

requirements in rules and regulations as proposed.

List of Subjects in 7 CFR Part 1230

Administrative practice and procedure, Advertising, Agricultural research, Marketing agreement, Meat and meat products, Pork and pork products.

For the reasons set forth in the preamble, 7 CFR part 1230 is amended as set forth below:

PART 1230—PORK PROMOTION, RESEARCH, AND CONSUMER INFORMATION

1. The authority citation for 7 CFR Part 1230 continues to read as follows:

Authority: 7 U.S.C. 4801–4819.

2. In § 1230.74, paragraph (b) is revised and (c) is removed to read as follows:

§ 1230.74 Prohibited use of distributed assessments.

(b) Organizations receiving distributions of assessments from the Board shall furnish the Board with annual financial statements audited by a certified public accountant of all funds distributed to such organizations pursuant to this subpart and any other reports as may be required by the Secretary or the Board in order to verify the use of such funds.

3. A new § 1230.115 is added to Subpart B—Rules and Regulations to read as follows:

§ 1230.115 Submission of annual financial statements.

State Pork Producer Associations, as defined in § 1230.25, that receive distributions of assessments pursuant to § 1230.72 and that receive less than \$30,000 in assessments annually, may satisfy the requirements of § 1230.74(b) by providing to the Board unaudited annual financial statements prepared by State association staff members or individuals who prepare annual financial statements, provided that two members of the State association attest to and certify such financial statements. Notwithstanding any provisions of the Order to the contrary, State associations that receive less than \$30,000 in distributed assessments annually and submit unaudited annual financial statements to the Board shall be required to submit an annual financial statement audited by a certified public accountant at least once every 5 years, or more frequently if deemed necessary by the Board or the Secretary. The Board may elect to conduct its own audit of the annual financial statements of State

Pork Producer Associations that receive less than \$2,000 in distributed assessments annually, every 5 years in lieu of the required financial statements.

Dated: June 22, 1995.

David R. Shipman,

Acting Deputy Assistant Secretary, Marketing and Regulatory Programs.

[FR Doc. 95-15948 Filed 6-28-95; 8:45 am]

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Issued in Burlington, MA, on June 14, 1995.

Robert Guyotte,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

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14 CFR Part 39

[Docket No. 94-NM-251-AD; Amendment 39-9280; AD 95-12-27]

Airworthiness Directives; Boeing Model 747-400 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to certain Boeing Model 747-400 series airplanes, that currently requires a revision to the input wiring for the flap control unit (FCU). This amendment requires a new systems test for the wiring of the trailing edge flap, and also expands the applicability of the existing AD to include additional airplanes. This amendment is prompted by a report indicating that a wiring error was not detected by the system test required by the existing AD. The actions specified by this AD are intended to prevent the possibility of an all-flaps-up landing due to the loss of control of all flap operations.

DATES: Effective July 31, 1995.

The incorporation by reference of Boeing Service Bulletin 747-27A2346, Revision 2, dated January 12, 1995, as listed in the regulations, is approved by the Director of the Federal Register as of July 31, 1995.

The incorporation by reference of Boeing Service Bulletin 747-27A2346, Revision 1, dated May 19, 1994, as listed in the regulations, was approved previously by the Director of the Federal Register as of August 10, 1994 (59 FR 35240, July 11, 1994).

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Kristin Larson, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, Seattle Aircraft Certification

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 94-ANE-21; Amendment 39-9227; AD 95-10-10]

Airworthiness Directives; Pratt & Whitney JT8D Series Turbofan Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule; correction.

SUMMARY: This document makes a correction to Airworthiness Directive (AD) 95-10-10 applicable to certain Pratt & Whitney (PW) JT8D series turbofan engines that was published in the **Federal Register** on May 22, 1995 (60 FR 27020). The complete listing of affected PW JT8D series turbofan engines in the Applicability paragraph was inadvertently omitted. This document corrects the Applicability paragraph. In all other respects, the original document remains the same.

DATES: Effective July 21, 1995.

SUPPLEMENTARY INFORMATION: A final rule airworthiness directive applicable to Pratt & Whitney (PW) JT8D series turbofan engines, was published in the **Federal Register** on May 22, 1995 (60 FR 27020). The following correction is needed:

On page 27021, in the third column, in the Applicability paragraph, in the third paragraph, third line, that begins with “-17, and -17AR turbofan engines”, it should read “-17, and -17AR turbofan engines containing front compressor fan hub Part Number (P/N) 817401 with the following serial numbers: J78892 through J80538, K32019 through K34018, L32197 through L34133, or M05722 through M07296; and all serial numbers of fan hubs P/N 594301, 640601, 743301, 749801, 750101, 791801, and 806001. These engines are installed on but not limited to Boeing 727 and 737 series, and McDonnell Douglas DC-9 series aircraft.”

Office, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (206) 227-1760; fax (206) 227-1181.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 94-14-21, amendment 39-8970 (59 FR 35240, July 11, 1994), which is applicable to certain Boeing Model 747-400 series airplanes, was published in the **Federal Register** on February 15, 1995 (60 FR 8591). The action proposed to continue to require a revision of the input wiring for the flap control unit (FCU), but would include the addition of a new systems test for the wiring of the trailing edge flap. The action also proposed to expand the applicability of the existing AD to include additional airplanes.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comments received.

The manufacturer states that the numbers referred to as "serial numbers" in paragraphs (a), (b), and (c) of the proposal are designated incorrectly, and requests that the numbers be referred to as "line numbers." The FAA concurs and has revised paragraphs (a), (b), and (c) of the final rule to reflect this change.

One commenter requests that the additional systems test for the wiring, as proposed in paragraph (c) in the AD, be deleted. The commenter states that this testing is unnecessary because the specific procedures provided by Boeing Service Bulletin 747-27A2346, Revision 1, dated May 19, 1994 (which is the appropriate source of service information for existing AD 94-14-21), ensure that the wires are installed in the correct pin locations. The FAA does not concur. This AD was prompted by a report from an operator indicating that a wiring error of the landing gear module was discovered after the accomplishment of the wiring systems check required by AD 94-14-21. The FAA has determined that the wiring systems check required by that AD does not provide adequate verification that the wiring modification was made correctly. Therefore, the additional systems check as specified in paragraph (c) of the final rule is required.

After careful review of the available data, including the comments noted above, the FAA has determined that air safety and the public interest require the adoption of the rule with the changes previously described. The FAA has determined that these changes will neither increase the economic burden

on any operator nor increase the scope of the AD.

There are approximately 310 Model 747-400 series airplanes of the affected design in the worldwide fleet. The FAA estimates that 36 airplanes of U.S. registry will be affected by this AD, that it will take approximately 0.5 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$1,080, or \$30 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39-8970 (59 FR 35240, July 11, 1994), and by adding a new airworthiness directive (AD), amendment 39-9280, to read as follows:

95-12-27 Boeing: Amendment 39-9280.

Docket 94-NM-251-AD. Supersedes AD 94-14-21, Amendment 39-8970.

Applicability: Model 747-400 series airplanes having line numbers 696 through 1036 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (d) of this AD to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

Note 2: Paragraph (a) of this AD merely restates the requirements of paragraph (a) of AD 94-14-21, amendment 39-8970. As allowed by the phrase, "unless accomplished previously," if those requirements of AD 94-14-21 have already been accomplished, this AD does not require that those actions be repeated.

To prevent the possibility of an all-flaps-up landing due to the loss of control of flap operations, accomplish the following:

(a) For airplanes having line numbers 696 through 1019 inclusive, and 1021 through 1026 inclusive: Within 30 days after August 10, 1994 (the effective date of AD 94-14-21, amendment 39-8970), revise the input wiring for the flap control unit (FCU) in accordance with Boeing Service Bulletin 747-27A2346, Revision 1, dated May 19, 1994, or Revision 2, dated January 12, 1995.

(b) For airplanes having line numbers 1020, and 1027 through 1036 inclusive: Within 30 days after the effective date of this AD, revise the input wiring for the FCU in accordance with Boeing Service Bulletin 747-27A2346, Revision 2, dated January 12, 1995.

(c) For airplanes having line numbers 696 through 1036 inclusive: Within 120 days after the effective date of this AD, perform the additional systems test for the wiring of the trailing edge flap in accordance with Boeing Service Bulletin 747-27A2346, Revision 2, dated January 12, 1995.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA, Transport Aircraft Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Seattle ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Seattle ACO.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(f) The wire systems test and wiring revision shall be done in accordance with Boeing Service Bulletin 747-27A2346, Revision 1, dated May 19, 1994, and Boeing Service Bulletin 747-27A2346, Revision 2, dated January 12, 1995. The incorporation by reference is approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. The incorporation by reference of Boeing Service Bulletin 747-27A2346, Revision 1, dated May 19, 1994, was approved previously by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51 as of August 10, 1994 (59 FR 35240, July 11, 1994). Copies may be obtained from Boeing Commercial Airplane Group, P.O. Box 3707, Seattle, Washington 98124-2207. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(g) This amendment becomes effective on July 31, 1995.

Issued in Renton, Washington, on June 9, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-14632 Filed 6-28-95; 8:45 am]

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14 CFR Part 39

[Docket No. 93-CE-59-AD; Amendment 39-9289; AD 95-13-09]

Airworthiness Directives; Grob Luft Und Raumfahrt Models G102 Astir CS, Club Astir IIb, Twin Astir, Speed Astir, Standard Astir II, and Speed Astir IIb Sailplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain Grob Luft Und Raumfahrt (Grob) Models G102 Astir CS, Club Astir IIb, Twin Astir, Speed

Astir, Standard Astir II, and Speed Astir IIb sailplanes. This action requires inspecting all elevator and rudder hinges for damage (delamination, cracks, corrosion, or buckling), and repairing any damaged parts. Several occurrences of inner elevator hinges separating during flight operation prompted this AD action. The actions specified by this AD are intended to prevent these hinges from separating, which could result in sailplane flutter and eventual loss of control of the sailplane.

DATES: Effective August 15, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 15, 1995.

ADDRESSES: Service information that applies to this AD may be obtained from Grob Luft und Raumfahrt, D-8939 Mattsies, Germany. This information may also be examined at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. Herman Belderok, Project Officer, Sailplanes, Small Airplane Directorate, Aircraft Certification Service, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone (816) 426-6932; facsimile (816) 426-2169.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Grob Models G102 Astir CS, Club Astir IIb, Twin Astir, Speed Astir, Standard Astir II, and Speed Astir IIb sailplanes was published in the **Federal Register** on January 18, 1995 (60 FR 3588). The action proposed to require inspecting all elevator and rudder hinges for damage (delamination, cracks, corrosion, or buckling), and repairing any damaged parts. Accomplishment of the proposed actions would be in accordance with the *III. Procedure* section of Grob Repair Instruction No. 306-27/1 to Service Bulletin TM 306-27/1, dated June 4, 1991.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposed rule or the FAA's determination of the cost to the public.

After careful review of all available information related to the subject presented above, the FAA has determined that air safety and the

public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

The FAA estimates that 146 sailplanes in the U.S. registry will be affected by this AD, that it will take approximately 1 workhour per sailplane to accomplish the required action, and that the average labor rate is approximately \$60 an hour. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$8,760. This figure takes into account that no affected sailplane owner/operator has accomplished the required inspection.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the final evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows: