AC, SA227-AT, SA227-BC, and SA227-TT airplanes in the Applicability section; and

—Incorrect designation for Model SA227-AC as SA227-AD in paragraph (a) of AD 95-17-09.

After examining all available information related to the subject discussed above, the FAA has determined that further AD action should be taken to prevent failure of the LH and RH essential bus when engine failure results in a blown generator current limiter, which could result in loss of airplane electrical power.

Since an unsafe condition has been identified that is likely to exist or develop in other Fairchild Aircraft SA226 and SA227 series airplanes of the same type design, this AD requires the same relocation of the left-hand and right hand essential bus current limiters as is contained in AD 95-17-09, and revises the Applicability section to reflect the serial number correction and model designation correction in paragraph (a) of that AD as previously described.

The FAA estimates that 876 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 4 workhours per airplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$98 per airplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$296,088 (\$338 per airplane). This figure is based on the assumption that no affected airplane owner/operator has incorporated the proposed modification. Fairchild Aircraft has informed the FAA that parts have not been distributed to any owner/operator of the affected airplanes.

The proposed action would only correct a model designation and certain serial numbers of certain SA227 series airplanes that are affected by AD 95-17-09. The cost impact upon the public specified in the proposed AD is exactly the same as that currently required by AD 95-17-09.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship

between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation

Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g); 40101, 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing Amendment 39-9339 (60 FR 43361, August 21, 1995), and by adding a new airworthiness directive (AD) to read as follows:

Fairchild Aircraft: Docket No. 95-CE-13-AD. Revises AD 95-17-09; Amendment 39-9339.

Applicability: The following model and serial number airplanes that utilize a direct current (DC) generator, certificated in any category.

Models	Serial Nos.
SA226-T, SA226-AT, SA226-TC, and SA226-T(B).	All.
SA227-AC, SA227-AT, SA227-BC, and SA227-TT.	420 through 783, and 785 through 789.
SA227-CC and SA227-DC.	784, and 790 through 883.

SB	Date	Models affected
226-24-034 .....	Sept. 29, 1994 .....	All affected SA226 models.
227-24-015 .....	Sept. 29, 1994 .....	SA227-AC, SA227-AT, SA227-BC, and SA227-TT.
CC7-24-002 .....	Sept. 29, 1994 .....	SA227-CC and SA227-DC.

(b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Fort Worth Airplane Certification Office (ACO), FAA, 2601 Meacham Boulevard, Fort Worth, Texas 76193-0150. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

(d) All persons affected by this directive may obtain copies of the document referred to herein upon request to Fairchild Aircraft, P.O. Box 790490, San Antonio, Texas 78279-0490; or may examine this document at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

(e) This amendment revises AD 95-17-09, Amendment 39-9339.

Issued in Kansas City, Missouri, on October 6, 1995.

Henry A. Armstrong,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-25440 Filed 10-12-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 95-CE-30-AD]

Airworthiness Directives; HB Flugtechnik GmbH Model HB-23/2400 Sailplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to HB

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 2,000 hours time-in-service after the effective date of this AD, unless already accomplished (compliance with AD 95-17-09).

To prevent failure of the left hand (LH) and right hand (RH) essential bus when engine failure results in a blown generator current limiter, which could result in loss of airplane electrical power, accomplish the following:

(a) Relocate the LH and RH essential bus current limiters (225 amp) to the battery bus (main bus tie) in accordance with Fairchild Aircraft Engineering Kit Drawing 27K82376, "Current Limiter Reusing Kit," as referenced in the following service bulletins (SB):

Flugtechnik GmbH (Flugtechnik) Model HB-23/2400 sailplanes. The proposed action would require inspecting (one time) the elevator control system for incorrect rigging and repetitively inspecting the threaded adjustable extension joints in the push rod to control lever connection for cracks, and, if cracked, replacing the threaded adjustable joints at both ends of the push rod. Cracking of the threaded adjustable extension joints and incorrect rigging of the elevator control system prompted the proposed action. The actions specified by the proposed AD are intended to prevent failure of the elevator control system, which, if not detected and corrected, could result in possible loss of elevator control and loss of the sailplane.

DATES: Comments must be received on or before December 14, 1995.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region,