

certificated for operation in the United States.

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require replacement of the return filter diaphragm assemblies on hydraulic systems 1 and 2 with modified filter units having new diaphragms. The actions would be required to be accomplished in accordance with the service bulletin described previously.

As a result of recent communications with the Air Transport Association (ATA) of America, the FAA has learned that, in general, some operators may misunderstand the legal effect of AD's on airplanes that are identified in the applicability provision of the AD, but that have been altered or repaired in the area addressed by the AD. The FAA points out that all airplanes identified in the applicability provision of an AD are legally subject to the AD. If an airplane has been altered or repaired in the affected area in such a way as to affect compliance with the AD, the owner or operator is required to obtain FAA approval for an alternative method of compliance with the AD, in accordance with the paragraph of each AD that provides for such approvals. A note has been included in this notice to clarify this long-standing requirement.

The FAA estimates that 119 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 2 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would be provided by the parts manufacturer at no cost to operators. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$14,280, or \$120 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by

contacting the Rules Docket at the location provided under the caption

ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Fokker: Docket 95-NM-38-AD.

Applicability: Model F28 Mark 0100 series airplanes equipped with Aircraft Porous Media Europe (APME) Limited hydraulic return filter assemblies having part numbers (P/N) QA07236 and QA07237, all serial numbers; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (c) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To prevent overpressure in the hydraulic return system, which could result in reduced braking performance and/or blown tires due to brake overheating, accomplish the following:

(a) Within 6 months after the effective date of this AD, replace the return filters, P/N's QA07236 and QA07237, on hydraulic systems 1 and 2, respectively, with modified return filter units, in accordance with Fokker Service Bulletin SBF100-29-025, dated December 31, 1993.

(b) As of the effective date of this AD, no person shall install on any airplane a return filter unit, P/N QA07236 or QA07237, on hydraulic system 1 or 2, respectively, unless that unit has been modified in accordance with Fokker Service Bulletin SBF100-29-025, dated December 31, 1993.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on May 19, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-12827 Filed 5-24-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 39

[Docket No. 94-NM-162-AD]

Airworthiness Directives; Beech Model 400, 400A, and MU-300-10 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Beech Model 400, 400A, and MU-300-10 airplanes. This proposal would require installation of an improved adjustment mechanism on the flightcrew seats and replacement of the existing aluminum seat reinforcement assemblies with steel assemblies. This

proposal is prompted by reports of incomplete latching of the existing adjustment mechanism and cracked reinforcement assemblies, which could result in sudden shifting of a flightcrew seat. The actions specified by the proposed AD are intended to prevent such shifting of a flightcrew seat, which could impair the flightcrew's ability to control the airplane.

DATES: Comments must be received by July 5, 1995.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 94-NM-162-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Raytheon Aircraft Company, P. O. Box 85, Wichita, Kansas 67201-0085. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas.

FOR FURTHER INFORMATION CONTACT: Larry Engler, Aerospace Engineer, Airframe Branch, ACE-118W, FAA, Small Airplane Directorate, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Mid-Continent Airport, Wichita, Kansas 67209; telephone (316) 946-4122; fax (316) 946-4407.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by

interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 94-NM-162-AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 94-NM-162-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The FAA has received several reports of incomplete latching of the adjustment mechanism on a flightcrew seat on Beech Model 400 series airplanes, which can result in a sudden shift of the seat position. The latching pins are designed to go into an adjustment hole on each rear leg assembly of the crew seat. However, due to the rigidity of the pin/tube assembly, both pins could not slip completely into the latched position unless the holes on both rear assemblies were aligned. Additionally, the FAA has received reports of cracking in the aluminum reinforcement assemblies of the flightcrew seat, which also may contribute to shifting of the seat. Shifting of a flightcrew seat during flight, if not corrected, could impair the flightcrews's ability to control the airplane.

The FAA has reviewed and approved Beechcraft Service Bulletin No. 2536, Revision 1, dated April 1995, which describes procedures for installing an improved adjustment mechanism on the flightcrew seats. This adjustment mechanism will allow each pin to slide into the latched position without both rear leg assemblies being aligned. The service bulletin also describes procedures for replacing the existing aluminum seat reinforcement assemblies with stronger steel assemblies.

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require installing an improved adjustment mechanism on the flightcrew seats, and replacing the existing aluminum seat reinforcement assemblies with steel assemblies. The actions would be required to be

accomplished in accordance with the service bulletin described previously.

As a result of recent communications with the Air Transport Association (ATA) of America, the FAA has learned that, in general, some operators may misunderstand the legal effect of AD's on airplanes that are identified in the applicability provision of the AD, but that have been altered or repaired in the area addressed by the AD. The FAA points out that all airplanes identified in the applicability provision of an AD are legally subject to the AD. If an airplane has been altered or repaired in the affected area in such a way as to affect compliance with the AD, the owner or operator is required to obtain FAA approval for an alternative method of compliance with the AD, in accordance with the paragraph of each AD that provides for such approvals. A note has been included in this notice to clarify this long-standing requirement.

There are approximately 169 Model 400, 400A, and MU-300-10 airplanes of the affected design in the worldwide fleet. The FAA estimates that 121 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 24 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$700 per airplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$258,940, or \$2,140 per airplane.

The total cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities

under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Beech Aircraft Corporation: Docket 94-NM-162-AD.

Applicability: Model 400 airplanes, serial numbers RJ-1 through RJ-65 inclusive; Model 400A airplanes, serial numbers RK-1 through RK-93 inclusive; and Model MU-300-10 airplanes, serial numbers A1001SA through A1011SA inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (b) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To prevent reduced controllability of the airplane due to a shifting of the flightcrew seat during flight, accomplish the following:

(a) Within 200 hours time-in-service after the effective date of this AD, install an improved adjustment mechanism on the

flightcrew seat, and replace the existing aluminum seat reinforcement assemblies with steel assemblies, in accordance with Beechcraft Service Bulletin No. 2536, Revision 1, dated April 1995.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Wichita Aircraft Certification Office (ACO), FAA, Small Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Wichita ACO.

(c) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on May 19, 1995.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 95-12828 Filed 5-24-95; 8:45 am]

BILLING CODE 4910-13-U

14 CFR Part 91 and 135

[Docket No. 25149, Notice 95-6; Special Federal Aviation Regulation (SFAR) No. 50-2]

RIN 2120-AF60

Special Flight Rules in the Vicinity of the Grand Canyon National Park

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Notice of proposed rulemaking (NPRM); correction.

SUMMARY: This document contains a correction to a Notice of Proposed Rulemaking (NPRM), Special Flight Rules in the Vicinity of the Grand Canyon National Park, SFAR No. 50-2, published in the **Federal Register** on April 12, 1995 (60 FR 18700).

FOR FURTHER INFORMATION CONTACT: Mrs. Ellen Crum, Air Traffic Rules Branch, ATP-230, Telephone (202) 267-8783.

SUPPLEMENTARY INFORMATION:

History

Federal Register Document, Docket No. 25149, published on April 12, 1995 (60 FR 18700), proposed to extend the effectiveness of SFAR No. 50-2. The Notice No. was omitted from the heading.

Correction to NPRM

The NPRM, published in the **Federal Register** on April 12, 1995 (60 FR 18700), is corrected as follows:

1. By adding the words "Notice 95-6;" on page 18700, first column, in the heading, after "Docket No. 25149,".

Issued in Washington, DC, on May 17, 1995.

Donald P. Byrne,

Assistant Chief Counsel, Office of the Chief Counsel.

[FR Doc. 95-12753 Filed 5-24-95; 8:45 am]

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DEPARTMENT OF LABOR

Occupational Safety and Health Administration

29 CFR Parts 1910, 1915, and 1926

[Docket No. H-049]

RIN 1218-0099

Respiratory Protection

AGENCY: Occupational Safety and Health Administration (OSHA), Labor.

ACTION: Scheduling of a technical panel discussion on assigned protection factors as part of the pending rulemaking hearing.

SUMMARY: By this document, the Occupational Safety and Health Administration (OSHA) announces the convening, pursuant to 29 CFR 1911.4, of a panel to discuss certain science-policy issues involved in respirator selection, focusing on the need for, and limitations of, assigning protection factors for respirators by class. This panel discussion will take place on June 15, 1995, as part of the scheduled rulemaking hearing on respiratory protection. Details on the process and procedures associated with the panel discussion are described below.

DATES: The hearing on the proposed rule will begin on June 6, 1995. The panel discussion is scheduled for 9:00 a.m. on June 15, 1995.

ADDRESSES: The hearing and panel discussion will be held in the auditorium of the U.S. Department of Labor, 200 Constitution Avenue, N.W., Washington, D.C.

FOR FURTHER INFORMATION CONTACT: Proposal: Mr. Richard Liblong, Office of Information and Consumer Affairs, Occupational Safety and Health Administration, 200 Constitution Avenue, N.W., Room N3647, Washington, D.C. 20210; (202) 219-8151.

Hearing: Mr. Thomas Hall, Division of Consumer Affairs, Occupational Safety