

Service Bulletin 52-JM 7752, dated December 17, 1991. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Jetstream Aircraft Limited, Manager Product Support, Prestwick Airport, Ayrshire, KA9 2RW Scotland; or Jetstream Aircraft Inc., Librarian, P.O. Box 16029, Dulles International Airport, Washington, DC 20041-6029. Copies may be inspected at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(f) This amendment (39-9256) becomes effective on July 24, 1995.

Issued in Kansas City, Missouri, on May 26, 1995.

Henry A. Armstrong,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-13623 Filed 6-6-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 94-NM-241-AD; Amendment 39-9253; AD 95-12-03]

Airworthiness Directives; Fokker Model F28 Mark 0100 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Fokker Model F28 Mark 0100 series airplanes, that requires installation of reinforcement plates at certain fuselage stations. This amendment is prompted by a report indicating that cracks were found in the frame strips at certain fuselage stations on a Model F28 Mark 0100 series airplane test article due to fatigue-related stress. The actions specified by this AD are intended to prevent such fatigue-related cracking, which could result in reduced structural integrity of the fuselage pressure vessel.

DATES: Effective July 7, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 7, 1995.

ADDRESSES: The service information referenced in this AD may be obtained from Fokker Aircraft USA, Inc., 1199 North Fairfax Street, Alexandria, Virginia 22314. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Tim Dulin, Aerospace Engineer, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (206) 227-2141; fax (206) 227-1320.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Fokker Model F28 Mark 0100 series airplanes was published in the **Federal Register** on March 3, 1995 (60 FR 11944). That action proposed to require installation of reinforcement plates at left and right fuselage stations 14911 and 17011.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter supports the proposed rule.

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

The FAA estimates that 45 airplanes of U.S. registry will be affected by this AD, that it will take approximately 160 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$3,800 per airplane. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$603,000, or \$13,400 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic

impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

95-12-03 Fokker: Amendment 39-9253. Docket 94-NM-241-AD.

Applicability: Model F28 Mark 0100 series airplanes, serial numbers 11244 through 11371 inclusive, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (b) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To prevent cracking in the frame strips at fuselage stations 14911 and 17011, which could result in reduced structural integrity of the fuselage pressure vessel, accomplish the following:

(a) Prior to the accumulation of 24,000 total flight cycles, or within 6 months after the effective date of this AD, whichever occurs

later, install reinforcement plates at left and right fuselage stations 14911 and 17011, in accordance with Fokker Service Bulletin SBF100-53-072, dated March 12, 1993.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this OAD can be accomplished.

(d) The installation shall be done in accordance with Fokker Service Bulletin SBF100-53-072, dated March 12, 1993. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fokker Aircraft USA, Inc., 1199 North Fairfax Street, Alexandria, Virginia 22314. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(e) This amendment becomes effective on July 7, 1995.

Issued in Renton, Washington, on May 26, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.
[FR Doc. 95-13502 Filed 6-6-95; 8:45 am]

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14 CFR Part 39

[Docket No. 94-CE-26-AD; Amendment 39-9249; AD 95-11-16]

Airworthiness Directives; SOCATA Groupe AEROSPATIALE TBM 700 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) that applies to certain SOCATA Groupe AEROSPATIALE (Socata) TBM 700 airplanes. This action requires installing pneumatic deicers on the elevator horn leading edges. Ice accumulation on one of the affected airplanes during flight testing in icing conditions prompted the required action. The actions specified

by this AD are intended to prevent ice accumulation on the elevator horn, which could lead to loss of control of the airplane.

DATES: Effective July 19, 1995.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 19, 1995.

ADDRESSES: Service information that applies to this AD may be obtained from the SOCATA Groupe AEROSPATIALE, Socata Product Support, Aeroport Tarbes-Ossun-Lourdes, B P 930, 65009 Tarbes Cedex, France; telephone 62.41.74.26; facsimile 62.41.74.32; or the Product Support Manager, U.S. AEROSPATIALE, 2701 Forum Drive, Grand Prairie, Texas 75053; telephone (214) 641-3614; facsimile (214) 641-3527. This information may also be examined at the FAA, Central Region, Office of the Assistant Chief Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT: Mr. Raymond A. Stoer, Program Officer, Brussels Aircraft Certification Office, FAA, Europe, Africa, and Middle East Office, c/o American Embassy, B-1000 Brussels, Belgium; telephone (322) 513.38.30; facsimile (322) 230.68.99; or Mr. Mike Kiesov, Aerospace Engineer, FAA, Small Airplane Directorate, 1201 Walnut Street, suite 900, Kansas City, Missouri 64106; telephone (816) 426-6934; facsimile (816) 426-2169.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to certain Socata TBM 700 airplanes was published in the **Federal Register** on January 20, 1995 (60 FR 4117). The action proposed to require installing pneumatic deicers on the elevator horn leading edges. Accomplishment of the proposed installation would be in accordance with Socata Technical Instruction of Modification No. OPT70 K020-30, dated February 1993.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the one comment received.

Socata recommends that AD action is not justified because it believes all owners/operators of the affected airplanes have installed pneumatic deicers on the elevator horn leading edges. The FAA does not concur that AD action is not justified even if all owners/operators may have already complied. AD's are issued to assure that

each affected airplane is in compliance with the action, and that those airplanes continue to be in compliance. Even if all owners/operators have complied with this action, the AD will ensure that these airplanes continue to have these pneumatic deicers installed and that any airplanes added to the U.S. registry will have pneumatic deicers installed. The AD is unchanged as a result of this comment.

No comments were received on the FAA's determination of the cost to the public.

After careful review of all available information including the comment discussed above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed except for minor editorial corrections. The FAA has determined that these minor corrections will not change the meaning of the AD and will not add any additional burden upon the public than was already proposed.

The FAA estimates that 20 airplanes in the U.S. registry will be affected by this AD, that it will take approximately 25 workhours per airplane to accomplish the required action, and that the average labor rate is approximately \$60 an hour. Parts cost \$3,710 per airplane. Based on these figures, the total cost impact of this AD on U.S. operators is estimated to be \$104,200. This figure is based upon the assumption that no affected airplane/operator has accomplished the required action. Socata has informed the FAA that it believes all affected airplane owners/operators have already accomplished the required installation. With this in mind, this action will impose no cost impact upon U.S. operators.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory