§ 27.1461

must cover the airspeed range over which the aircraft is to be operated, the range of altitude to which the aircraft is limited, and 360 degrees of heading. Correlation may be established on the ground as appropriate.

- (d) Each recorder container must:
- (1) Be either bright orange or bright yellow:
- (2) Have a reflective tape affixed to its external surface to facilitate its location under water; and
- (3) Have an underwater locating device, when required by the operating rules of this chapter, on or adjacent to the container which is secured in such a manner that they are not likely to be separated during crash impact.
- (e) When both a cockpit voice recorder and a flight data recorder are required by the operating rules, one combination unit may be installed, provided that all other requirements of this section and the requirements for cockpit voice recorders under this part are met.

[Amdt. 27–22, 53 FR 26144, July 11, 1988, as amended by Amdt. 27–43, 73 FR 12564, Mar. 7, 2008; 74 FR 32800, July 9, 2009]

§ 27.1461 Equipment containing high energy rotors.

- (a) Equipment containing high energy rotors must meet paragraph (b), (c), or (d) of this section.
- (b) High energy rotors contained in equipment must be able to withstand damage caused by malfunctions, vibration, abnormal speeds, and abnormal temperatures. In addition—
- (1) Auxiliary rotor cases must be able to contain damage caused by the failure of high energy rotor blades; and
- (2) Equipment control devices, systems, and instrumentation must reasonably ensure that no operating limitations affecting the integrity of high energy rotors will be exceeded in service.
- (c) It must be shown by test that equipment containing high energy rotors can contain any failure of a high energy rotor that occurs at the highest speed obtainable with the normal speed control devices inoperative.
- (d) Equipment containing high energy rotors must be located where rotor failure will neither endanger the

occupants nor adversely affect continued safe flight.

[Amdt. 27-2, 33 FR 964, Jan. 26, 1968]

Subpart G—Operating Limitations and Information

§27.1501 General.

- (a) Each operating limitation specified in §§27.1503 through 27.1525 and other limitations and information necessary for safe operation must be established.
- (b) The operating limitations and other information necessary for safe operation must be made available to the crewmembers as prescribed in §§ 27.1541 through 27.1589.

(Secs. 313(a), 601, 603, 604, and 605 of the Federal Aviation Act of 1958 (49 U.S.C. 1354(a), 1421, 1423, 1424, and 1425); and sec. 6(c) of the Dept. of Transportation Act (49 U.S.C. 1655(c)))

[Amdt. 27-14, 43 FR 2325, Jan. 16, 1978]

OPERATING LIMITATIONS

§27.1503 Airspeed limitations: general.

- (a) An operating speed range must be established.
- (b) When airspeed limitations are a function of weight, weight distribution, altitude, rotor speed, power, or other factors, airspeed limitations corresponding with the critical combinations of these factors must be established.

§27.1505 Never-exceed speed.

- (a) The never-exceed speed, V_{NE} , must be established so that it is—
 - (1) Not less than 40 knots (CAS); and
- (2) Not more than the lesser of—
- (i) 0.9 times the maximum forward speeds established under § 27.309;
- (ii) 0.9 times the maximum speed shown under §§ 27.251 and 27.629; or
- (iii) 0.9 times the maximum speed substantiated for advancing blade tip mach number effects.
- (b) V_{NE} may vary with altitude, r.p.m., temperature, and weight, if—
- (1) No more than two of these variables (or no more than two instruments integrating more than one of these variables) are used at one time; and