(2) The maximum ambient atmospheric temperature for which engine cooling was demonstrated must be stated in the performance information section of the Flight Manual, if the applicable regulations under which the aircraft was type certificated do not require ambient temperature on engine cooling operating limitations in the Flight Manual.

[EFFECTIVE DATE NOTE: By Amdt. 21-92, 74 FR 53385, Oct. 16, 2009, §21.5 was amended by revising paragraph (a), effective Apr. 14, 2010. This effective date was subsequently postponed to Apr. 16, 2011 at 75 FR 9095, Mar. 1, 2010. For the convenience of the user, the revised text is set forth as follows:]

§ 21.5 Airplane or Rotorcraft Flight Manual.
(a) With each airplane or rotorcraft not type certificated with an Airplane or Rotorcraft Flight Manual and having no flight time before March 1, 1979, the holder of a type certificate (including amended or supplemental type certificates) or the licensee of a type certificate must make available to the owner at the time of delivery of the aircraft a current approved Airplane or Rotorcraft Flight Manual.

§ 21.6 Manufacture of new aircraft, aircraft engines, and propellers.
(a) Except as specified in paragraphs (b) and (c) of this section, no person may manufacture a new aircraft, aircraft engine, or propeller based on a type certificate unless the person—
(1) Is the holder of the type certificate or has a licensing agreement from the holder of the type certificate to manufacture the product; and
(2) Meets the requirements of subpart F or G of this part.
(b) A person may manufacture one new aircraft based on a type certificate without meeting the requirements of paragraph (a) of this section if that person can provide evidence acceptable to the FAA that the manufacture of the aircraft by that person began before August 5, 2004.
(c) The requirements of this section do not apply to—
(1) New aircraft imported under the provisions of §§21.183(c), 21.184(b), or 21.185(c); and
(2) New aircraft engines or propellers imported under the provisions of §21.500.


§ 21.7 Continued airworthiness and safety improvements for transport category airplanes.
(a) On or after December 10, 2007, the holder of a design approval and an applicant for a design approval must comply with the applicable continued airworthiness and safety improvement requirements of part 26 of this subchapter.
(b) For new transport category airplanes manufactured under the authority of the FAA, the holder or licensee of a type certificate must meet the applicable continued airworthiness and safety improvement requirements specified in part 26 of this subchapter for new production airplanes. Those requirements only apply if the FAA has jurisdiction over the organization responsible for final assembly of the airplane.


§ 21.8 Approval of articles.
If an article is required to be approved under this chapter, it may be approved—
(a) Under a PMA;
(b) Under a TSO;
(c) In conjunction with type certification procedures for a product; or
(d) In any other manner approved by the FAA.


§ 21.9 Replacement and modification articles.
(a) If a person knows, or should know, that a replacement or modification article is reasonably likely to be installed on a type-certificated product, the person may not produce that article unless it is—
(1) Produced under a type certificate;
§ 21.11

(a) Procedural requirements for the issue of type certificates for aircraft, aircraft engines, and propellers; and
(b) Rules governing the holders of those certificates.

§ 21.13 Eligibility.

Any interested person may apply for a type certificate.

[Amtd. 21–25, 34 FR 14068, Sept. 5, 1969]

§ 21.15 Application for type certificate.

(a) An application for a type certificate is made on a form and in a manner prescribed by the Administrator and is submitted to the appropriate Aircraft Certification Office.

(b) An application for an aircraft type certificate must be accompanied by a three-view drawing of that aircraft and available preliminary basic data.

(c) An application for an aircraft engine type certificate must be accompanied by a description of the engine design features, the engine operating characteristics, and the proposed engine operating limitations.


EFFECTIVE DATE NOTE: By Amdt. 21–92, 74 FR 53385, Oct. 16, 2009, §21.15 was amended by removing the words “Aircraft Certification Office” in paragraph (a) and adding in their place the words “aircraft certification office”, effective Apr. 14, 2010. This effective date was subsequently postponed to become effective Apr. 16, 2011 at 75 FR 9995, Mar. 1, 2010.

§ 21.16 Special conditions.

If the Administrator finds that the airworthiness regulations of this subchapter do not contain adequate or appropriate safety standards for an aircraft, aircraft engine, or propeller because of a novel or unusual design feature of the aircraft, aircraft engine or propeller, he prescribes special conditions and amendments thereto for the product. The special conditions are issued in accordance with Part 11 of this chapter and contain such safety standards for the aircraft, aircraft engine or propeller as the Administrator finds necessary to establish a level of

Subpart B—Type Certificates

SOURCE: Docket No. 5085, 29 FR 14564, Oct. 24, 1964, unless otherwise noted.

§ 21.11 Applicability.

This subpart prescribes—