

are invited to comment on this rule by submitting such written data, views, or arguments as they may desire.

Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 95-NM-190-AD." The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the

Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

95-20-51 Boeing: Amendment 39-9398. Docket 95-NM-190-AD.

Applicability: All Model 767-200 and 767-300 series airplanes, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (d) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To prevent the collapse of the main landing gear due to stress corrosion cracking in the outer cylinder of the aft trunnion, accomplish the following:

Note 2: The inspections required by this AD are in addition to, not in lieu of, the inspections required by AD 95-19-10, amendment 39-9372.

(a) Within 48 clock hours (not flight hours) after the effective date of this AD, perform an external general visual inspection of the lower half of the aft trunnion of the main landing gear (MLG) to detect obvious signs of damage, cracking, missing pieces; or obvious visible corrosion emanating from the aft trunnion bushing fillet seal or from the aft trunnion crossbolt hole.

Note 3: For the purpose of this AD, "external general visual inspection" means that the inspection is to be conducted within one foot of the area to be inspected. If necessary, the area should be wiped clean with a rag. Finally, mirrors and additional lighting should be used, as needed, to increase the probability of visually detecting discrepancies. This inspection does not require disassembly of the MLG.

(b) Prior to four years from the date the MLG is placed in service or overhauled, or within 48 clock hours (not flight hours) after the inspection required by paragraph (a) of this AD is accomplished, whichever occurs later, repeat the inspection required by paragraph (a) of this AD. Thereafter, repeat the inspection at intervals not to exceed 48 clock hours.

(c) If any discrepancy is detected during any inspection required by this AD, prior to further flight, repair in accordance with a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle ACO. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 4: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(f) This amendment becomes effective on October 17, 1995, to all persons except those persons to whom it was made immediately effective by telegraphic AD T95-20-51, issued on September 25, 1995, which contained the requirements of this amendment.

Issued in Renton, Washington, on October 4, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 95-25157 Filed 10-11-95; 8:45 am]
BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 95-NM-179-AD; Amendment 39-9396; AD 95-21-10]

Airworthiness Directives: Fokker Model F28 Mark 0100 and Model F28 Mark 0070 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that is applicable to all Fokker Model F28 Mark 0100 and Model F28 Mark 0070 series airplanes. This action requires revising the Limitations Section of the Airplane Flight Manual to include information that will enable the flightcrew to identify failures of the emergency direct current (DC)/alternating current (AC) bus power supply and to take appropriate corrective actions. This amendment is prompted by one report indicating that a diode failed, which resulted in battery drain and loss of the emergency DC bus power supply; and another report indicating that the circuit breaker of the transformer rectifier unit No. 3 tripped, which resulted in the loss of the emergency DC/AC bus power supply. The actions specified in this AD are intended to ensure that the flightcrew is advised of the potential hazard related to failures of the emergency bus power supply, and the procedures necessary to address it.

DATES: Effective October 27, 1995.

Comments for inclusion in the Rules Docket must be received on or before December 11, 1995.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 95-NM-179-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Information concerning this amendment may be obtained from or examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Ruth Harder, Aerospace Engineer, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (206) 227-1721; fax (206) 227-1149.

SUPPLEMENTARY INFORMATION: The Rijksluchtvaartdienst (RLD), which is the airworthiness authority for the Netherlands, recently notified the FAA that an unsafe condition may exist on all Fokker Model F28 Mark 0100 and Model F28 Mark 0070 series airplanes. The RLD advises that it has received a report that a diode failed on a Fokker Model F28 Mark 0100 series airplane. The flightcrew had no indication of this failure until the battery voltage dropped below a certain value. The RLD also advises that it has received another similar report, but on a Fokker Model Mark 0070 series airplane, in which the circuit breaker of the transformer rectifier unit No. 3 of the emergency

direct current (DC) bus supply tripped; this situation resulted in an oscillating behavior of the electrical relays, causing the failure of not only the systems powered by the emergency DC bus, but also of the systems powered by the emergency alternating current (AC) bus. The cause of these failures is unknown at this time.

Failure of a diode in the emergency DC bus supply could result in a battery drain, and the loss of the emergency DC bus and the subsequent loss of all systems powered by it. If the circuit breaker of the transformer rectifier unit No. 3 of the emergency DC bus supply trips, the resultant oscillations of the electrical relays could result in loss of both the emergency DC bus and emergency AC bus; this situation could lead to loss of on-side displays, autopilot, pressure control, and all communications, which could reduce the ability of the flightcrew to control the airplane.

Fokker has developed procedural information, for inclusion in the Airplane Flight Manual (AFM) of the affected airplanes, that will enable the flight crew to identify failures of the emergency DC/AC bus power supply and to take appropriate corrective actions. The RLD classified this AFM material as mandatory, and issued Dutch airworthiness directive BLA 1995-089/2 (A), dated September 29, 1995, in order to assure the continued airworthiness of these airplanes in the Netherlands.

This airplane model is manufactured in the Netherlands and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the RLD has kept the FAA informed of the situation described above. The FAA has examined the findings of the RLD, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, this AD is being issued to ensure that the flightcrew is advised of the potential hazard related to failures of the emergency DC/AC bus power supply, and the procedures necessary to address it. This AD requires revising the Abnormal and Normal Procedures sections of the FAA-approved AFM to include information that will enable the

flightcrew to identify failures of the emergency DC/AC bus power supply and to take appropriate procedures necessary to address it.

This is considered to be interim action. The manufacturer has advised it currently is developing a modification that will positively address the unsafe condition addressed by this AD. Once this modification is developed, approved, and available, the FAA may consider additional rulemaking.

Since a situation exists that requires the immediate adoption of this regulation, it is found that notice and opportunity for prior public comment hereon are impracticable, and that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

Although this action is in the form of a final rule that involves requirements affecting flight safety and, thus, was not preceded by notice and an opportunity for public comment, comments are invited on this rule. Interested persons are invited to comment on this rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified under the caption **ADDRESSES**. All communications received on or before the closing date for comments will be considered, and this rule may be amended in light of the comments received. Factual information that supports the commenter's ideas and suggestions is extremely helpful in evaluating the effectiveness of the AD action and determining whether additional rulemaking action would be needed.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify the rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this AD will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this rule must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 95-NM-179-AD." The postcard will be date stamped and returned to the commenter.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the

national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

The FAA has determined that this regulation is an emergency regulation that must be issued immediately to correct an unsafe condition in aircraft, and that it is not a "significant regulatory action" under Executive Order 12866. It has been determined further that this action involves an emergency regulation under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979). If it is determined that this emergency regulation otherwise would be significant under DOT Regulatory Policies and Procedures, a final regulatory evaluation will be prepared and placed in the Rules Docket. A copy of it, if filed, may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 USC 106(g), 40101, 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

95-21-10 Fokker: Amendment 39-9396. Docket 95-NM-179-AD.

Applicability: All Model F28 Mark 0070 and Model F28 Mark 0100 series airplanes, certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To ensure that the flightcrew is advised of the potential hazard related to failures of the emergency direct current (DC)/alternating current (AC) bus power supply, and the procedures necessary to address it, accomplish the following:

(a) For all airplanes: Within 7 days after the effective date of this AD, revise the Abnormal Procedures section of the FAA-approved

Airplane Flight Manual (AFM) to include the following statement. This may be accomplished by inserting a copy of this AD in the AFM.

"Section 4—Abnormal Procedures Add to Sub-section 4.04—Electrical Power

STANDBY ANNUNCIATOR PANEL RED AC SUPPLY LIGHT "ON"

On overhead electric panel:

GEN LOAD—CHECK

- If all generator loads are approximately zero:

LOSS OF AC SUPPLY PROCEDURE—APPLY

- If not all generator loads are approximately zero:

DC EMER BUS SUPPLY TRU3 CIRCUIT BREAKER—CHECK

- If circuit breaker has tripped:

DC EMER BUS SUPPLY TRU3 CIRCUIT BREAKER—RESET

—If reset is unsuccessful:

L and R AUDIO—ALTN

Anticipate the effects of an eventual EMER DC BUS failure, see EMER DC BUS FAULT procedure.

- If circuit breaker has not tripped:

L and R AUDIO—ALTN

Anticipate the effects of an eventual EMER DC BUS failure, see EMER DC BUS FAULT procedure."

(b) For all airplanes: Within 7 days after the effective date of this AD, revise the Normal Procedures section of the FAA-approved AFM to include the following statement. This may be accomplished by inserting a copy of this AD in the AFM.

"Section 5—Normal Procedures Insert in front of Sub-section 5.01.01—Take-off

- After engine start, select the Standby Annunciator Panel (SAP) backup mode ON via the BACKUP p/b at the SAP.

- Keep the SAP in the backup mode for the whole duration of flight until engine shutdown.

- Monitor the SAP.

Note: Failure conditions as presented on the SAP bypass the Flight Warning Computer (FWC) and are not subject to alert inhibition. Be aware that the red LG light on the SAP will illuminate in case one or both thrustlever(s) are below the minimum take-off position and the landing gear is not down."

(c) For all Model F28 Mark 0070 series airplanes; and for all Model F28 Mark 0100 in pre-SBF100-24-009 configuration or in post SBF100-24-030 configuration: Within 7 days after the effective date of this AD, revise the Abnormal Procedures section of the FAA-approved Airplane Flight Manual (AFM) to include the following statement. This may be accomplished by inserting a copy of this AD in the AFM.

"Section 4—Abnormal Procedures Add to Sub-section 4.04—Electrical Power

ERRATIC ELECTRICAL SYSTEM BEHAVIOR

In case of a continuous rattling sound, caused by the fast switching of relays and accompanied by blanking or erratic behavior of the three displays on the electric panel:

BATTERIES—SELECT MOMENTARILY OFF, THEN ON

AFFECTION SYSTEMS—RESTORE IF REQD

If the red AC SUPPLY light on the SAP comes ON:

SAP RED AC SUPPLY LIGHT 'ON' PROCEDURE—APPLY'

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

Note: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(e) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(f) This amendment becomes effective on October 27, 1995.

Issued in Renton, Washington, on October 4, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-25160 Filed 10-11-95; 8:45 am]

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14 CFR Part 39

[Docket No. 95-NM-184-AD; Amendment 39-9389 AD 95-21-04]

Airworthiness Directives; McDonnell Douglas Model MD-11 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; request for comments.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to certain McDonnell Douglas Model MD-11 series airplanes, that currently requires modification of the support structure of the cargo liner. That AD was prompted by a report of chafing and arcing in the vacuum waste exhaust heater that caused a spark to ignite the surrounding insulation blankets. The actions specified in that AD are intended to prevent fire and/or smoke due to chafing and arcing of the vacuum waste exhaust port heater. This amendment expands the applicability of the existing rule to include additional affected airplanes. It also provides for an alternative method of modification.

DATES: Effective October 27, 1995.